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FILE: RESULTS OF AN INVESTIGATION OF HYPERSONIC VISC

EFFECTS ON AN 0.01 SCALE SPACE SHUTTLE ORBITER 51-0

AEDC-VKF HYPERVELOCITY WIND TUNNEL (OA81)

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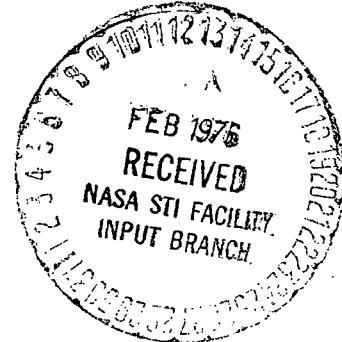
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CHRYSLER
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THE FOLLOWING CHANGES APPLY TO PUBLICATION: DMS-DR-2152

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AEDC-VKF HYPERVELOCITY WIND TUNNEL (OA81)

NUMBER: DR-2152 DATE: October, 1974 BRANCH: Flight Technology

The attached Revision A of DMS-DR-2152 completely replaces DMS-DR-2152 dated October, 1974.

This revision is published to:

- 1) incorporate the updated data released in May 1975 for Mach 16, Re/ft. 0.5×10^6 .
- 2) correct the improper XCP/L values in the old report
- 3) add \bar{V}_∞ values to the data array
- 4) add contamination factor (F) to Mach 20 data
- 5) add data plots

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PAGE 1 OF 1

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December, 1975

REVISION A
DMS-DR-2152
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RESULTS OF AN INVESTIGATION OF HYPERSONIC
VISCOS INTERACTION EFFECTS ON AN 0.01
SCALE SPACE SHUTTLE ORBITER 51-0 MODEL IN
THE AEDC-VKF HYPERVELOCITY WIND TUNNEL
(OA81)

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for

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Johnson Space Center
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Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number: AEDC F VA489
NASA Series Number: OA81
Model Number: 51-0
Test Dates: 26 November through 27 December 1973
Occupancy Hours: 90

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Chrysler Corporation Space Division assumes no responsibility for the data presented other than display characteristics.

RESULTS OF AN INVESTIGATION OF HYPERSONIC
VISCOS INTERACTION EFFECTS ON AN 0.01
SCALE SPACE SHUTTLE ORBITER 51-0 MODEL IN
THE AEDC-VKF HYPERVELOCITY WIND TUNNEL
(OA81)

By

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ABSTRACT

The major hypersonic aerodynamic results obtained experimentally from the 0.010-scale (51-0) 140A/B Orbiter vehicle are presented herein. The results were obtained from the AEDC-VKF Hypervelocity Wind Tunnel F under test OA81.

The test was conducted over an angle-of-attack range from 20 to 35 degrees for nominal Mach numbers of 16 and 20. The Mach 16 data were obtained at Reynolds numbers, based on model length, of nominally 0.54×10^6 and 1.18×10^6 . The Mach number 20 data were obtained at nominal Reynolds numbers of 0.08×10^6 and 0.27×10^6 . Various elevon deflections ($\delta_e = -40$ degrees, 0 degrees, +15 degrees) and body flap deflections ($\delta_{BF} = -11.7$ degrees, 0 degrees, +16.3 degrees) were tested for determination of control effectiveness. The test objectives were as follows:

- 1) Verify December 1973 Aerodynamic Design Data Book estimates for $M_\infty > 10$.
- 2) Determine configuration forward and aft trim boundaries and control characteristics.
- 3) Determine high Mach number - low Reynolds number effects.
- 4) Determine if positive control surface flow separation is present and magnitude of separation effects.

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NOMENCLATURE
General

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
a		speed of sound; m/sec, ft/sec
C_p	CP	pressure coefficient; $(p_1 - p_\infty)/q$
M	MACH	Mach number; V/a
p		pressure; N/m ² , psf
q	$Q(\text{NSM})$ $Q(\text{PSF})$	dynamic pressure; $1/2\rho V^2$, N/m ² , psf
RN/L	RN/L	unit Reynolds number; per m, per ft
v		velocity; m/sec, ft/sec
α	ALPHA	angle of attack, degrees
β	BETA	angle of sideslip, degrees
ψ	PSI	angle of yaw, degrees
ϕ	PHI	angle of roll, degrees
ρ		mass density; kg/m ³ , slugs/ft ³

Reference & C.G. Definitions

Ab		base area; m ² , ft ²
b	BREF	wing span or reference span; m, ft
c.g.		center of gravity
$\frac{l}{c}$ REF	LREF	reference length or wing mean aerodynamic chord; m, ft
S	SREF	wing area or reference area; m ² , ft ²
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis

SUBSCRIPTS

b	base
l	local
s	static conditions
t	total conditions
∞	free stream

NOMENCLATURE (Continued)

Body-Axis System

<u>SYMBOL</u>	<u>PLOT SYMBOL</u>	<u>DEFINITION</u>
C_N	CN	normal-force coefficient; $\frac{\text{normal force}}{qS}$
C_A	CA	axial-force coefficient; $\frac{\text{axial force}}{qS}$
C_Y	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
C_{A_b}	CAB	base-force coefficient; $\frac{\text{base force}}{qS}$ $-A_b(p_b - p_\infty)/qS$
C_{A_f}	CAF	forebody axial force coefficient, $C_A - C_{A_b}$
C_m	CLM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS_f \text{REF}}$
C_n	CYN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qS_b}$
C_l	CBL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qS_b}$

Stability-Axis System

C_L	CL	lift coefficient; $\frac{\text{lift}}{qS}$
C_D	CD	drag coefficient; $\frac{\text{drag}}{qS}$
C_{D_b}	CDB	base-drag coefficient; $\frac{\text{base drag}}{qS}$
C_{D_f}	CDF	forebody drag coefficient; $C_D - C_{D_b}$
C_Y	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
C_m	CLM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS_f \text{REF}}$
C_n	CLN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qS_b}$
C_l	CSL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qS_b}$
L/D	L/D	lift-to-drag ratio; C_L/C_D
L/DF	L/DF	lift to forebody drag ratio; C_L/C_{D_f}

NOMENCLATURE - Continued
Additions to Standard List

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
C_{AE}	CAE	estimated axial-force coefficient, ($C_{Af_E} + C_{Ab_E}$) $M = 20$ data only
C_{Af_E}	CAFE	estimated forebody axial-force coefficient (see reference 1); $M = 20$ data only, see page 16
C_{Af}	CAF	forebody axial-force coefficient, ($C_A - C_{Ab}$) or ($C_{Ap} + C_{Ay}$)
C_{AM}	CAM	experimental axial-force coefficient, Mach = 20 data only; includes viscous force produced by tunnel flow contamination (see reference 1) and base pressure drag
C_{Af_M}	CAFM	experimental forebody axial-force coefficient, ($C_{AM} - C_{Ab}$); $M = 20$ data only
C_{Av}		viscous axial-force coefficient
C_{Ap}		pressure axial-force coefficient
C_{PB}	CPB	base pressure coefficient, $(P_B - P_\infty)/q_\infty$
C_∞		Chapman-Rubesin viscosity coefficient, $(\mu_w/\mu_\infty)(T_\infty/T_w)$
C'_∞		factor of proportionality as defined in Data Reduction section
F	C-FCTR	AEDC contamination correction factor, $(C_{Af_M}/C_{Af_E}) - 1$; see reference 5; $M = 20$ data only
F_{Af_C}		axial force resulting from flow contamination, $F_{Af_C} = F_{Af_M} - F_{Af_E}$
F_{Af}		forebody axial force corrected for tunnel flow contamination ($M_\infty = 20$ only)
C_{N_M}		measured normal force coefficient

NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
F_{AM}		measured total axial force, 1b
F_{AfM}		measured forebody axial force, 1b
F_N		normal force, 1b
F_{NA}		normal force at aft normal-force gage location, 1b
F_{NF}		normal force at forward normal-force gage location, 1b
h		enthalpy
ℓ_B	L	reference body length, 12.903 in.
M_∞	MACH	Mach number
MRP	MRP	moment reference point
M_y		pitching moment in the body axis system, in-lb $M_y = [\text{moment arm}_1 \text{ (in.)} \times F_{NF}] - [\text{moment arm}_2 \text{ (in.)} \times F_{NA}] + 0.25 \text{ in. } (F_{AM} \text{ or } F_{Af}), \text{ in-lb}$
P_∞	P	free-stream static pressure, psia
P_0	P0	reservoir pressure, psi
P'_0	POP	total pressure behind normal shock in test section, psia
Pr		Prandtl number, 0.71
P_B		model base pressure, psia
\dot{q}_0	Q0	stagnation heat-transfer rate on 1.0-in.-diam probe, $\text{Btu}/\text{ft}^2\text{-sec}$
$q(I)$		stagnation heat-transfer rate inferred from shoulder heat gages of 1.0-in.-diam hemisphere-cylinder, $\text{Btu}/\text{ft}^2\text{sec}$

NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
$Re_{\infty L}$	RE-L	Reynolds number based on free-stream conditions and model length
S_{wet}		wetted area of model lower surface, in ²
St_{∞}		Stanton number, $q_w / \rho_{\infty} U_{\infty} (h_0 - h_w)$
St_{∞} avg		average Stanton number acting on lower model surface
t	TIME	time, milliseconds
T_{∞}	T-INF	Free-stream temperature, °R
T_0	T0	reservoir temperature, °R
T_w		model surface temperature, °R
U_{∞}	U-INF	free-stream velocity, ft/sec
\bar{V}_{∞}	V-INF	hypersonic viscous parameter, $M_{\infty} \sqrt{C_{\infty}} / \sqrt{RE-L}$
\bar{V}'_{∞}	VLBAR	viscous interaction parameter, $M_{\infty} \sqrt{C'_{\infty}} / \sqrt{RE-L}$
$X_{cp/L}$ M		experimental center-of-pressure location, $(XMRP - CLM x l_{ref}/CN) / L_B$
$X_{cp/L}$	XCP/L	corrected center-of-pressure location, (see reference 5) $XCP/L = 0.65 - \bar{c}/L_B$ (CLM/CN)
Z		vertical coordinate of model axis system, see Fig. 2
α_S		sector angle of attack, deg
β	BETA	sideslip angle, deg
δ		control surface deflection angle, deg; positive deflections are:
AILRON		aileron-left trailing edge down
ELEVTR		elevator-trailing edge down
BDFLAP		bodyflap-trailing edge down
SPDBRK		Speedbrake

NOMENCLATURE (Continued)

<u>Symbol</u>	<u>Plot Symbol</u>	<u>Definition</u>
μ_∞		free-stream viscosity
μ_w		viscosity at wall temperature
τ		local shear stress, psi

Subscripts

e_L or e_R	elevon left and right, respectively
BF	bodyflap
M	experimental values
0	stagnation conditions
SB	speedbrake
∞	free-stream conditions

REMARKS

The data resulting from test OA81 contained several anomalies.

1) Tunnel flow was contaminated with fine dust particles produced by vaporization of copper, tungsten, and beryllium during electrical discharge in the tunnel arc chamber. The contamination problem was most severe at Mach 20/Reynolds number 0.27×10^6 and least severe at Mach 20/Reynolds number 0.08×10^6 . These data have been empirically corrected for contamination effects according to the relationship given in Figure 4.

2) Data recorded at different times during a test shot for the same nominal test condition exhibited significant scatter. This is caused primarily by the timewise variation of freestream conditions (inherent in most hotshot facilities).

3) Flow in the tunnel core had a slightly non-uniform pressure profile. Corrections for flow non-uniformities were determined testing the model in an inverted position and applied to the data.

4) The test section flow was divergent (source flow) because the tunnel F nozzle is conical. Effects of source flow were empirically corrected by procedures developed by AEDC.

Data corrections are summarized in the Data Reduction section of this report. A detailed description of these corrections is given in Reference 1. References 3 through 5 provide detailed independent

REMARKS (Concluded)

analyses of these data, explanation of their anomalies, and judiciously plotted results.

CONFIGURATIONS INVESTIGATED

The model (designated no. 51-0) was an 0.010-scale representation of the Space Shuttle 140A/B Orbiter Configuration, fabricated from magnesium (to minimize weight). The model was comprised of the following components:

<u>Symbol</u>	<u>Description</u>
B ₂₆	140A/B fuselage
C ₉	Configuration 3A canopy
E ₂₆	Configuration 4 elevon
F ₇	Configuration 3A bodyflap
M ₇	Configuration 3A OMS pods
N ₂₈	Configuration 3A OMS nozzles
R ₅	Configuration 3A rudder
V ₈	Configuration 3A vertical stabilizer
W ₁₁₆	Configuration 4 wing

Model elevon deflection angles of -40, 0 and 15 degrees were investigated. Bodyflap deflection angles of -11.7, 0, and 16.3 degrees were investigated. Speedbrake deflection angle was 55 degrees. Figure 2 presents a sketch of the model configuration. Table III provides detailed model component dimensional data.

INSTRUMENTATION

Arc chamber pressure, test section pitot pressure, and probe stagnation and cylindrical section heat transfer rates were measured to determine tunnel flow conditions (see reference 1). Pitot pressures were measured with ± 15 psid strain gauge pressure transducers calibrated at the specific pressure level occurring during each test condition. The stagnation heat transfer rates used in determining the tunnel flow conditions were inferred from measurements made on the cylindrical sections of two 1-inch diameter hemisphere-cylinder probes using 10 mil resistance thermometer slug calorimeters. Co-axial thermocouple gauges located at the stagnation points of these same probes were used as flow contamination monitors. The arc chamber reservoir pressure was measured using two strain gauge transducers, each having full-scale calibrated ranges of 5, 10, and 25 thousand psia.

Aerodynamic forces were measured using six-component force balances developed by AEDC VKF for use in hotshot-type tunnels. The balance load cells were instrumented with semiconductor strain gauges, and semiconductor accelerometers provided compensation for model inertia loads that result from vibrations of the model and its support hardware.

Base pressure measurements were made using Invar pressure transducers with a range of 0.001-0.1 psid and were mounted on the sting at the base of the model with the gauge orifice pointing downstream and protruding approximately 1/16-inch beyond the base of the model.

All test data were recorded on a 70-channel digital system capable of

INSTRUMENTATION - Concluded.

scanning all channels in one msec and storing up to 150 scans of data.

Basic data reduction was done off-line with a digital computer.

As a backup to the digital system, as well as to provide a quick-look at data results, output of each data channel was recorded on an oscillograph.

TEST FACILITY DESCRIPTION

The Hypervelocity Wind Tunnel F (reference 1) is an arc-driven wind tunnel of the hotshot type and capable of providing Mach numbers from about 8 to 20 over a Reynolds number per foot range from 0.05×10^6 to 70×10^6 . Test sections of 108-inch diameter ($M_\infty = 14$ to 20) and 54-inch diameter ($M_\infty = 10$ to 17) are available using a four-degree half-angle conical nozzle. The range of Mach numbers at a particular test section in the conical nozzle is obtained by using various throat diameters. The $M_\infty = 8$ contoured nozzle has a 25-inch exit diameter which connects to the 54-inch diameter test section and provides a free-jet exhaust. Nitrogen is the test gas used for aerodynamic studies; air is used for combustion tests. The test gas is confined to either a 1.0-, 2.5-, or a 4.0-cu-ft arc chamber where it is heated and compressed by an electric arc discharge. The increase in pressure results in a diaphragm rupture with the subsequent flow expansion through the nozzle. Test times are typically from 40 to 200 msec. Shadowgraph and Schlieren coverage are available at both test sections.

This test was conducted in the 108-inch diameter test section of the conical nozzle for both the Mach 16 and 20 test conditions. The 2.5-ft³ arc chamber was used and useful test times for Mach 16 and 20 were approximately 40 to 70 msec, respectively. Because of the relatively short test times, the model wall temperature remained essentially invariant from the initial value of approximately 540°R; thus, $T_w/T_0 = 0.15$ which approximates the condition of practical interest for reentry vehicles.

DATA REDUCTION

Freestream test conditions were computed using timewise measurements of reservoir pressure, test section pitot pressure, and reference hemisphere cylinder stagnation and shoulder heat transfer rates. Fay and Riddell stagnation point heat transfer theory was used to compute reservoir enthalpy. Quasi-steady isentropic flow equations were used to compute other freestream parameters.

Balance data were reduced to coefficient form in body and stability axes by standard AEDC methods using the following constants:

<u>Description</u>	<u>Value</u>	
	<u>full scale</u>	<u>model scale</u>
reference area, S , ft 2	2690.0	0.2690
reference chord, \bar{c} , in.	474.8	4.748
reference span, b , in.	936.7	9.367
reference body length, ℓ_B , in	1290.3	12.903
moment reference point		
XMRP, in. X_0	1076.7	10.767
YMRP, in. Y_0	0.0	0.0
ZMRP, in. Z_0	375.0	3.75
base area, ft. 2	417.4	4.174

Mach 16 data were corrected for source and non-uniform flow as follows:

$$C_N = \text{measured } C_N [1 + \text{source correction} \pm \text{non-uniform correction}]$$

$$= \text{measured } C_N [1 + 0.01 \pm 0.0175].$$

$$X_{cp}/\ell = \text{measured } X_{cp}/\ell + \text{source correction} \pm \text{non-uniform correction}$$

DATA REDUCTION - Continued

$$= \text{measured } X_{cp}/\ell + 0.002 \pm 0.005$$

$$C_m = [\frac{X_{MRP}}{\ell_B} - X_{cp}/\ell] \frac{C_N \ell_B}{\bar{c}}$$

Mach 20 data contains an additional correction for contamination, computed as follows:

$$\dot{q}_0/\dot{q}(I) = \text{contamination factor}$$

= ratio of heat transfer rate measured or facility hemisphere-cylinder probe stagnation point to heat transfer rate inferred by measurements on shoulder of probe.

so that;

$$C_{Af_E} = \frac{C_{Af_M}}{1+F}$$

where,

C_{Af_M} = balance measured forebody axial-force coefficient

F = AEDC contamination correction factor

= function of $\dot{q}_0/\dot{q}(I)$ from figure 4

and;

$$\Delta C_A = C_{Af_M} - C_{Af_E}$$

C_N = measured $C_N [1 + (\text{source correction}) + (\text{non-uniform correction})]$

= measured $C_N [1 + 0.01 \pm 0.03]$

X_{cp}/ℓ = measured $X_{cp}/\ell + \text{source correction} + \text{non-uniform correction}$ ~~+ contamination correction~~

$$X_{cp}/\ell = \text{measured } X_{cp}/\ell + 0.002 + 0.006 - \frac{\Delta C_A}{C_{N_M}} \frac{\ell_c}{\ell_B}$$

DATA REDUCTION - Concluded

where

- ℓ_c = empirical vertical length from contamination axial force center of pressure to moment reference point
 = 1.0 inches model scale

$$C_m = \left[\frac{X_{MRP}}{\ell_B} - \frac{x_{cp}/\ell}{c} \right] C_N \frac{\ell_B}{c}$$

AEDC viscous parameter, which is derived and explained in reference 1, is computed as follows:

$$\bar{V}_\infty = \frac{M_\infty \sqrt{C_\infty}}{\sqrt{Re_\infty L}}$$

where

$$C_\infty = \frac{\mu_{wall} T_\infty}{\mu_\infty T_{wall}}$$

The NASA/LaRC viscous interaction parameter is defined as: *

$$\bar{V}'_\infty = \frac{M_\infty \sqrt{C'_\infty}}{\sqrt{Re_\infty L}}$$

where

$$C'_\infty = \left[\frac{T'}{T_\infty} \right]^K \left[\frac{T_\infty + 122.1 \times 10^{-(5/T_\infty)}}{T' + 122.1 \times 10^{-(5/T')}} \right]^J$$

with the Monaghan's empirical relationship (reference 6) given by:

$$\frac{T'}{T_\infty} = 0.468 + 0.532 \frac{T_w}{T_\infty} + 0.195 \frac{\gamma-1}{2} M_\infty^2$$

where

- T_∞ = Freestream static temperature, degrees Kelvin
 T_w = Wall temperature ($367^\circ K$), degrees Kelvin
 T' = Reference temperature, degrees Kelvin

and

K and J are empirical constants. For air, K = 0.5 and J = 1.0

* Equations below are based on temperature in $^\circ K$, but tab. data are $^\circ R$.

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1. Siler, L. G. and J. C. Coats, "Static-Stability and Axial-Force Characteristics of the NASA STS 0.010-Scale Model (51-0) of the Vehicle 4 Orbiter at Mach Numbers 16 and 20 (Test OA-81)," AEDC-DR-74-25 (March 15, 1974).
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TABLE I.

TABLE II. - DATASET/RUN NUMBER COLLATION SUMMARY

TEST: TA-81 AEDC

TEST : #A-81 AEDC		DATA SET/RUN NUMBER COLLATION SUMMARY										TEST RUN NUMBERS					
DATA SET IDENTIFIER	CONFIGURATION	SCND.	PARAMETERS/VALUES						MACH	NO. OF RUNS	ANGLE OF ATTACK - DEGS						
			α	β	Rn/V	Δe	Δr	ΔS_B	ΔS_E		20	25	30	35			
S ₁ RTΦ C01	(BCEFMN)(WE)(VR)	A C	.1	-46	C	.55	-11.7			16	4	4	-3	2	1		
	O2	A C	.15	-40	O	.55	-11.7			20	4	8	7	6	5		
	O3	B O	1.1	0	O	.55	-11.7			16	3	11	10	9			
	O4	A O	1.1	0	O	.55	0			16	4	1.5	14	13	12		
	O5	A O	.15	0	O	.55	C			20	4	19	18	17	16		
	O6	C O	.15	0	O	.55	0			20	2	48			21		
	O7	A O	.50	0	O	.55	0			16	4	25	24	23	22		
	O8	A O	.07	0	O	.55	0			20	4	29	28	27	26		
	O9	D O	.5	0	O	.55	C			16	1				30		
	O10	D O	.07	0	O	.55	0			20	2				49	31	
	O11	B C	1.1	0	O	.55	16.3			16	3				34	33	32
	O12	A C	1.1	15	O	.55	16.3			16	4				38	37	36
	O13	A C	.25	15	C	.55	16.3			20	4				42	41	40
	O14	(INVERTED) C C	1.1	0	O	.55	-11.7			16	2				43	44	
	O15	C C	.25	0	O	.25	-11.7			20	2				45	46	
	O16	D C	.5	0	O	.55	-11.7			16	1				47		

RSETS SSETS RSETS SSETS

TIME	CN	CLM	XCP/L	CL	CD	L/D	CAF	VILBAR	MACH	ALPHA	10
TIME	PΦ	P	Q(PSI)	TΦ	IRE-L	CPB	V-INF	VILBAR	MACH	ALPHA	19
TIME	CN	CLM	XCP/L	CL	CD	L/D	CAF	CAM	CAM	ALPHA	10
TIME	PΦ	P	Q(PSI)	TΦ	IRE-L	CPB	V-INF	VILBAR	C-ECCR	MACH	ALPHA

SCH	20°	25°	30°	35°
A	x	x	x	x
B		x	x	x
C		x		x
D				x

*REVISED 4/24/74
TABLE III. - MODEL DIMENSIONAL DATA

MODEL COMPONENT: BODY - B₂₆

GENERAL DESCRIPTION: Configuration 140A/B orbiter fuselage

NOTE: B₂₆ is identical to B₂₄ except underside of fuselage has been refaired to accept W₁₁₆.

MODEL SCALE: 0.010

MODEL DRAWING: SS-A00147, RELEASE 12.

DRAWING NUMBER

VL70-000143B, -000200, -000205, -006089, -000145,
VL70-000140A, -000140B

DIMENSION:

*Length (OML: Fwd Sta. X_O=235), In.

1293.3

12.933

*Length (IML: Fwd Sta. X_O=238), In.

1290.3

12.903

*Max Width (@ X_O = 1528.3), In.

264.0

2.640

Max Depth (@ X_O = 1464), In.

250.0

2.500

Fineness Ratio

Area - Ft²

Max Cross-Sectional

340.88

0.0341

Planform

Wetted

Base

TABLE III. - Continued.

*REVISED 4/24/74

MODEL COMPONENT: CANOPY - CGENERAL DESCRIPTION: Configuration 3A. Canopy used with fuselage B26.

MODEL SCALE: 0.010 MODEL DRAWING: SS-A00147, RELEASE 12

DRAWING NUMBER VL70-000143A

<u>DIMENSION:</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
* Length ($x_0 = 434.643$ to 578)	143.357	1.434
Max Width (@ $x_0 = 513.127$)	152.412	1.524
Max Depth (@ $x_0 = 485.0$)	25.000	0.250
Fineness Ratio		
Area		
Max Cross-Sectional		
Planform		
Wetted		
Base		

TABLE III. - Continued.

*REVISED 4/24/74

MODEL COMPONENT : ELEVON - E₂₆

GENERAL DESCRIPTION : Configuration 140A/B Orbiter elevons

Data are for one of two sides.

MODEL SCALE: 0.010

MODEL DRAWING: SS-A00148, RELEASE 6

DRAWING NUMBER VI70-000200, -006089, -006092

DIMENSIONS	FULL SCALE	MODEL SCALE
Area - Ft ²	210.00	0.0210
Span (equivalent) In.	349.2	3.492
Inb'd equivalent chord, In.	118.004	1.1180
Outb'd equivalent chord , In.	55.192	0.552
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	0.2096	0.2096
At Outb'd equiv. chord	0.4004	0.4004
Sweep Back Angles, degrees		
Leading Edge	0.00	0.00
Trailing Edge	- 10.056	- 10.056
Hingeline (Product of area & c)	0.00	0.00
* Area Moment (Normal-to-hinge-line), Ft ³	1528.25	0.0015
*Mean Aerodynamic Chord, In.	90.7	0.907

TABLE III. - Continued.

MODEL COMPONENT: BODY FLAP - F₇GENERAL DESCRIPTION: Configuration 140A/B Orbiter body flap

MODEL SCALE: 0.010 MODEL DRAWING: SS-A00147, RELEASE 12

DRAWING NUMBER VL70-000140A, VL70-000145

<u>DIMENSION:</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Length ($X_0 = 1520$ to $X_0 = 1613$), In.	93.000*	0.930
Max Width, In.	262.00	2.620
Max Depth ($X_0 = 1520$), In.	23.00	0.230
Fineness Ratio		
Area - Ft ²		
Max Cross-Sectional		
Planform	142.6	0.143
Wetted		
Base		

*Model dim. measured from model sta. 15.20

TABLE III. - Continued.

MODEL COMPONENT : OMS/RCS PODS - M₇GENERAL DESCRIPTION : Configuration 140A/B Orbiter OMS/RCS podsMODEL SCALE: 0.010 MODEL DRAWING: SS-A00147, RELEASE 12DRAWING NUMBER: VL70-000145

DIMENSIONS	FULL SCALE	MODEL SCALE
Length (OMS Fwd Sta $X_O = 1233.0$), In.	<u>327.000</u>	<u>3.270</u>
Max Width (@ $X_O = 1450.0$), In.	<u>94.50</u>	<u>0.945</u>
Max Depth (@ $X_O = 1493.0$), In.	<u>109.000</u>	<u>1.090</u>
Fineness Ratio	_____	_____
Area	_____	_____
Max. Cross-Sectional	_____	_____
Planform	_____	_____
Wetted	_____	_____
Base	_____	_____

TABLE III. - Continued.

MODEL COMPONENT: NOZZLES - N₂₈

GENERAL DESCRIPTION: Configuration 140A/B Orbiter OMS Nozzles

MODEL SCALE: 0.010 MODEL DRAWING: SS-A00106, RELEASE 5 (Contour)

DRAWING NUMBER: VL70-000140A (Location)

DIMENSIONS: FULL SCALE MODEL SCALE

MACH NO.

Length - In.

Gimbal Point to Exit Plane
Throat to Exit Plane

Diameter - In.

Exit
Throat
Inlet Area - ft²Exit
Throat

Gimbal Point (Station) - In.

Upper Nozzle Left Nozzle, In.

X	1518.00	15.180
Y	- 88.0	- 0.880
Z	492.00	4.920

Right
Lower Nozzles

X	1518.00	15.180
Y	88.0	0.880
Z	492.0	4.920

Null Position - Deg.

Upper Nozzle Left Nozzle:

Pitch	15°49'	15°49'
Yaw	12°17'	12°17'

Right

Lower Nozzle

Pitch	15°49'	15°49'
Yaw	12°17'	12°17'

TABLE III. - Continued.

*REVISED 4/24/74

MODEL COMPONENT: RUDDER - R₅GENERAL DESCRIPTION: Configuration 140C orbiter rudder (identical to Configuration 140A/B Rudder).MODEL SCALE: 0.010DRAWING NUMBER: VL70-000146B, VL70-000095

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
* Area - Ft ²	<u>100.15</u>	<u>.010</u>
Span (equivalent) , In.	<u>201.0</u>	<u>2.010</u>
Inb'd equivalent chord, In.	<u>91.585</u>	<u>0.916</u>
Outb'd equivalent chord , In.	<u>50.833</u>	<u>0.508</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
At Outb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>34.83</u>	<u>34.83</u>
Trailing x ading Edge	<u>26.25</u>	<u>26.25</u>
Hingeline (Product of Area & c)	<u>34.83</u>	<u>34.83</u>
* Area Moment (Normal to hinge line), Ft ³	<u>610.92</u>	<u>0.00061</u>
*Mean Aerodynamic Chord, In.	73.2	0.732

TABLE III. - Continued.

*REVISED 6/1/74

MODEL COMPONENT: VERTICAL - V₈

GENERAL DESCRIPTION: Configuration 140C orbiter vertical tail (identical to configuration 140A/B vertical tail).

MODEL SCALE: 0.010

DRAWING NUMBER: VL70-000140C, VL70-000146B

DIMENSIONS:

FULL SCALE MODEL SCALE

TOTAL DATA

Area (Theo) - Ft ²		
Planform	413.253	.04133
Span (Theo) - In.	315.72	3.157
Aspect Ratio	1.675	1.675
Rate of Taper	0.507	0.507
Taper Ratio	0.404	0.404
Sweep-Back Angles, Degrees.		
Leading Edge	45.000	45.000
Trailing Edge	26.25	26.25
0.25 Element Line	41.13	41.13

Chords:

Root (Theo) WP	268.50	2.685
Tip (Theo) WP	108.47	1.085
MAC	199.81	1.998
Fus. Sta. of .25 MAC	1463.35	14.634
W.P. of .25 MAC	635.52	6.355
B.L. of .25 MAC	0.00	0.00

Airfoil Section

Leading Wedge Angle - Deg.	10.00	10.00
Trailing Wedge Angle - Deg.	14.92	14.92
Leading Edge Radius	2.00	0.020

Void Area

13.17 0.0013

Blanketed Area

0.00 0.00

TABLE III. - Concluded.

*REVISED 4/24/74

MODEL COMPONENT: WING-W₁₁₆

GENERAL DESCRIPTION: Configuration 4

NOTE: Identical to W₁₁₄, except airfoil thickness. Dihedral angle is along trailing edge of wing.

TEST NO.	DWG. NO.	VL70-000140A, -000200
DIMENSIONS:	FULL-SCALE	MODEL SCALE
TOTAL DATA		
Area (Theo.) Ft ²		
Planform	2690.00	0.2690
Span (Theo) In.	936.68	9.367
Aspect Ratio	1.177	1.177
Rate of Taper	0.200	0.200
Taper Ratio	3.500	3.500
Dihedral Angle, degrees	0.500	0.500
Incidence Angle, degrees	+ 3.000	+ 3.000
Aerodynamic Twist, degrees		
Sweep Back Angles, degrees		
Leading Edge	45.000	45.000
Trailing Edge	- 10.056	- 10.056
0.25 Element Line	35.209	35.209
Chords:		
Root (Theo) B.P.O.O.	689.24	6.892
Tip, (Theo) B.P.	137.85	1.379
MAC	474.81	4.748
* Fus. Sta. of .25 MAC	1136.83	11.368
* W.P. of .25 MAC	290.58	2.906
* B.L. of .25 MAC	182.13	1.821
EXPOSED DATA		
* Area (Theo) Ft ²	1751.50	0.175
* Span, (Theo) In. BP108	720.68	7.207
* Aspect Ratio	2.059	2.059
Taper Ratio	0.245	0.245
Chords		
* Root BP108	562.09	5.621
* Tip 1.00 b 2	137.85	1.379
* MAC	392.83	3.928
* Fus. Sta. of .25 MAC	1185.98	11.860
* W.P. of .25 MAC	294.30	2.943
* B.L. of .25 MAC	251.77	2.518
Airfoil Section (Rockwell Mod NASA) XXXX-64		
Root b = 2	0.113	0.113
Tip b = 2	0.120	0.120
Data for (1) or (2) Sides		
Leading Edge Cuff * Planform Area Ft ²	113.18	0.0113
* Leading Edge Intersects Fus M. L. @ Sta	500.00	5.000
* Leading Edge Intersects Wing @ Sta	1024.00	10.240

Notes:

- Positive directions of force coefficients, moment coefficients, and angles are indicated by arrows
- For clarity, origins of wind and stability axes have been displaced from the center of gravity

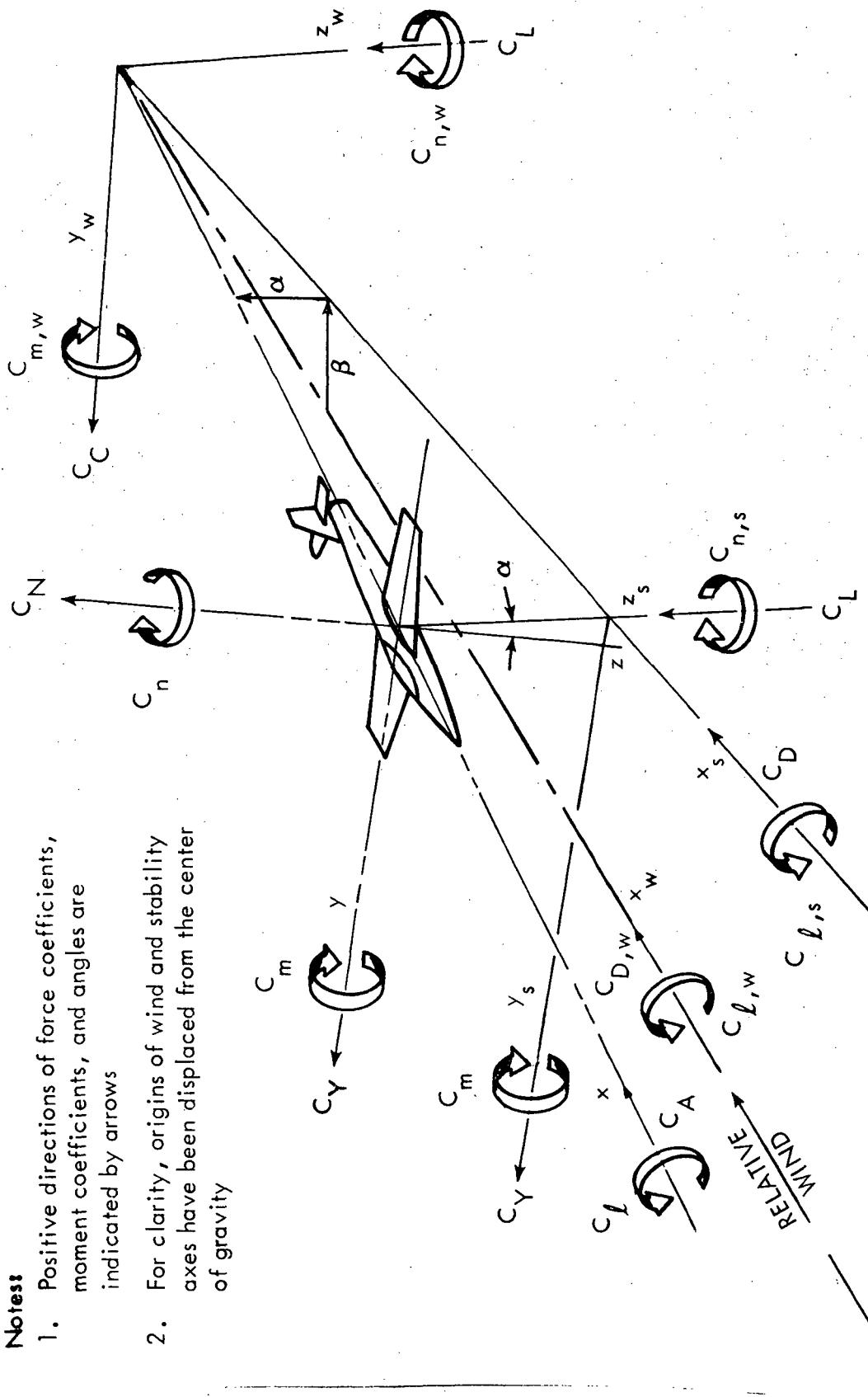


Figure 1.- Axis Systems.

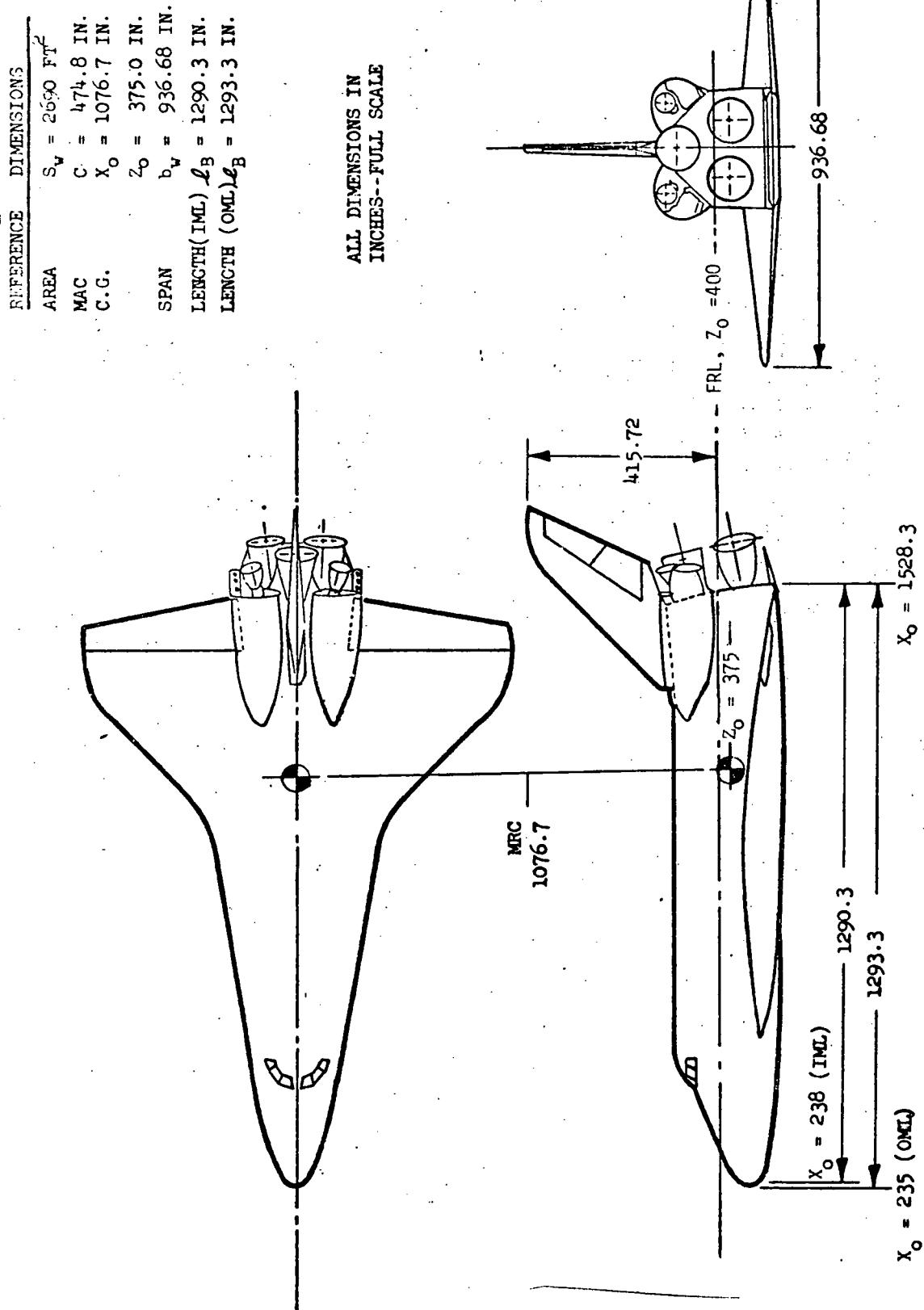
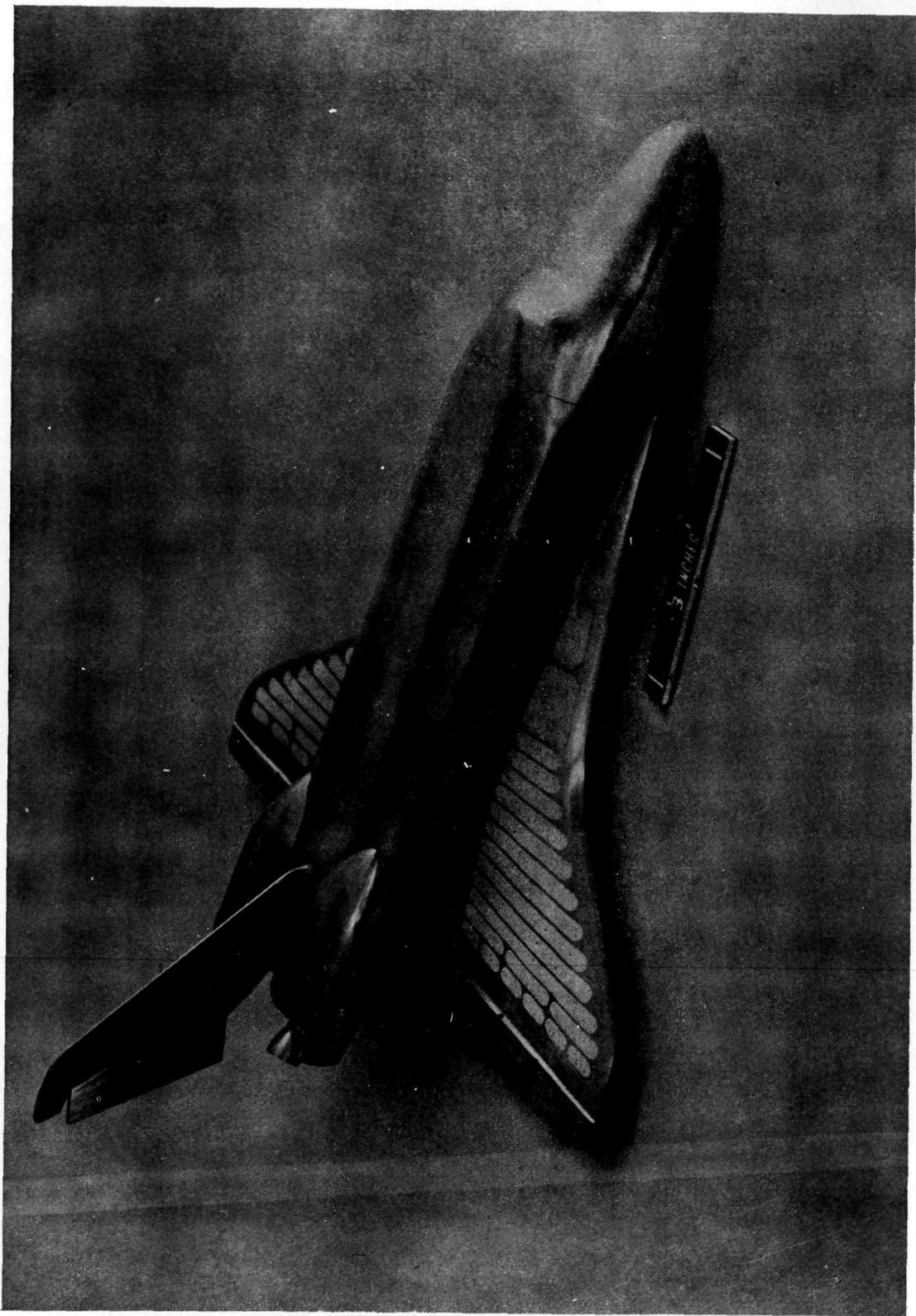


Figure 2. - Model sketch.

Figure 3. Model Photograph



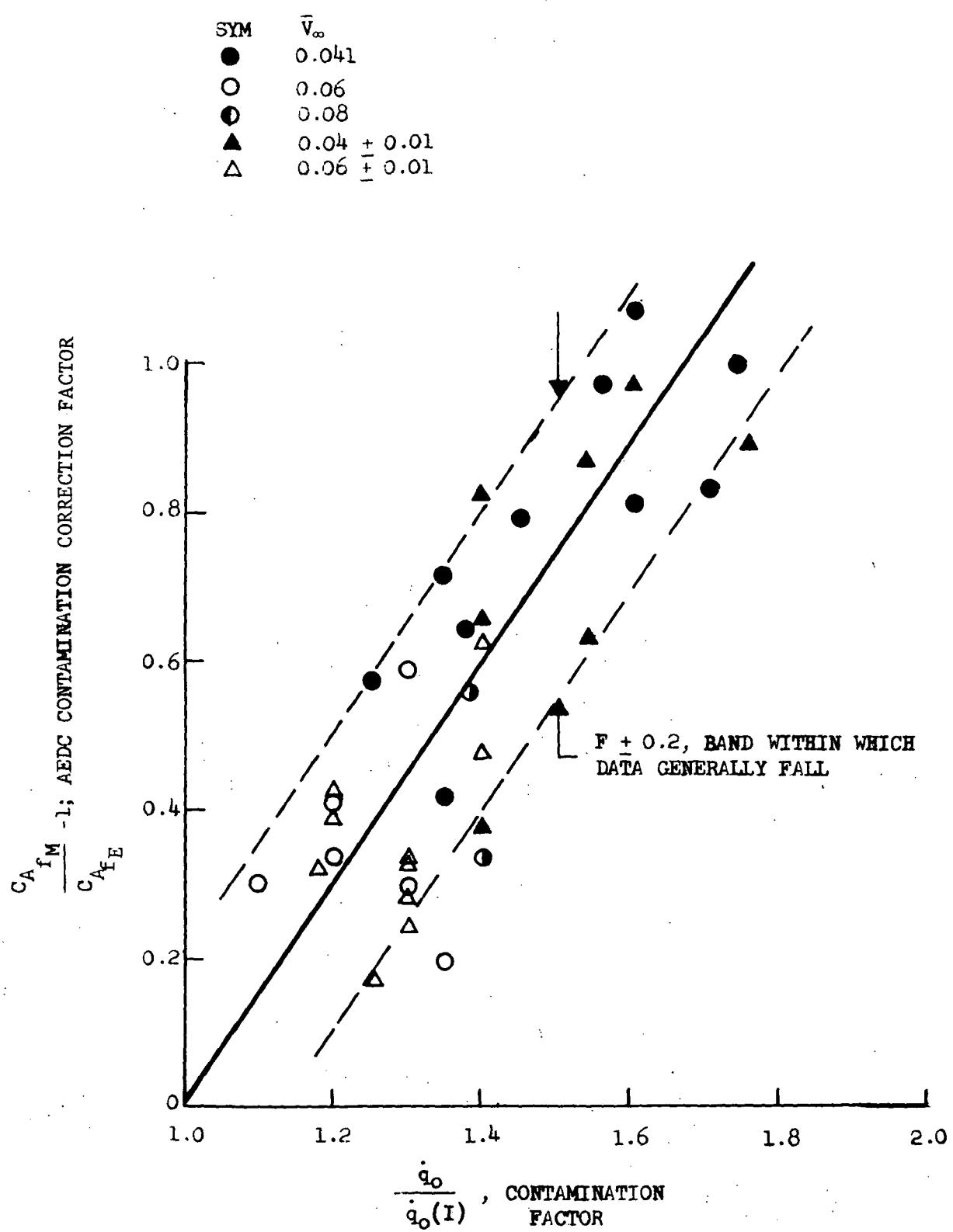
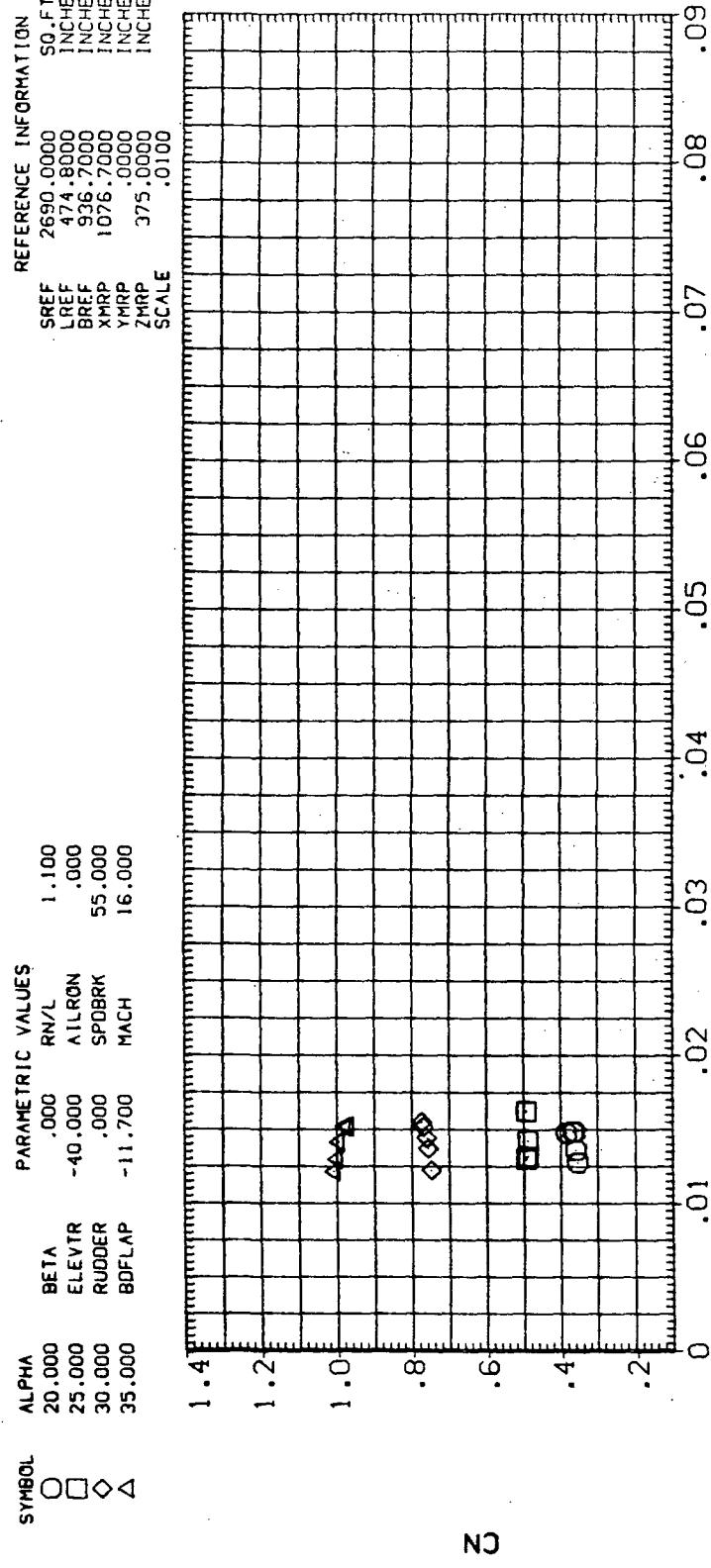
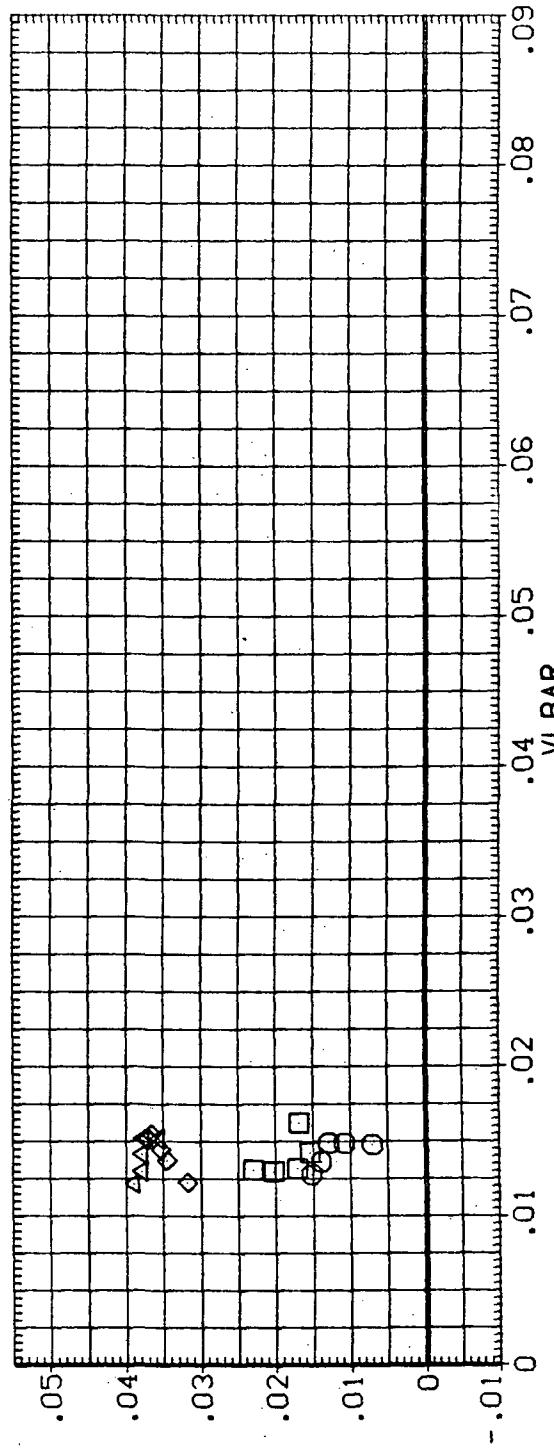


Figure 4. - AEDC Contamination Correction Factor.

AEDC VAA489(0A-81). (B26C9F7M7N28)(W116E26)(V8RS5)(FT0001)



C_z



CLM

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5) (FT0001)

SYMBOL	ALPHA	PARAMETRIC VALUES			REFERENCE INFORMATION
		BETA	RN/L	1.100	
○	20.000	.000	AIRON	.000	SREF 2690.0000 LREF 474.8000 BREF 936.7000 XMRP 1076.7000 YMRP .0000 ZMRP 375.0000 SCALE .0100
□	25.000	-40.000	SPDBRK	55.000	
◊	30.000	.000	MACH	16.000	
△	35.000	BDFLAP	-11.700		

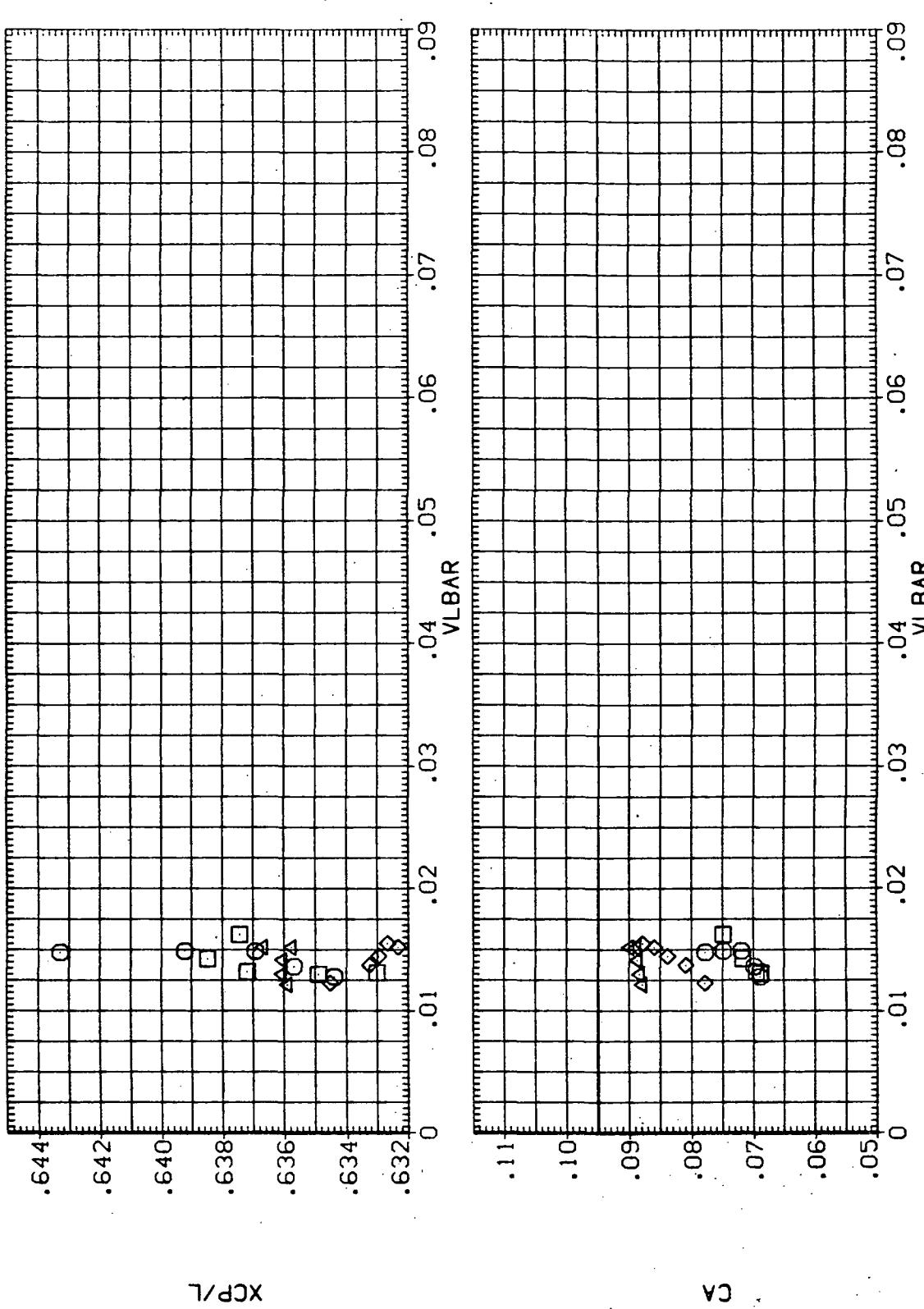


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VA489(0A-81). (B26C9F7M7N28)(W116E26)(V8R5) (FT0001)

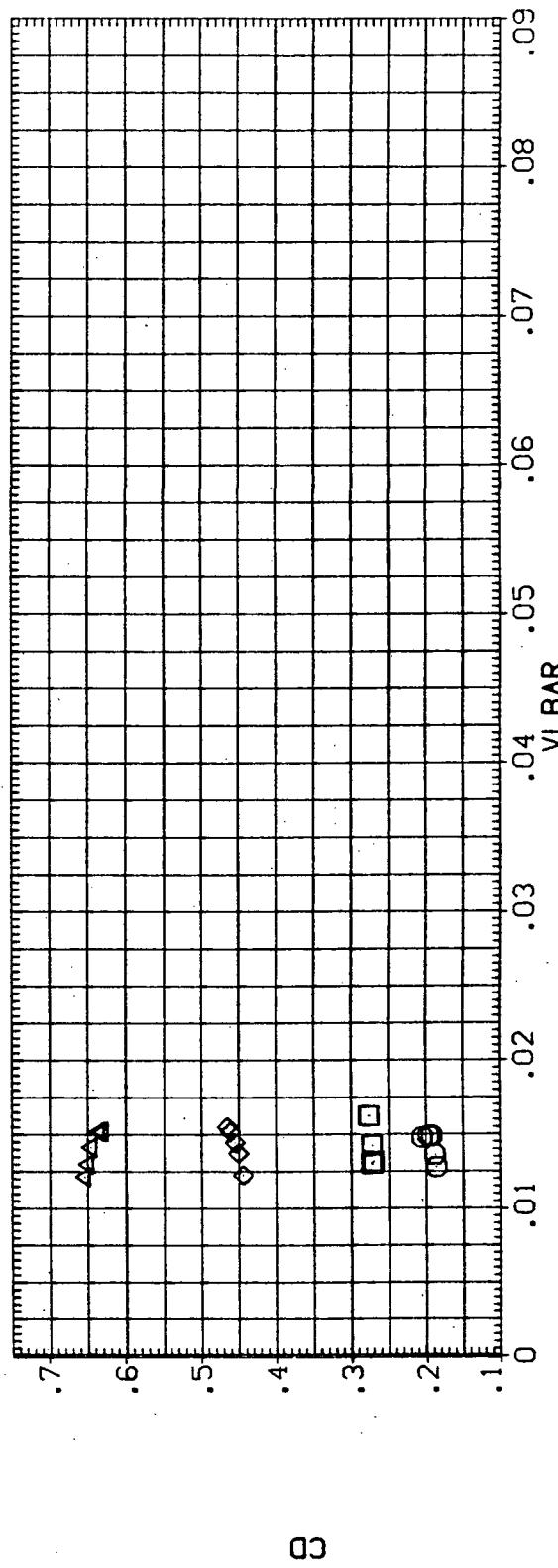
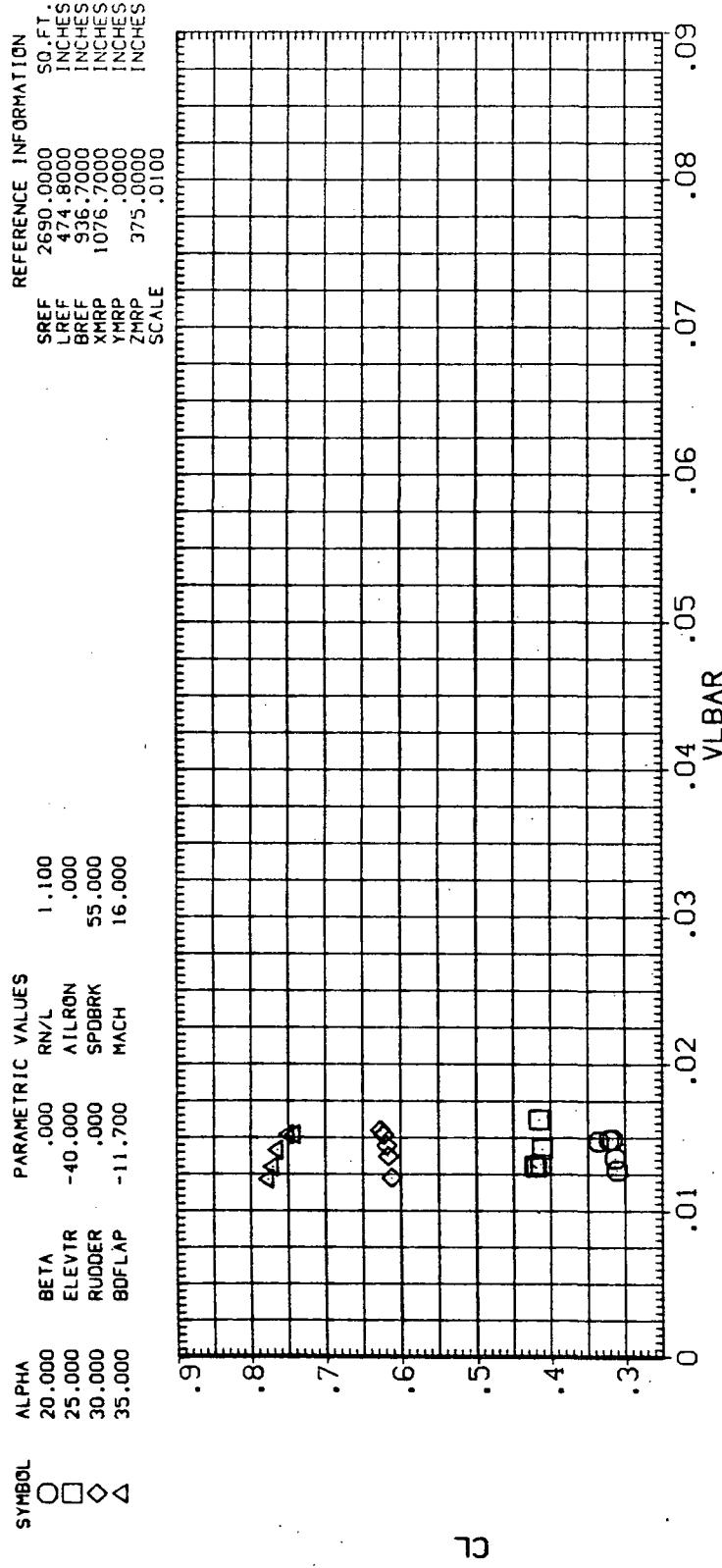


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0002)

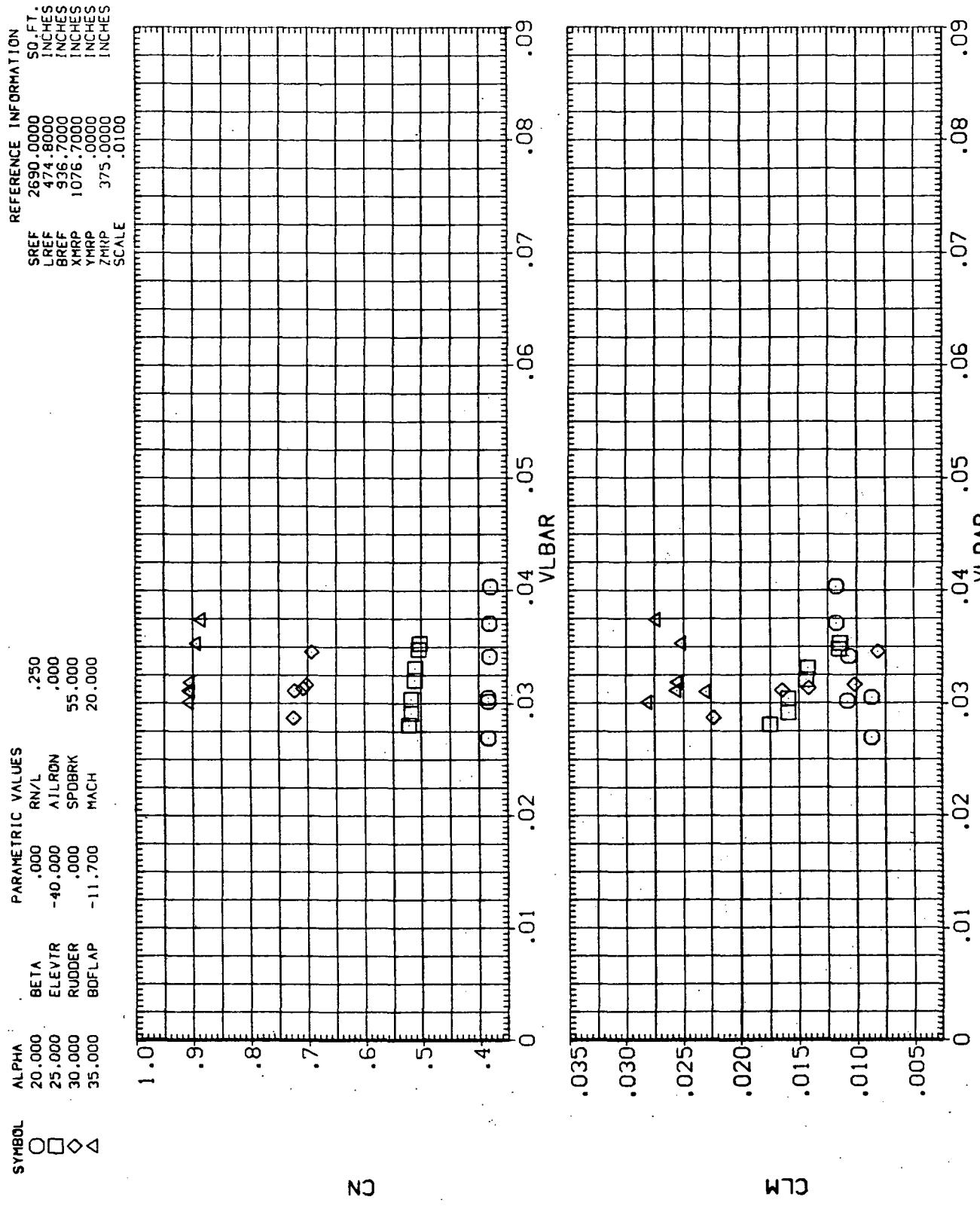


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VAA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5) (FT0002)

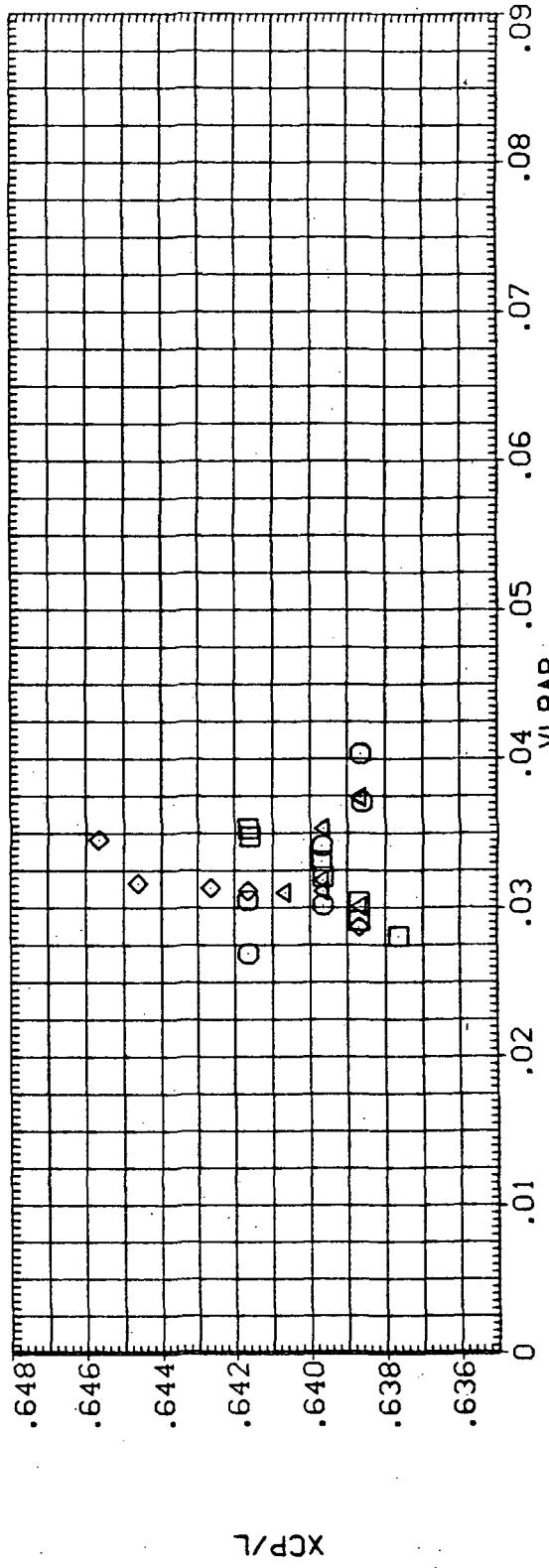
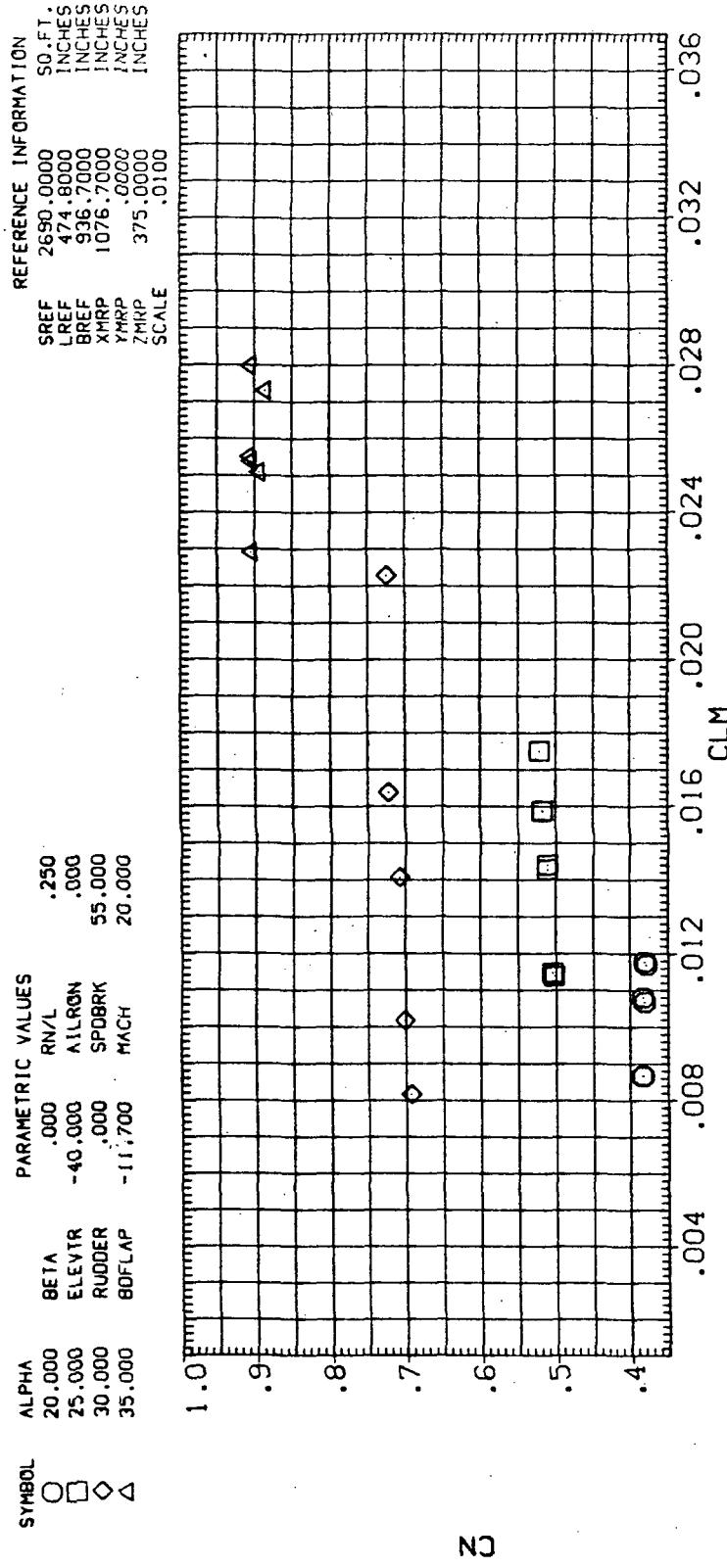


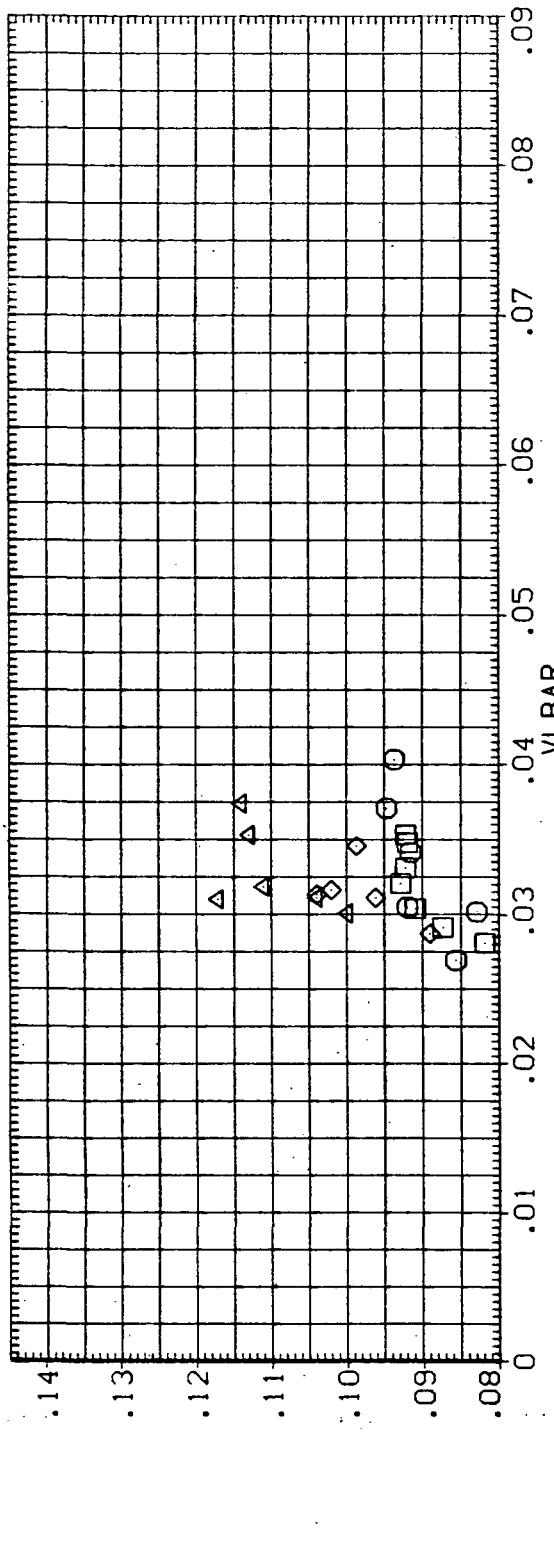
FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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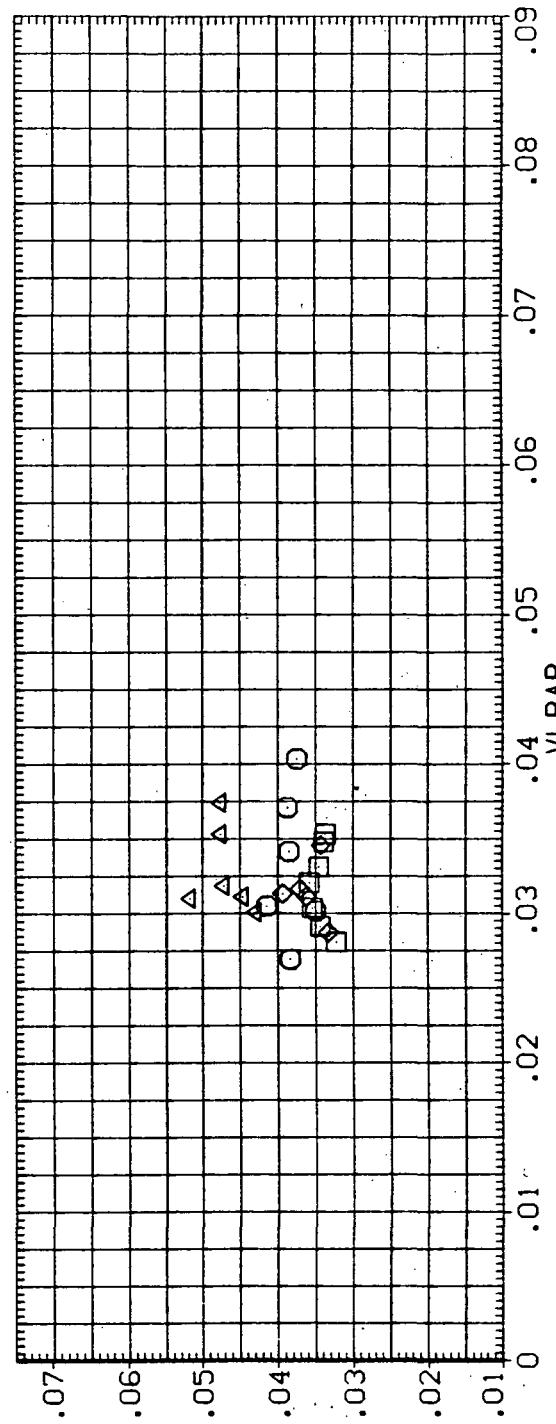
AEDC V489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5) (FT0002)

SYMBOL	ALPHA	BETA	RN/L	.250
○	20.000	.000	AIRON	.000
□	25.000	-40.000	SPDBRK	55.000
◊	30.000	.000	MACH	20.000
△	35.000	-11.700		

REFERENCE INFORMATION	
SREF	2690.0000
LREF	.474-.8000
BREF	.936-.7000
XMRP	1076-.7000
YMRP	.0000
ZMRP	.375-.0000
SCALE	.0100



CAE



CAE

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

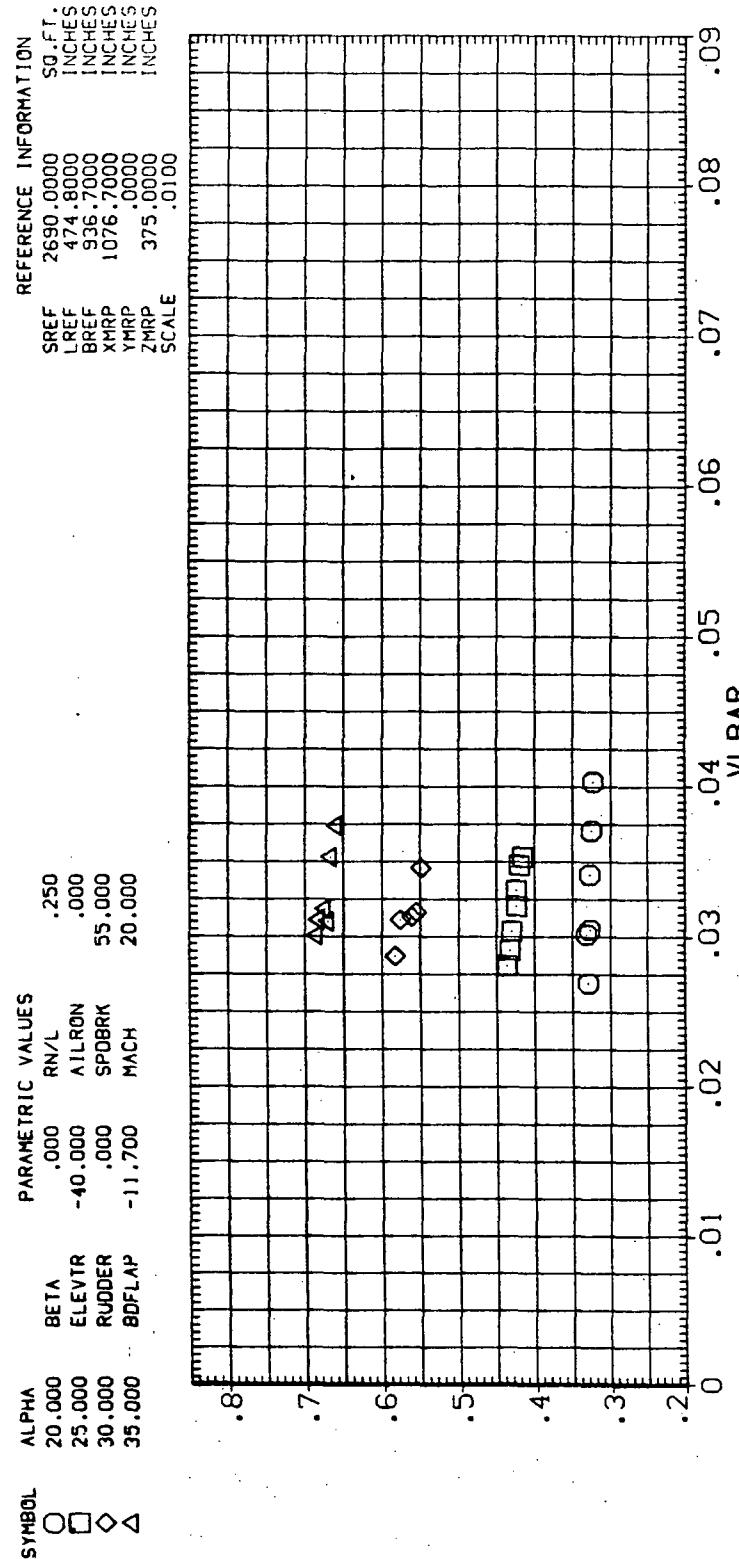
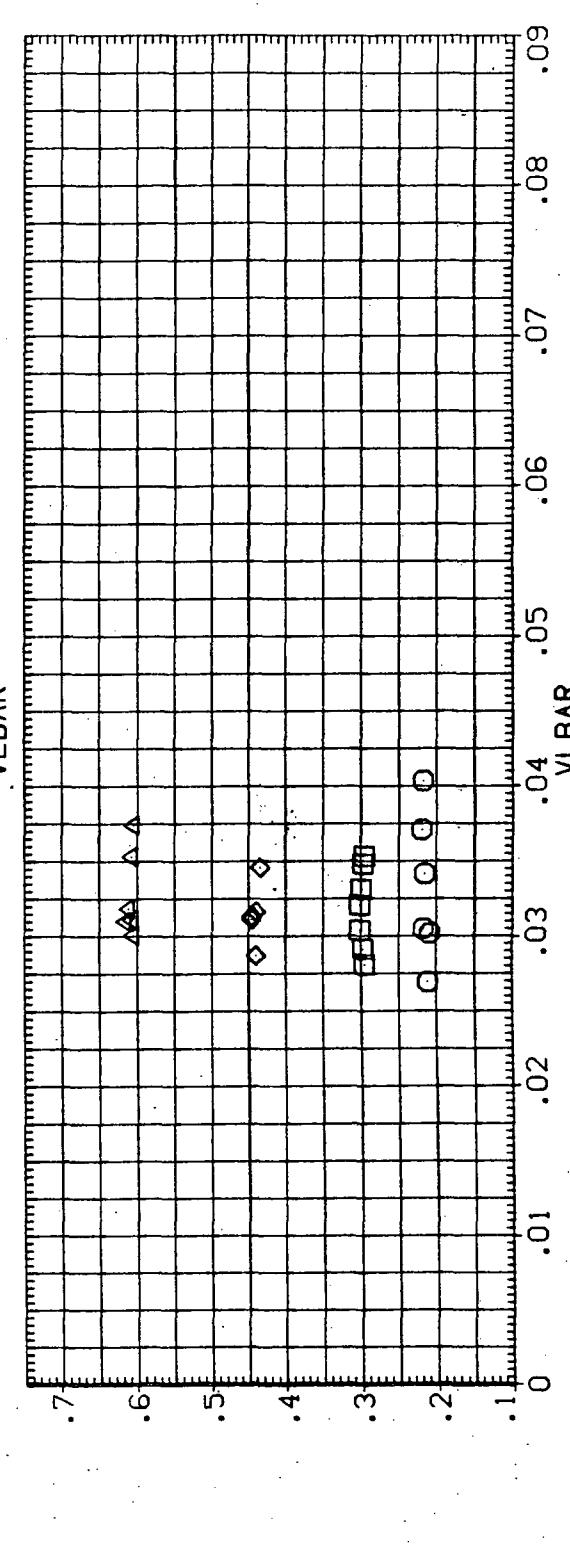
 C_L  C_L

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VAA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0003)

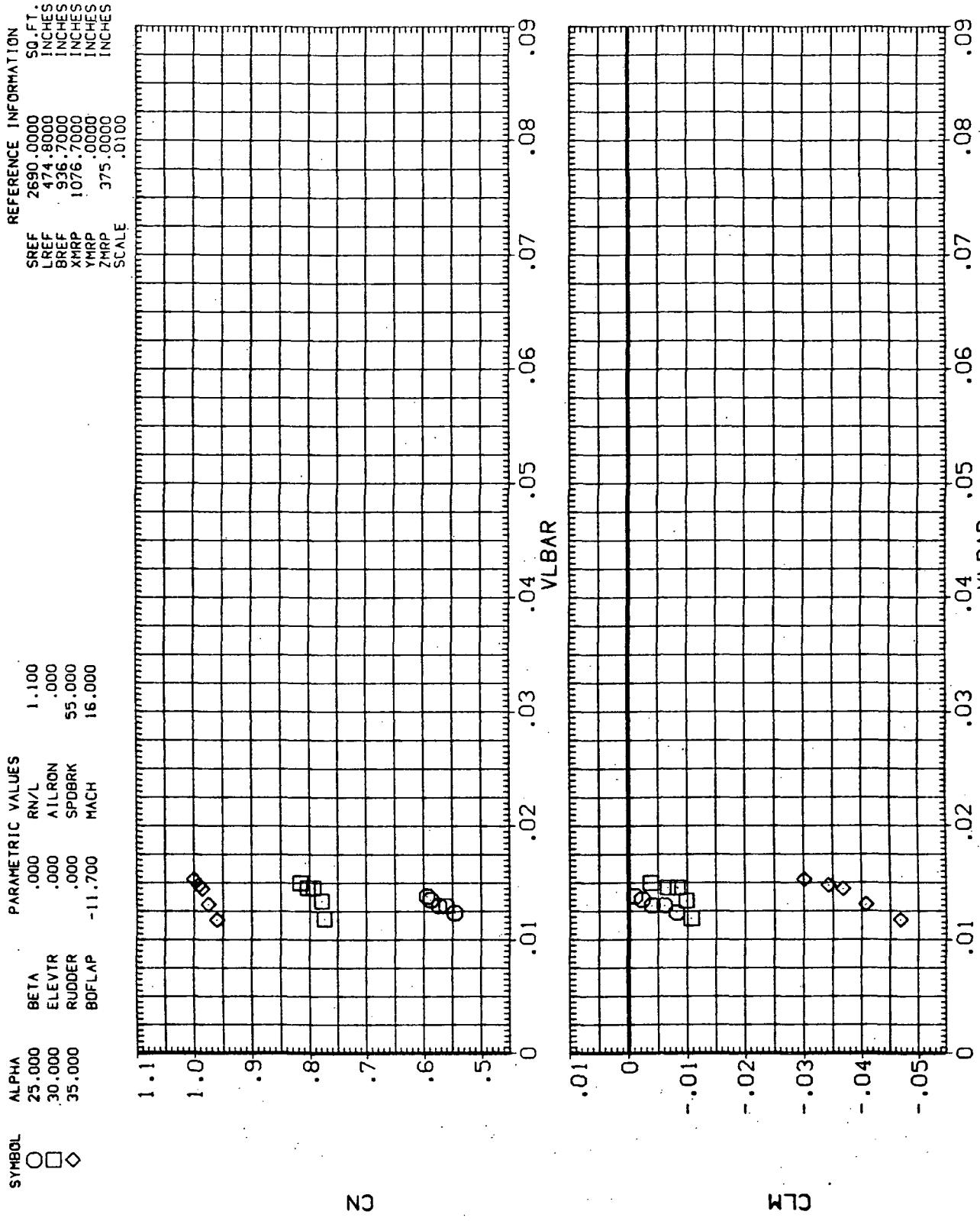


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

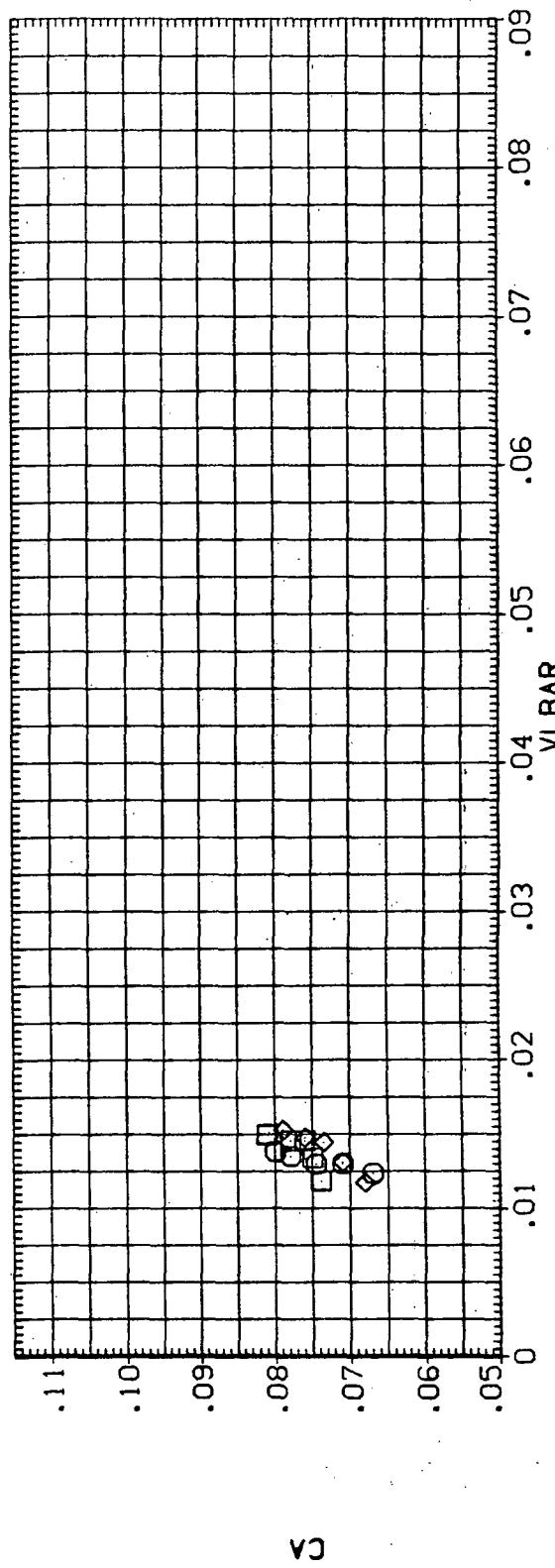
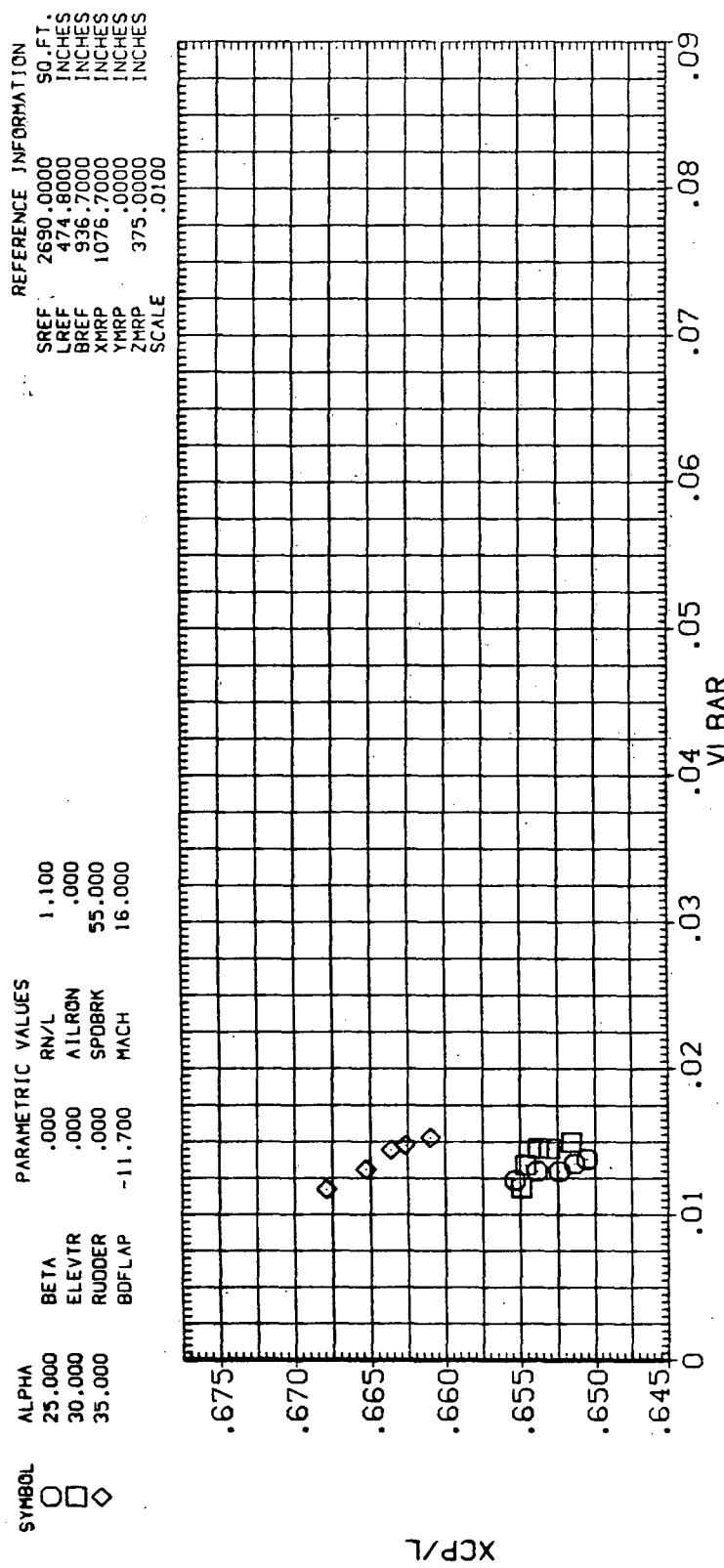


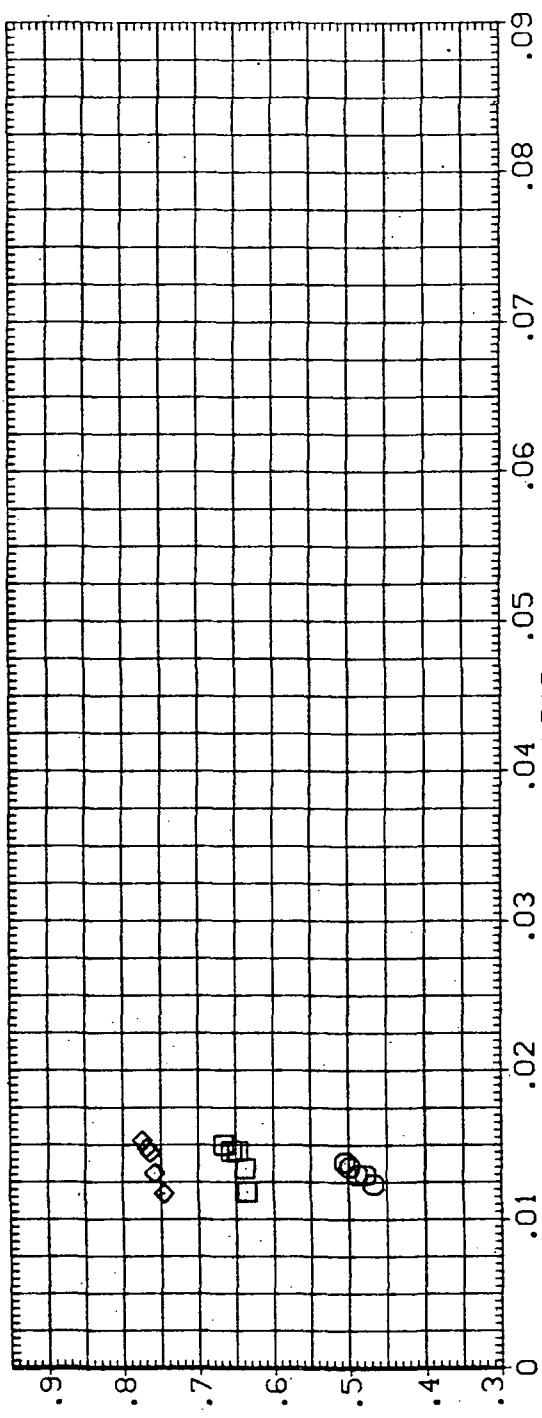
FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

PAGE

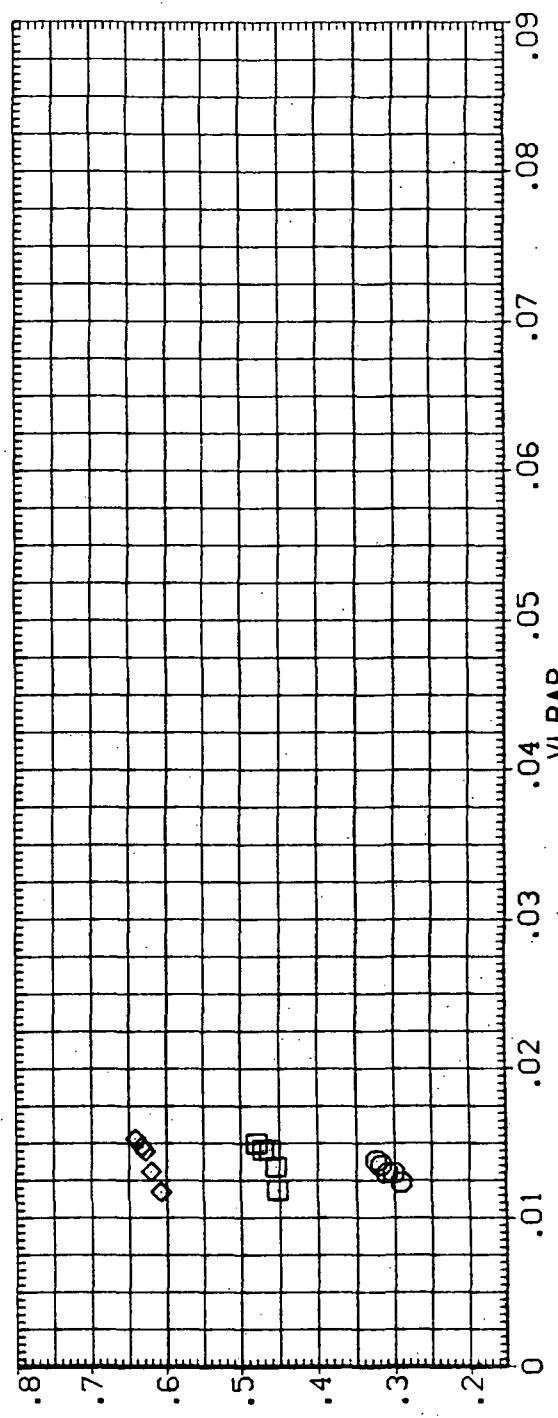
42

AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0003)

SYMBOL	ALPHA	PARAMETRIC VALUES			REFERENCE INFORMATION
		BETA	RNL	1.100	
○	25.000	.000	AILRON	.000	SREF 2690.0000 LREF 474.8000 BREF INCHES XMRP 936.7000 YMRP 1076.7000 ZMRP INCHES SCALE .0100
□	30.000	.000	SPOBRK	.55.000	
◊	35.000	.000	MACH	.16.000	
		BOFLAP	-11.700		



C3



C6

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

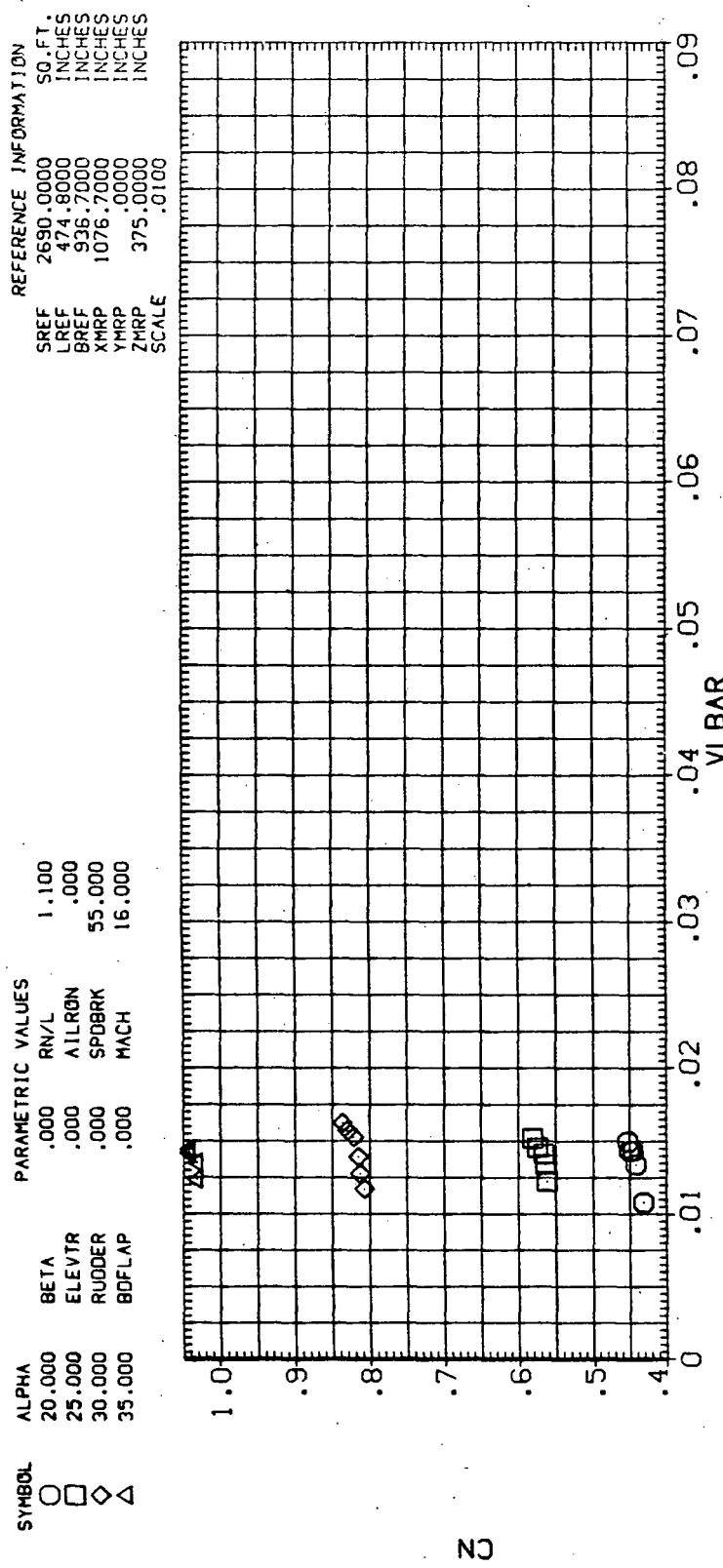
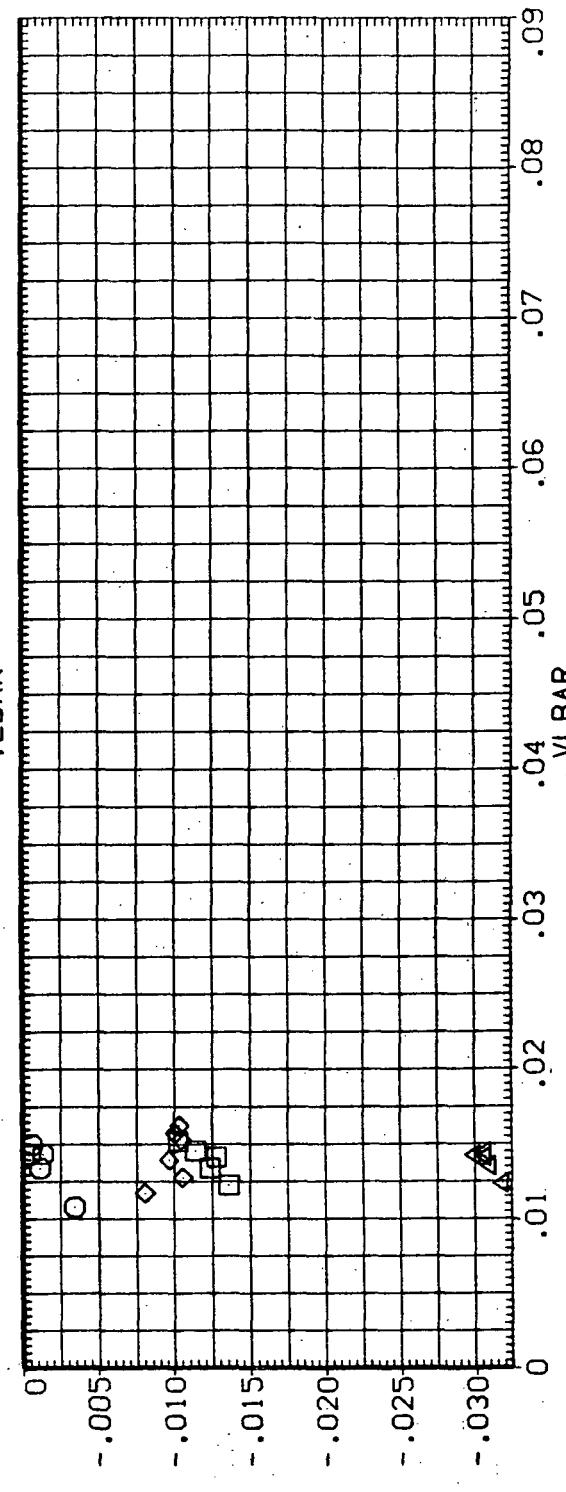
 C_n  C_L

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VAA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0004)

SYMBOL	ALPHA	PARAMETRIC VALUES			REFERENCE INFORMATION S0, FT. SREF 2690.0000 LREF 474.8000 BREF 936.7000 XMRP 1076.7000 YMRP 375.0000 ZMRP .0000 SCALE .0100
		BETA	RN/L	AILRON	
○	20.000	.000	.000	.000	
□	25.000	.000	.000	.000	
◊	30.000	.000	.000	.000	
△	35.000	.000	.000	.000	

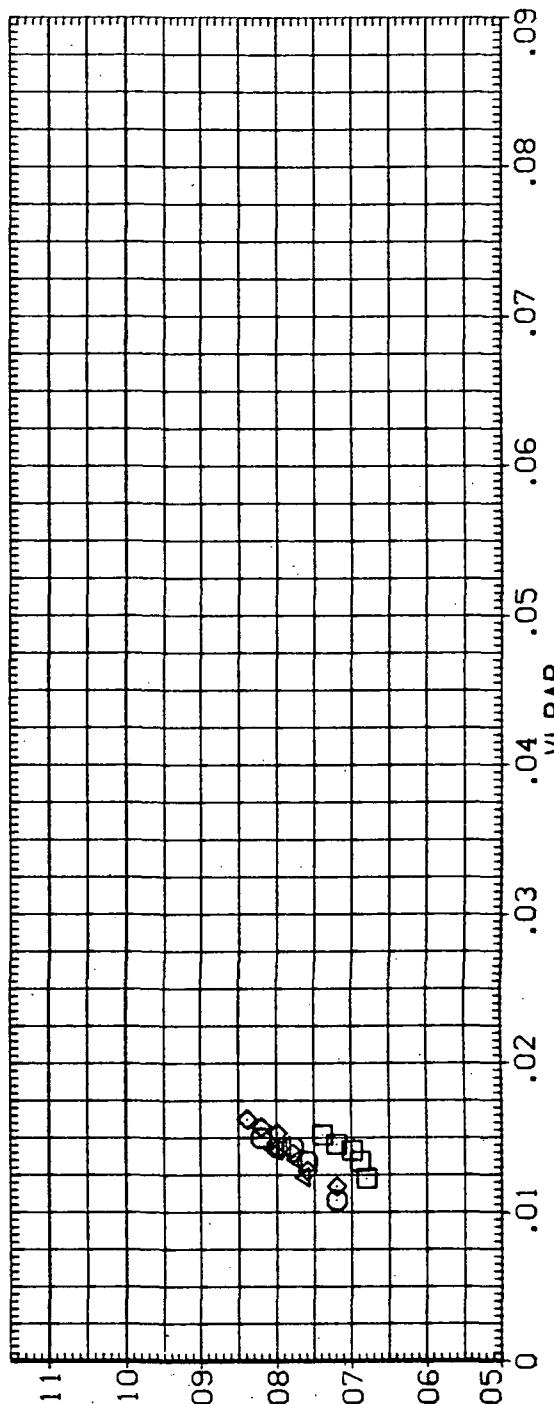
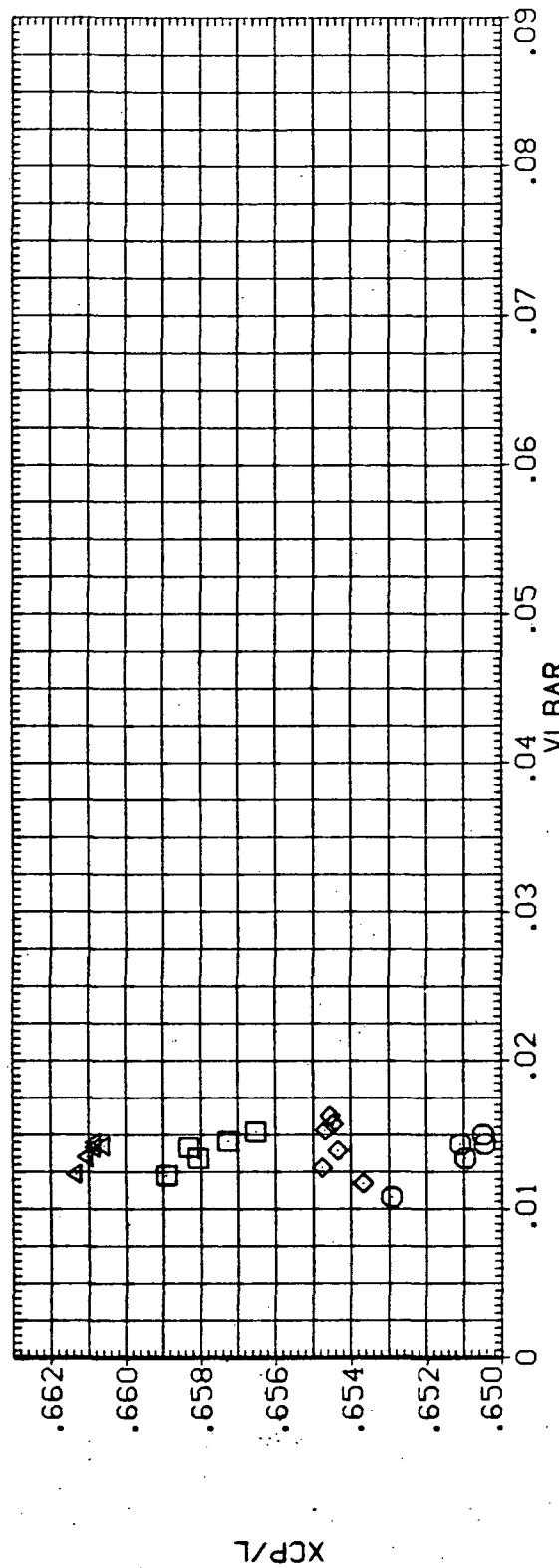
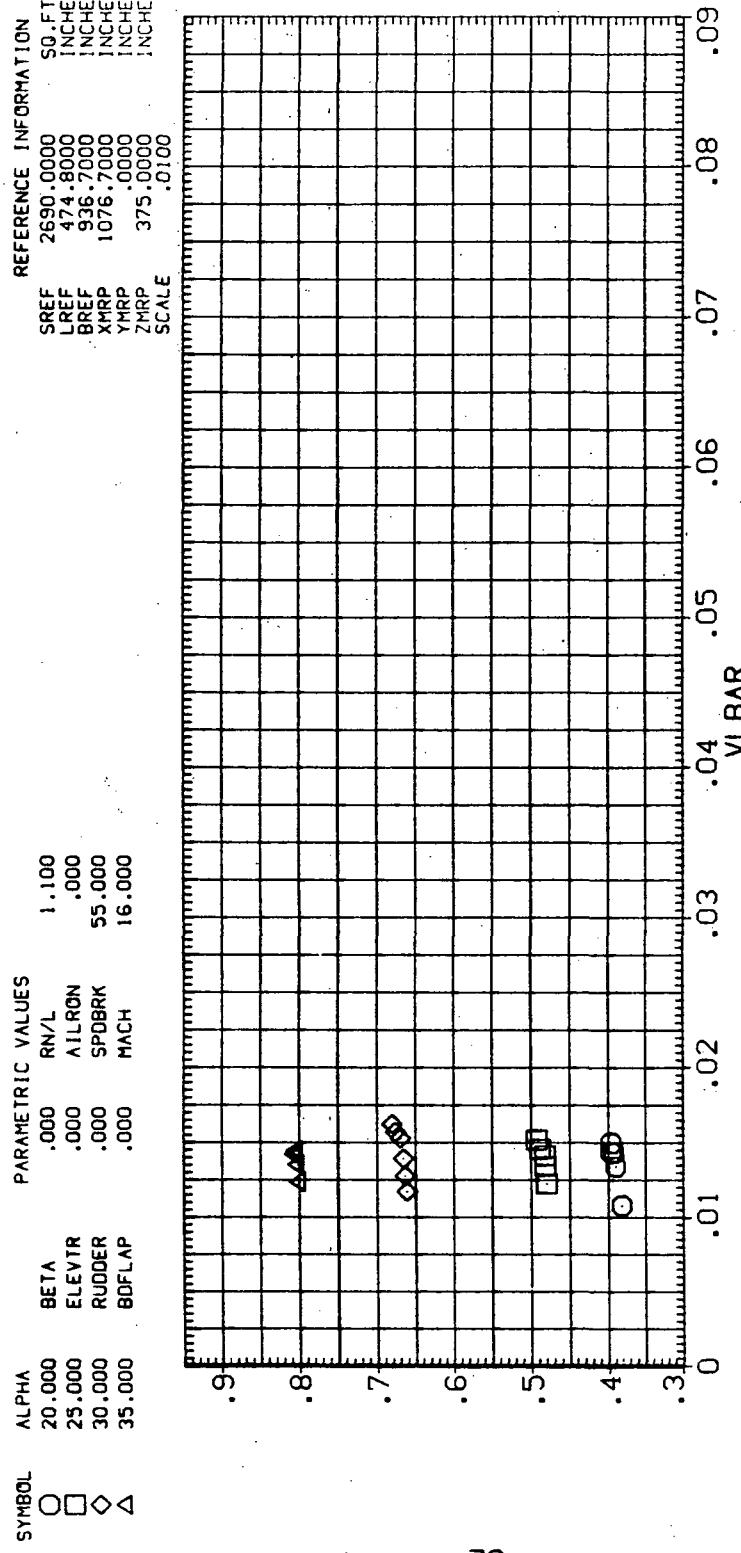
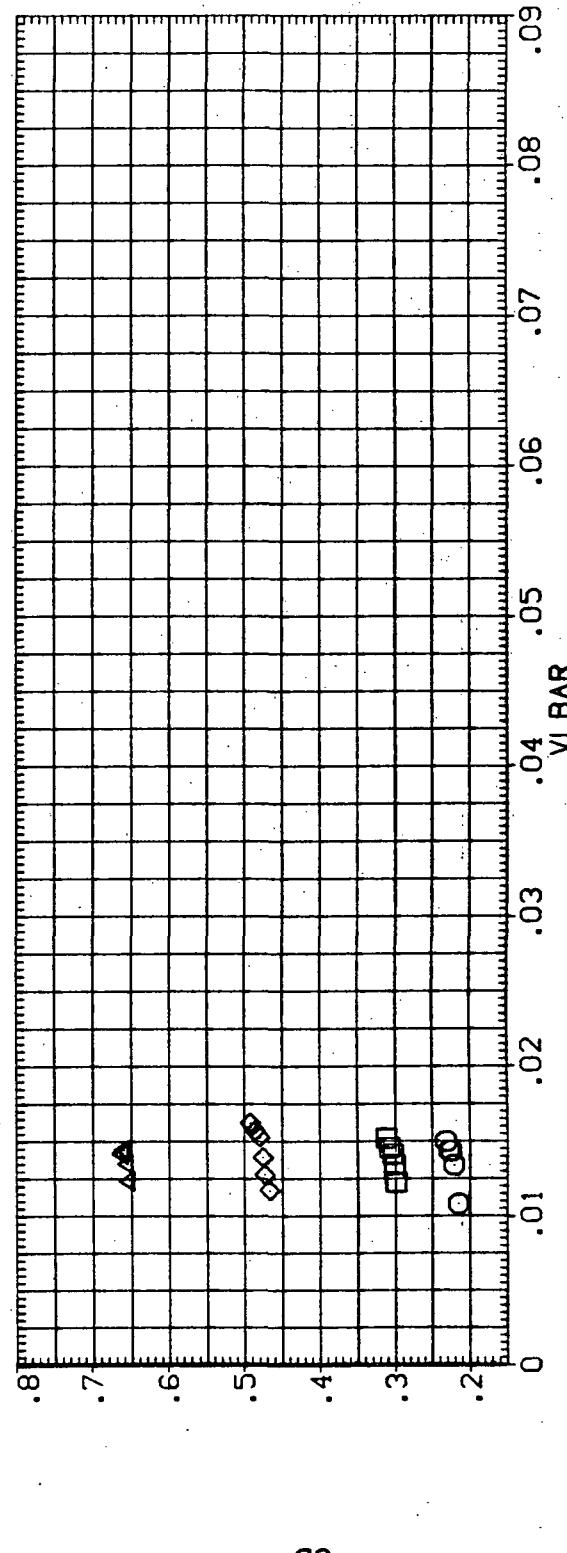


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B



CL



CG

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5) (FT0005)

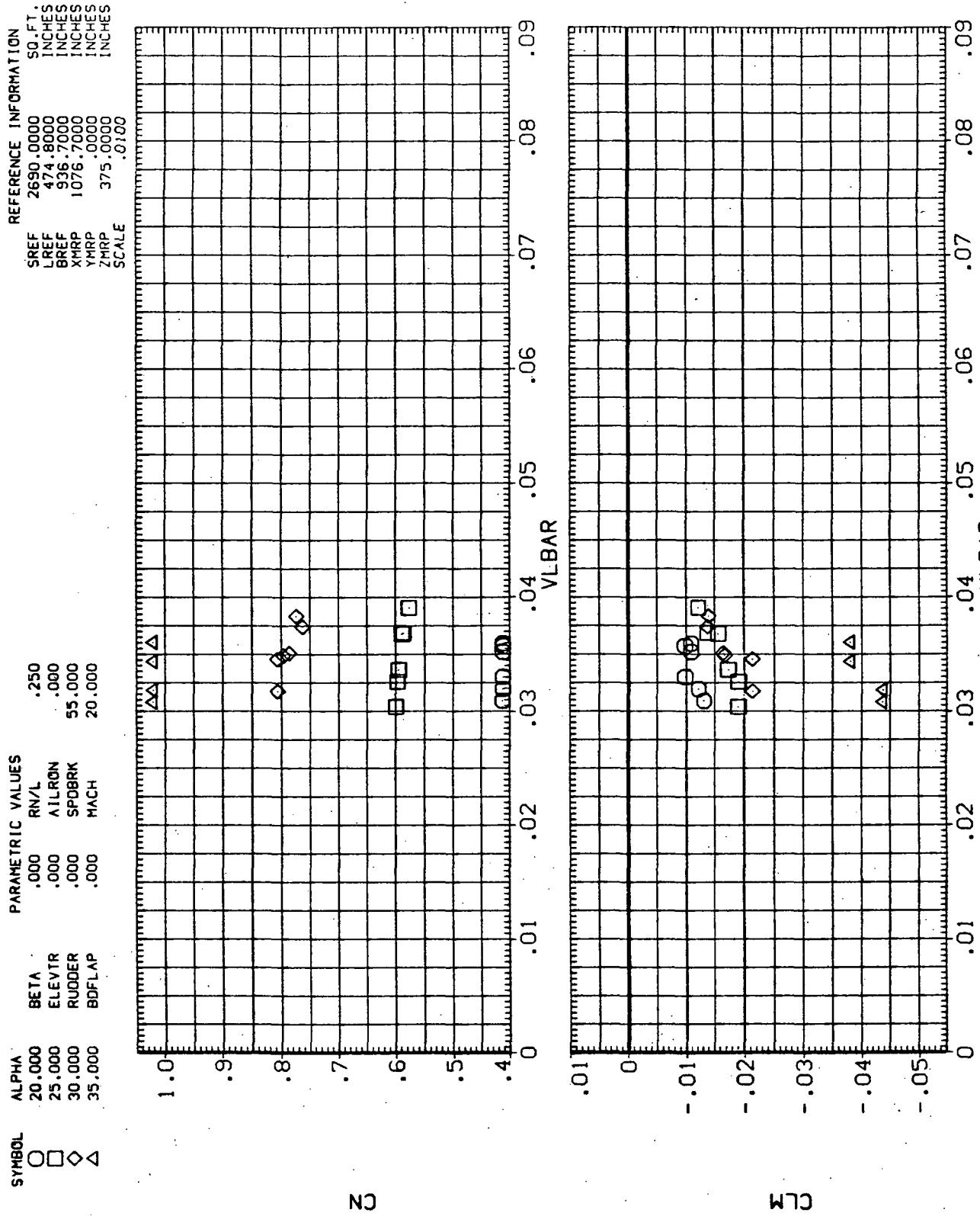


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0005)

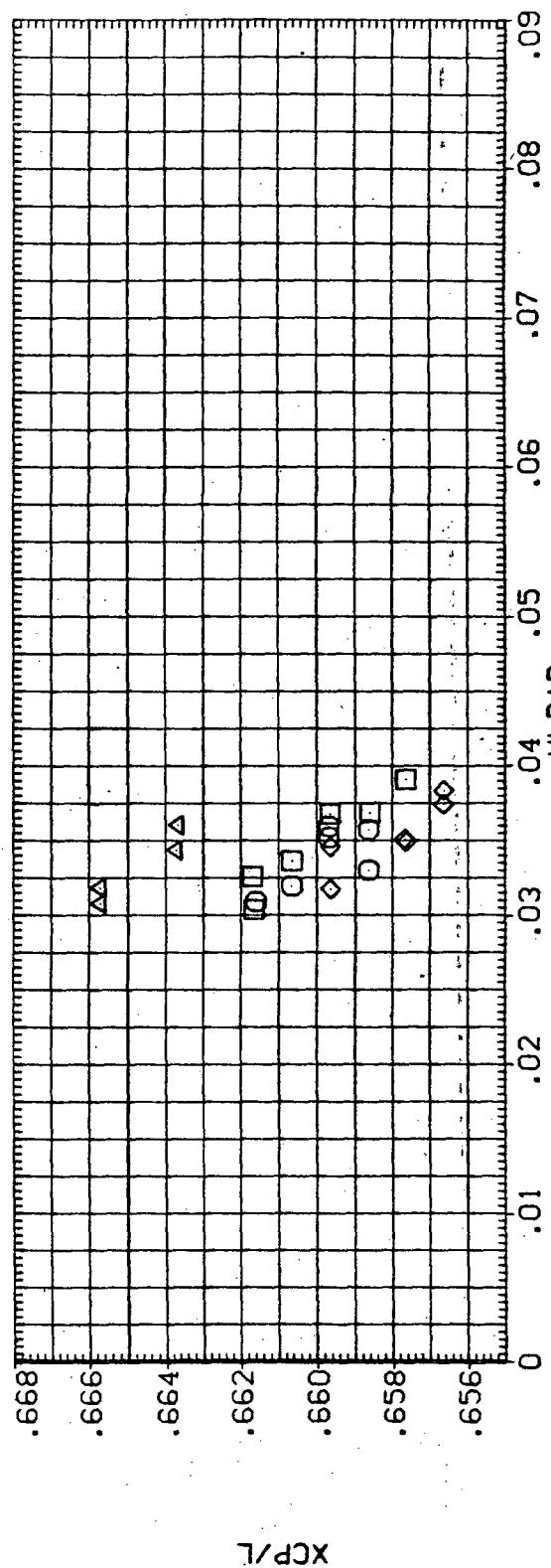
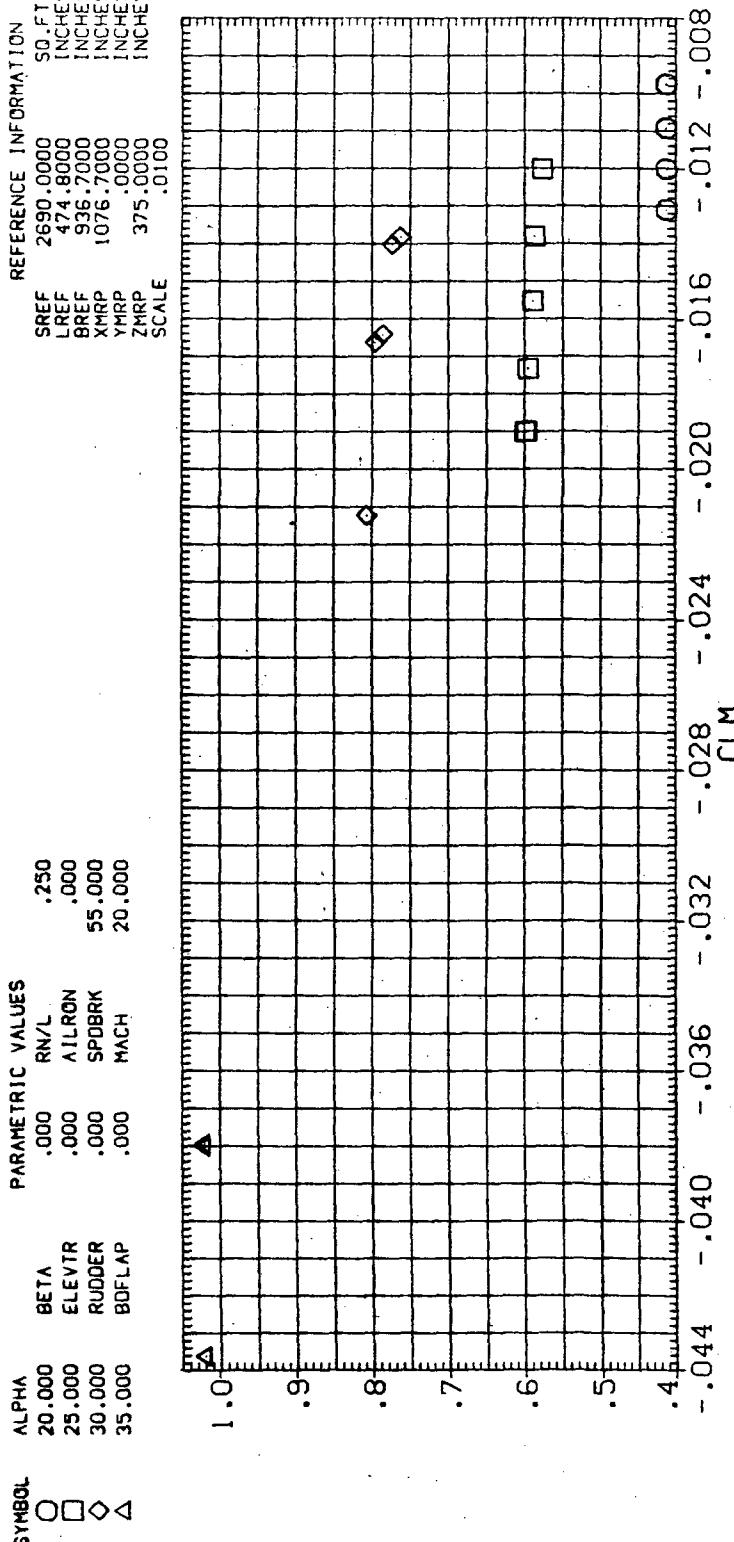
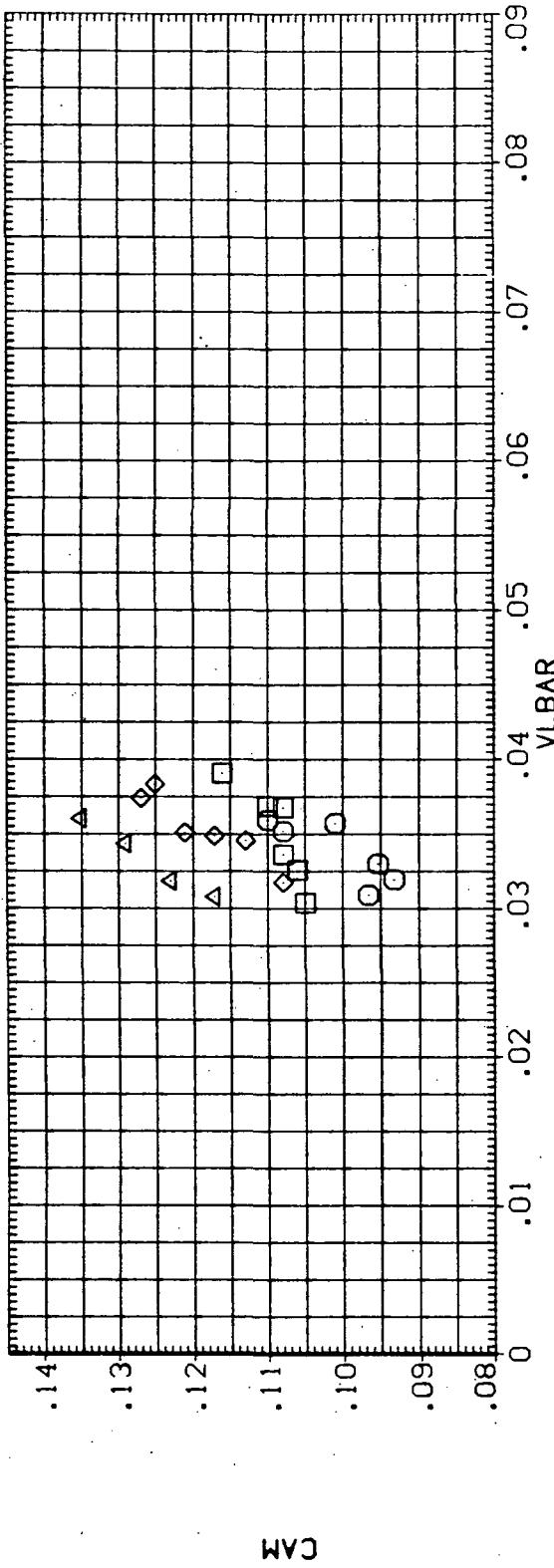


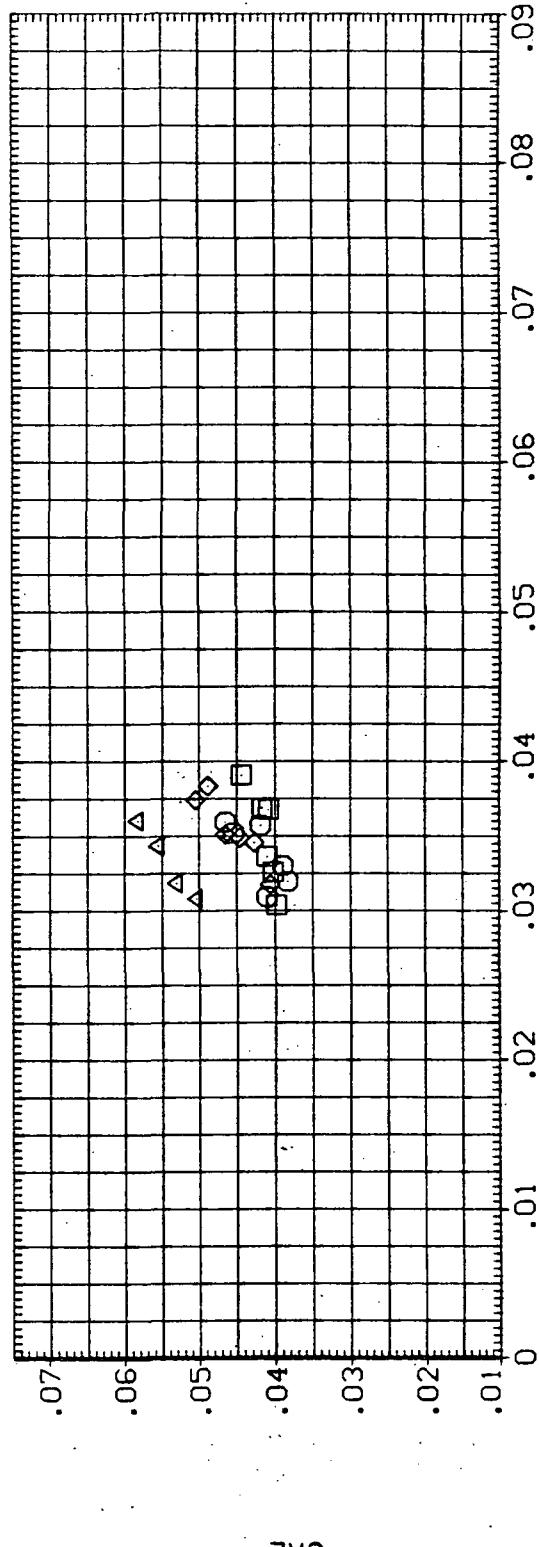
FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0005)



CAE



CAE

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

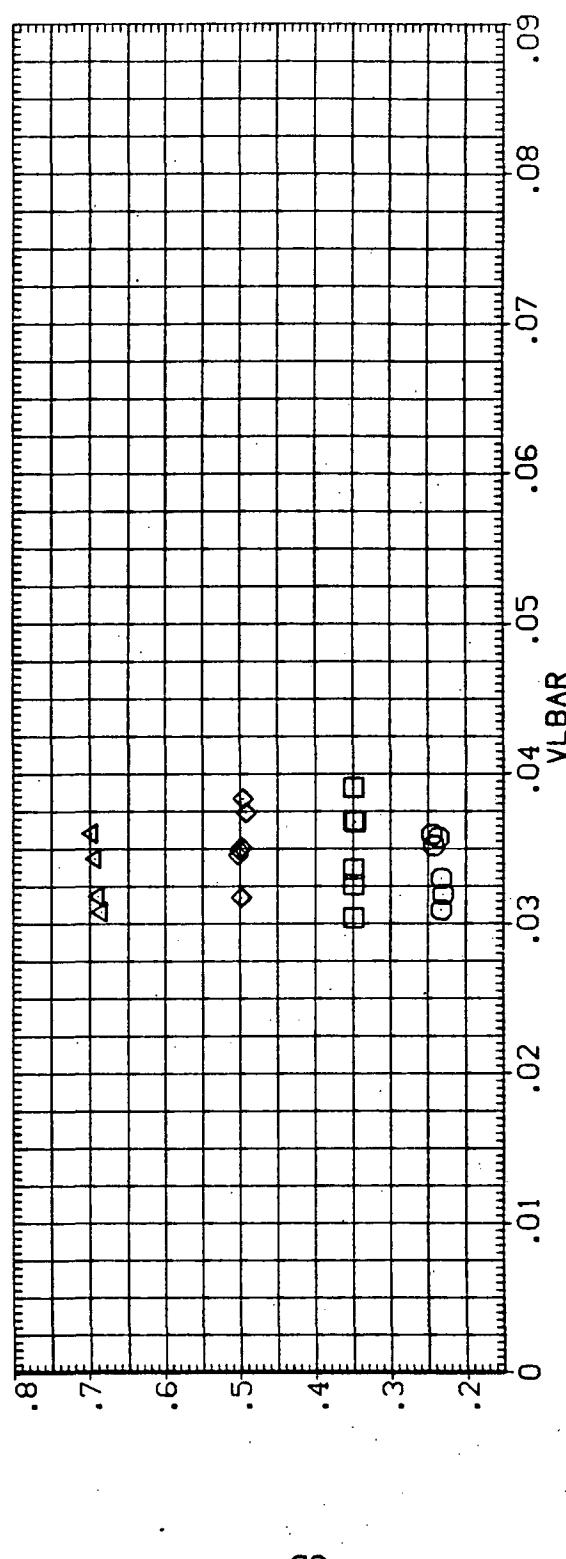
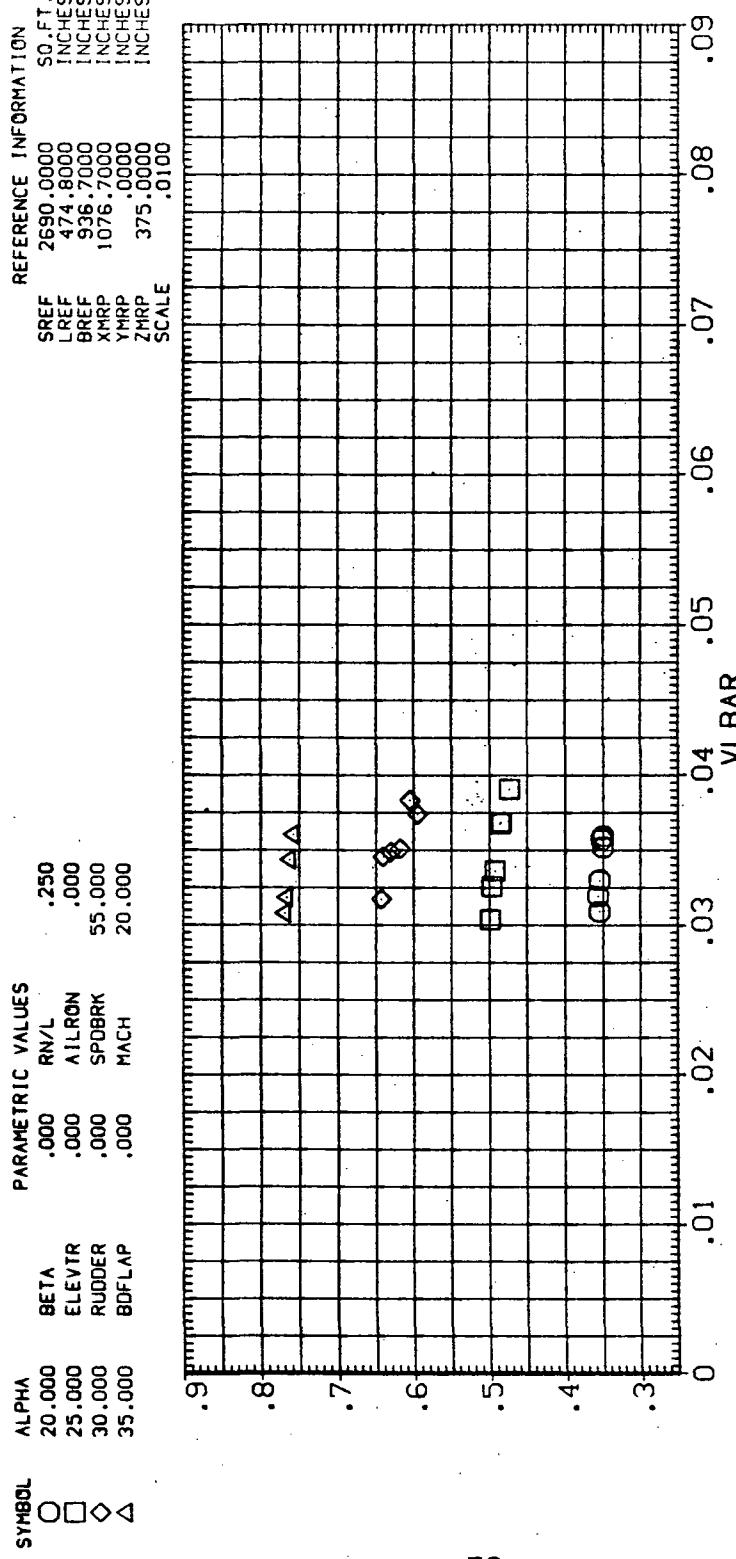


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0006)

PARAMETRIC VALUES	REFERENCE INFORMATION
ALPHA .000	SREF 2690.0000
BETA .000	LREF 474.8000
ELEVTR .000	BREF 936.7000
RUDDER .000	XMRP 1076.7000
BOFLAP .000	YMRP 375.0000
MACH .000	ZMRP .0100
	SCALE

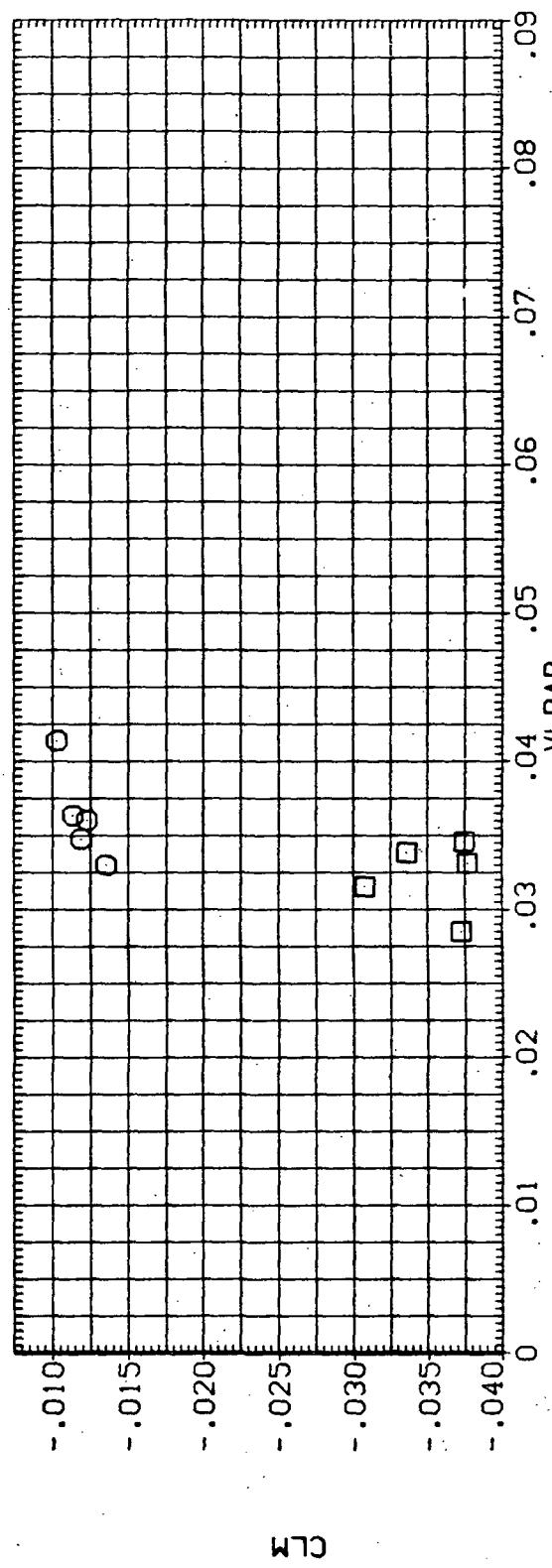
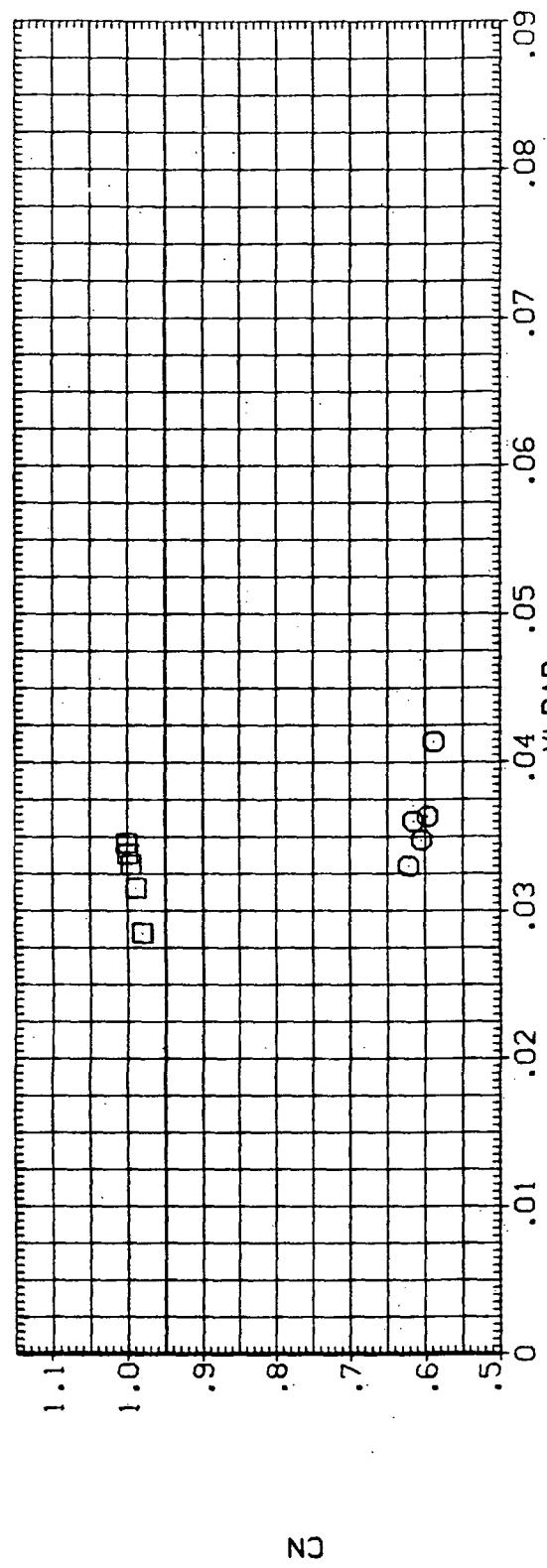
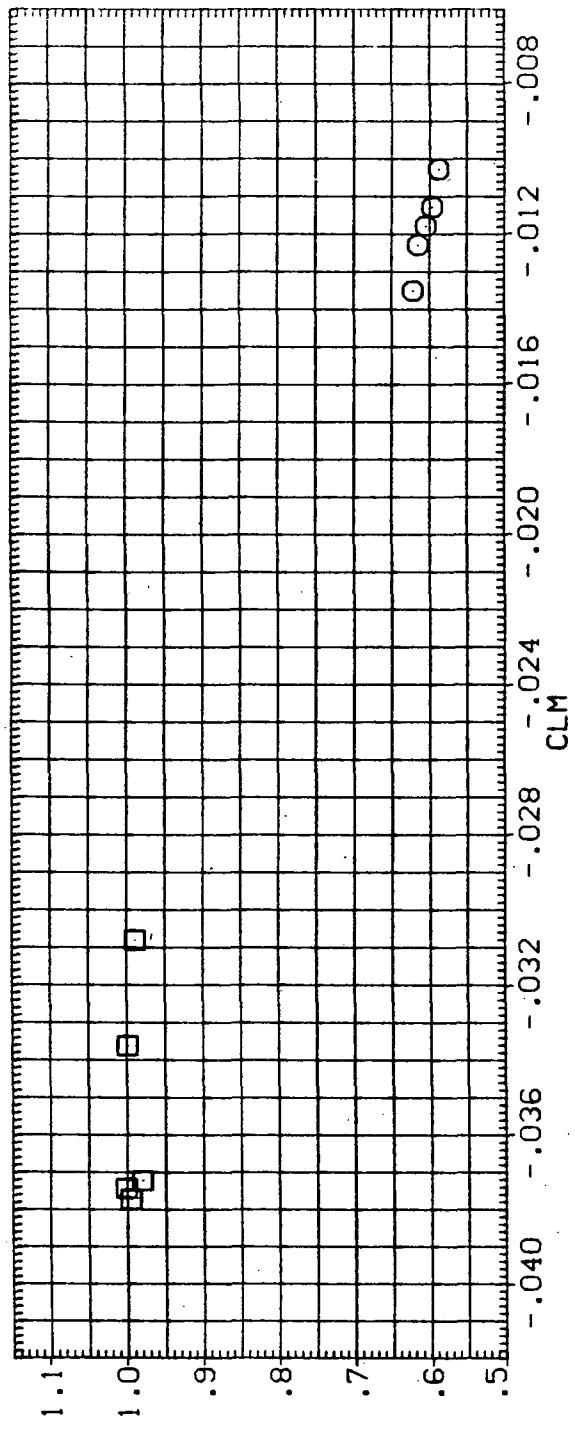
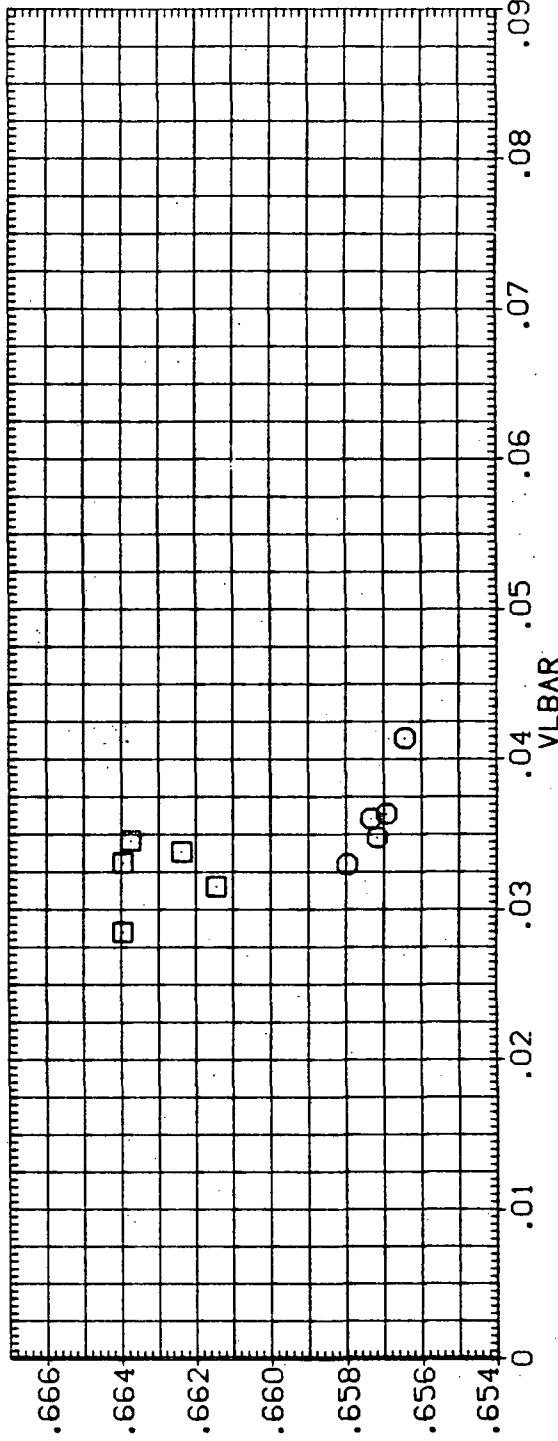


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

SYMBOL	PARAMETRIC VALUES			REFERENCE INFORMATION
	ALPHA	BETA	RNL	
O	25.000	.000	.250	SREF 2690.0000 LREF 474.8000 BREF 936.7000 XMRP .1076.7000 YMRP .375.0000 ZMRP .0100 SCALE .0100
□	35.000	.000	.000	



Cn



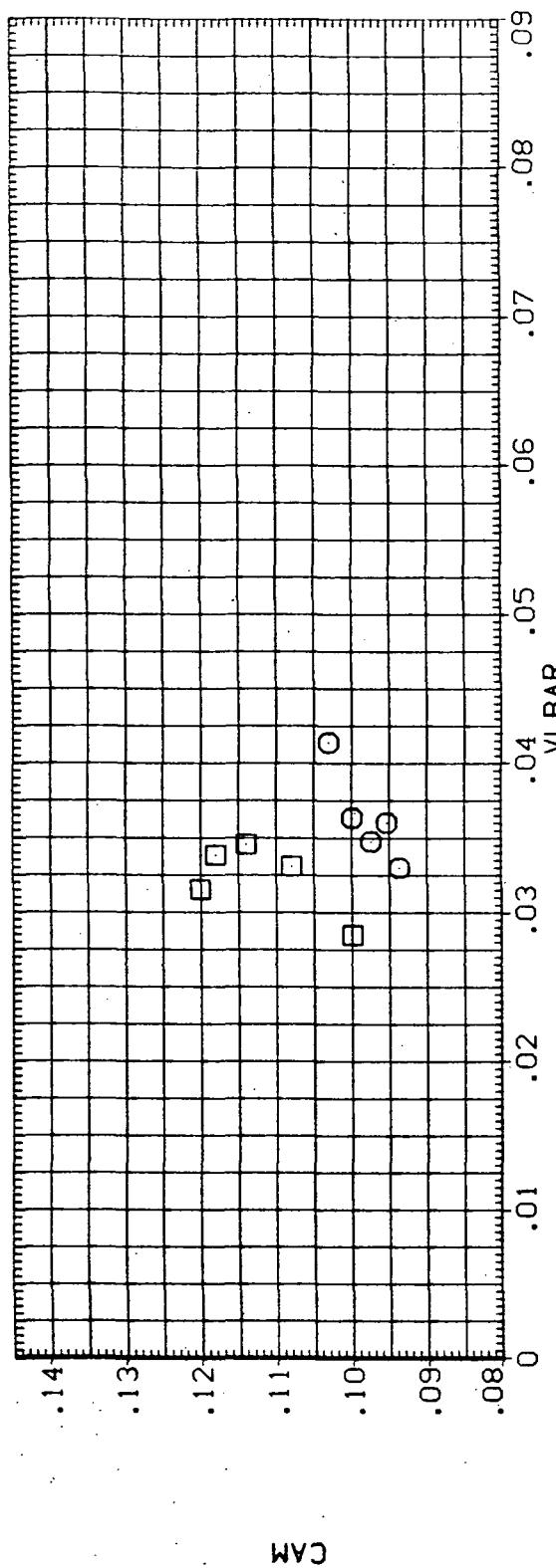
XCP/L

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

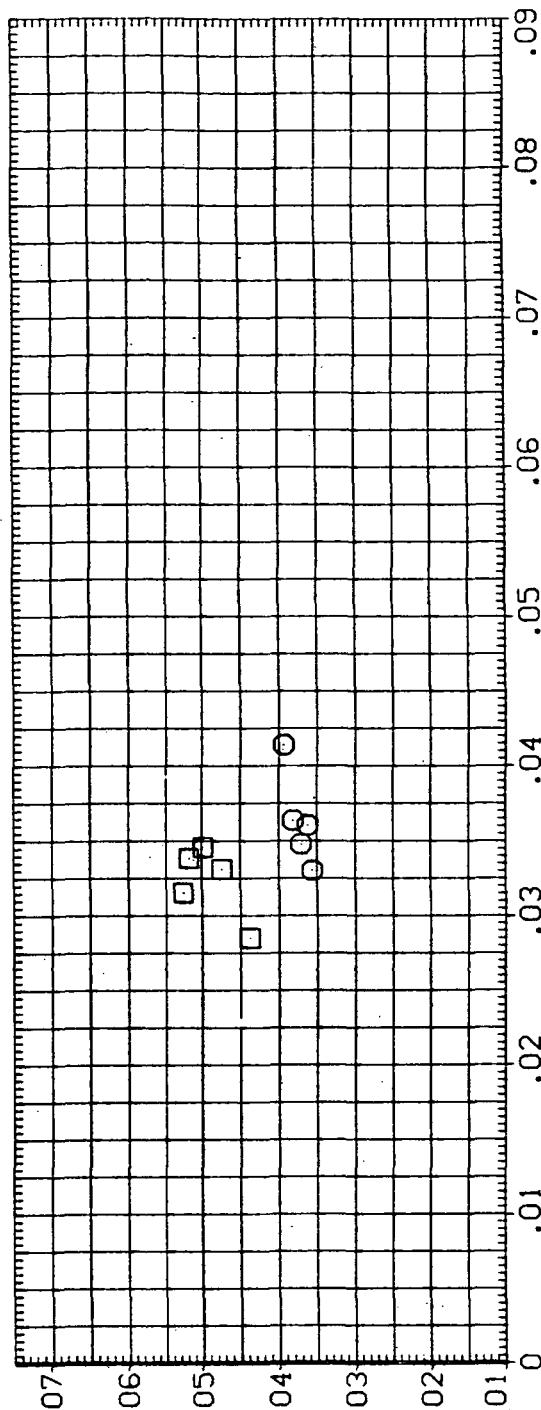
PAGE 52

AEDC VAA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5) (FT0006)

		REFERENCE INFORMATION			
SYMBOL	ALPHA	BETA	R/N/L	.250	SO.FT.
O	25,000	ELEVTR	.000	.000	2690,0000 474,8000 INCHES
□	35,000	RUDDER	.000	.000	936,7000 1076,7000 INCHES
		BDFLAP	.000	MACH	1075,0000 375,0000 INCHES SCALE .0100



CAM



CAE

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VA489(0A-81). (B26C9F7M7N28) (W116E26) (V8R5) (FT0006)

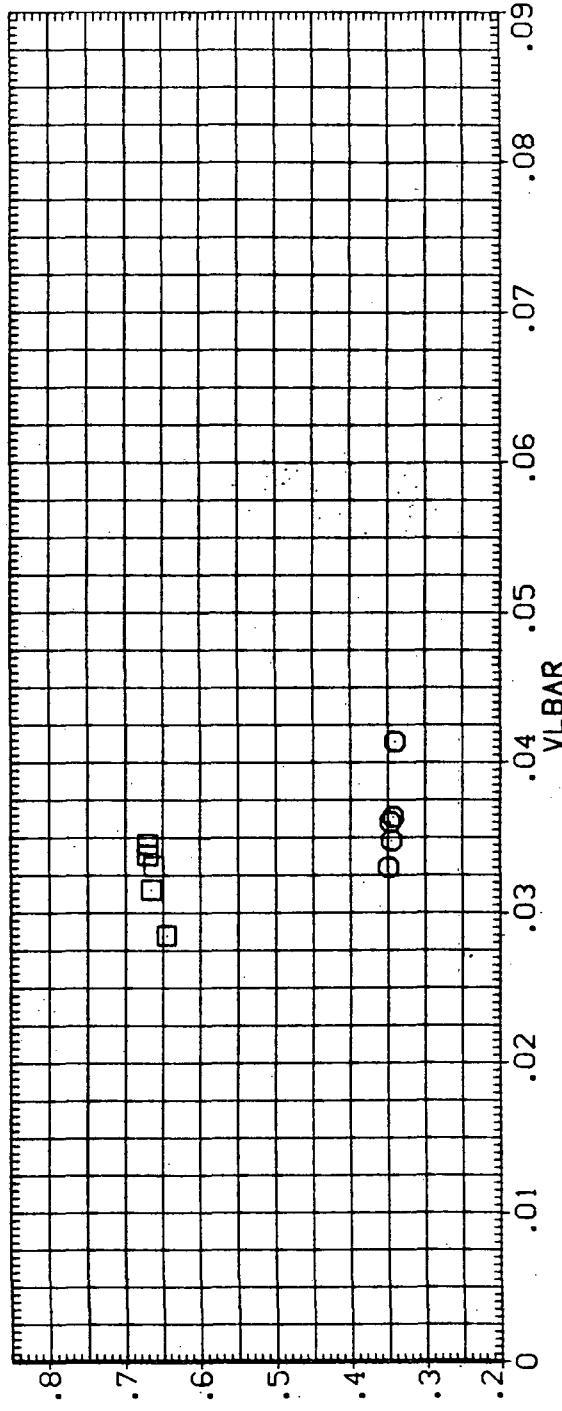
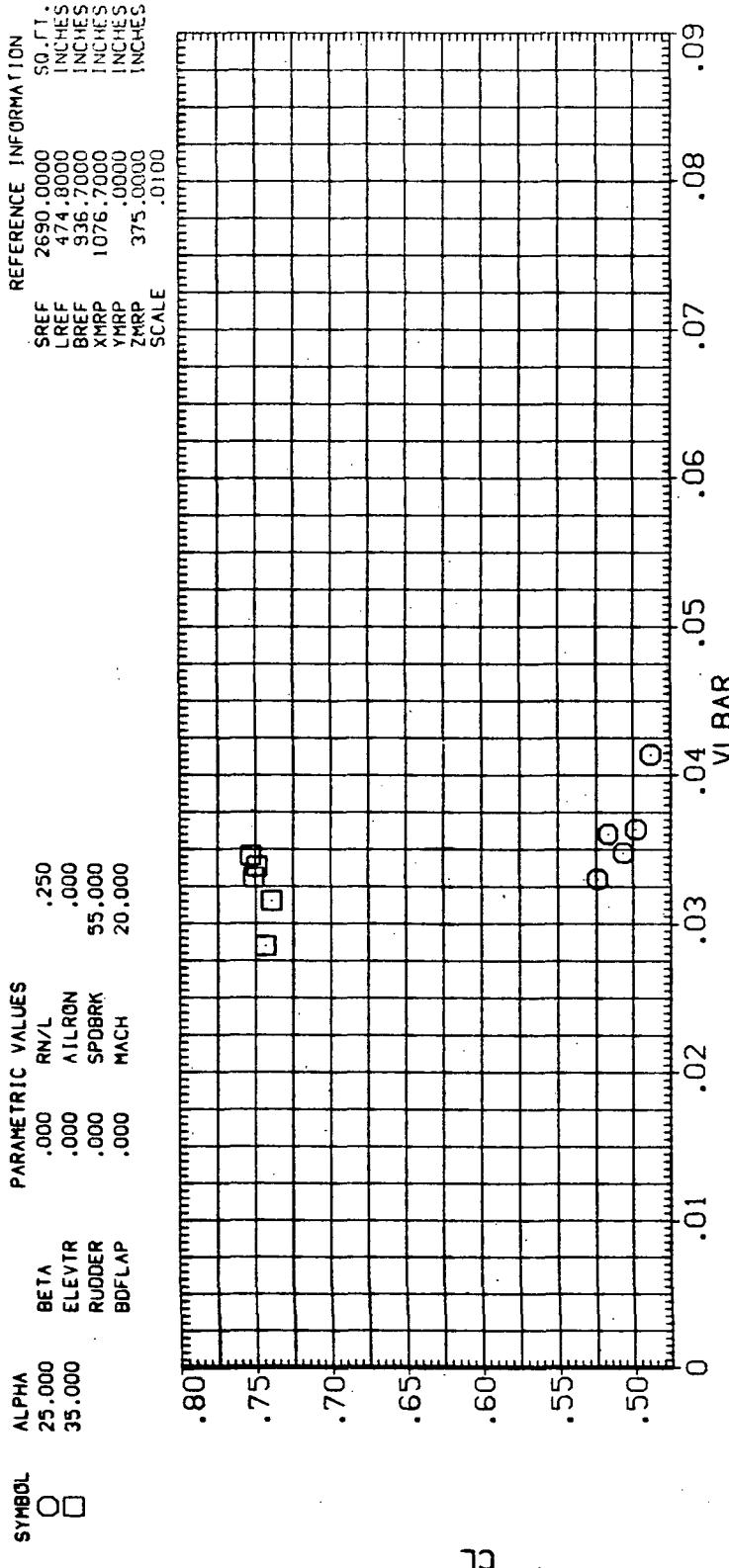


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VAA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5) (FT0007)



REFERENCE INFORMATION

	SQ.FT.	INCHES	INCHES	INCHES	INCHES	INCHES
SREF	2690.0000					
LREF	474.8000					
BREF	936.7000					
XMRP	1076.7000					
YMRP	.0000					
ZMRP	.375.0000					
SCALE	.0100					

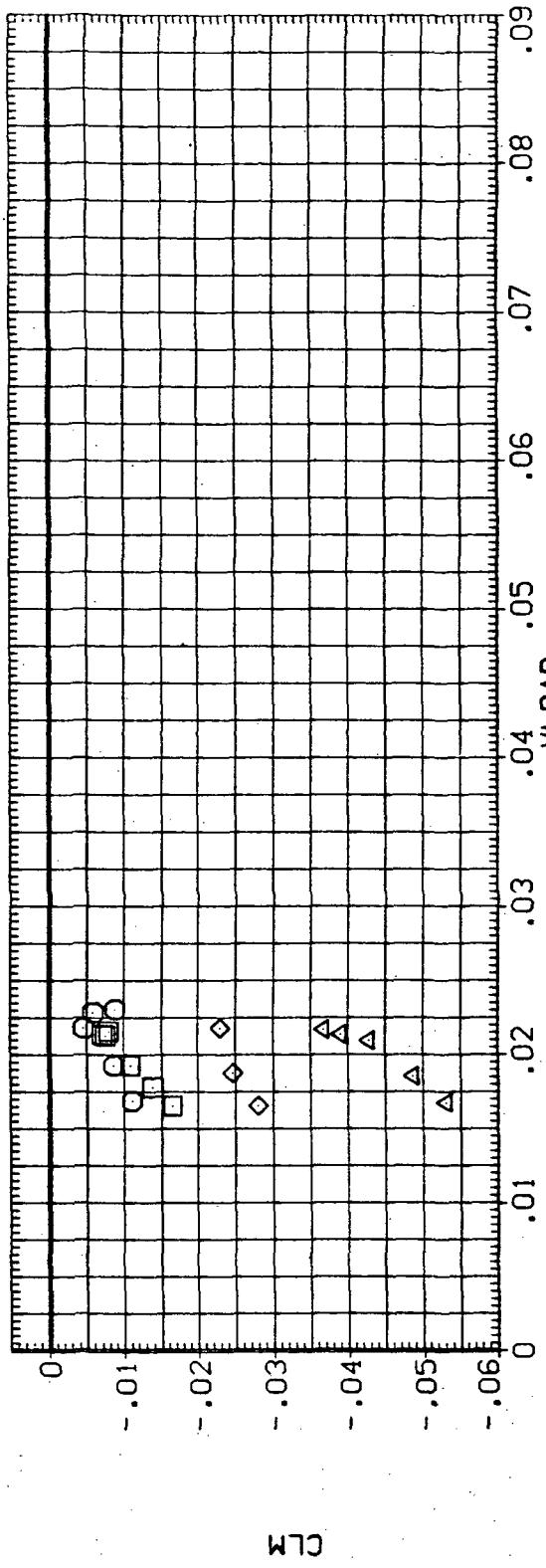
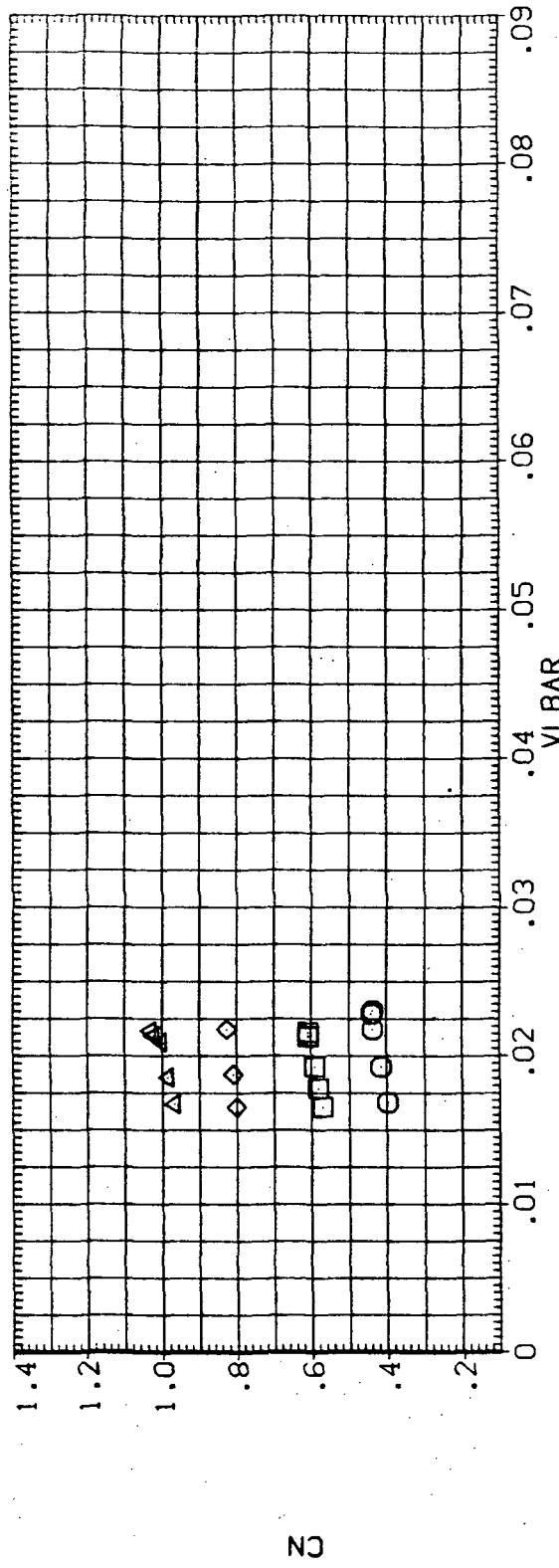


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

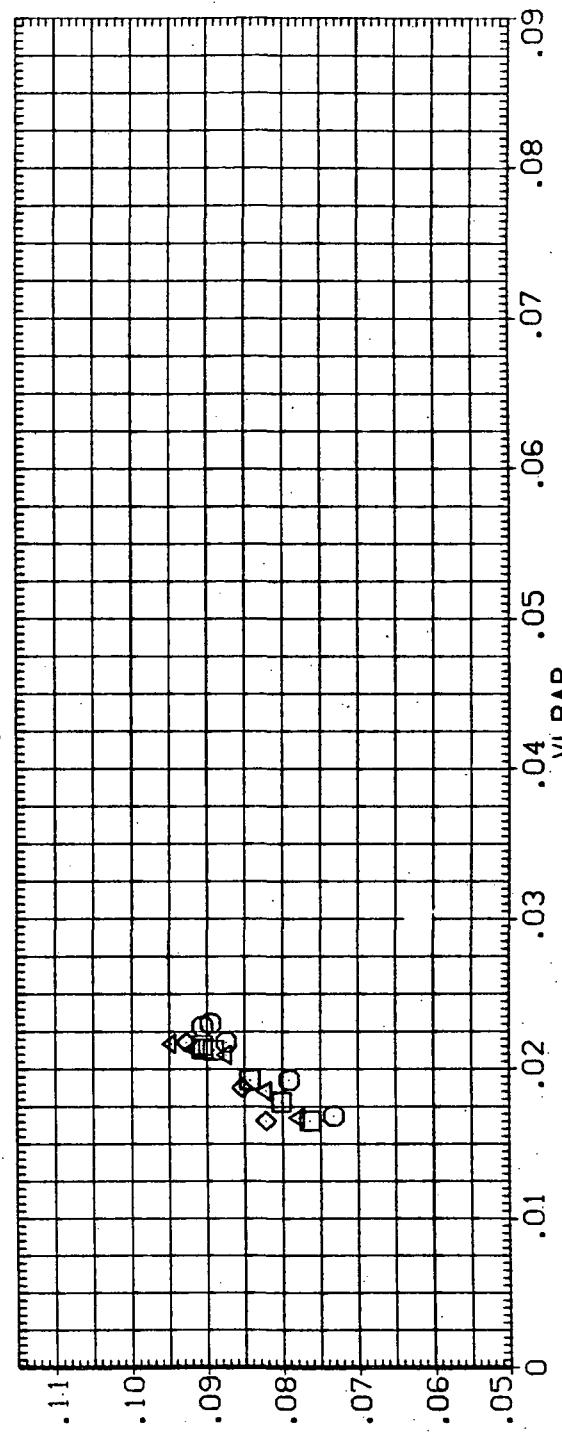
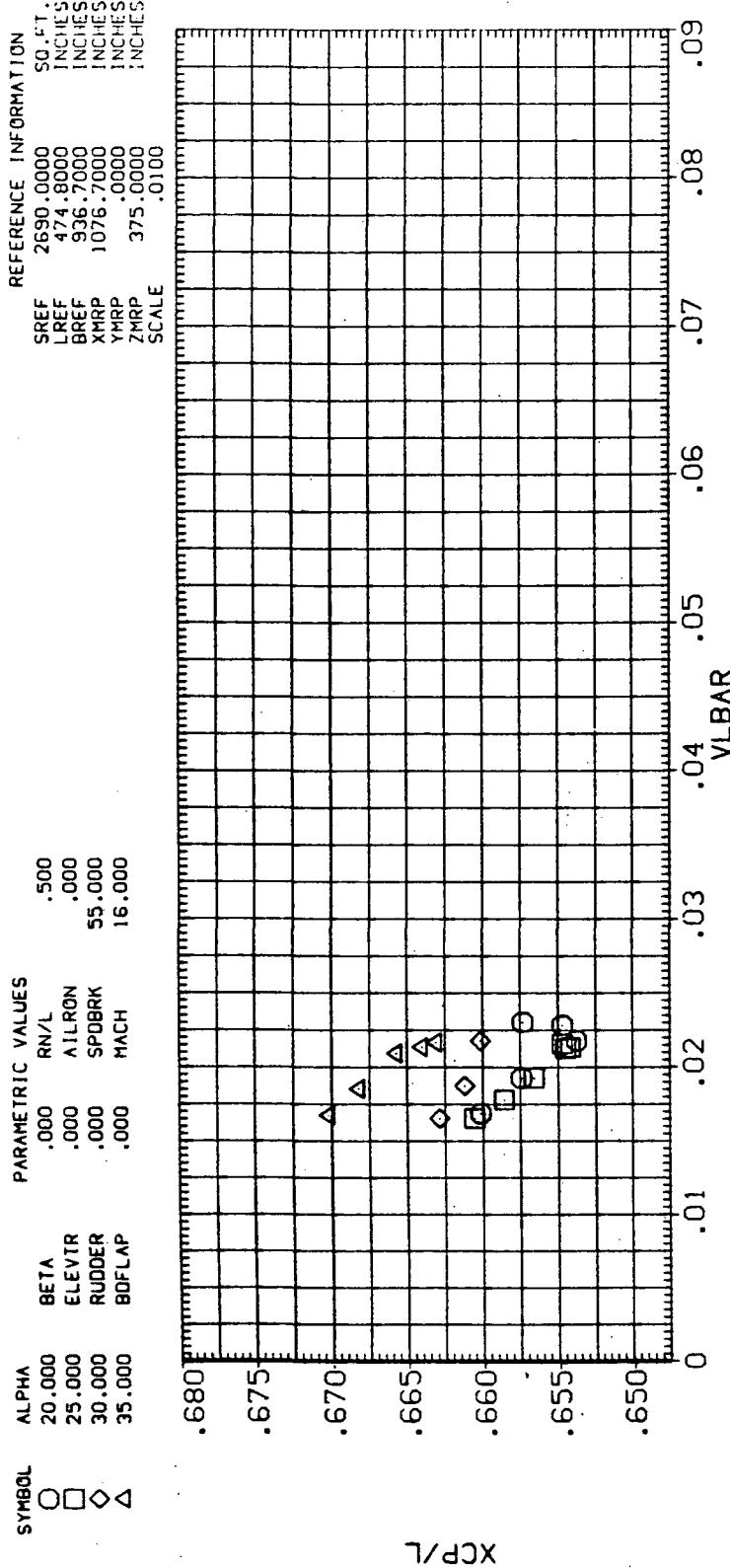
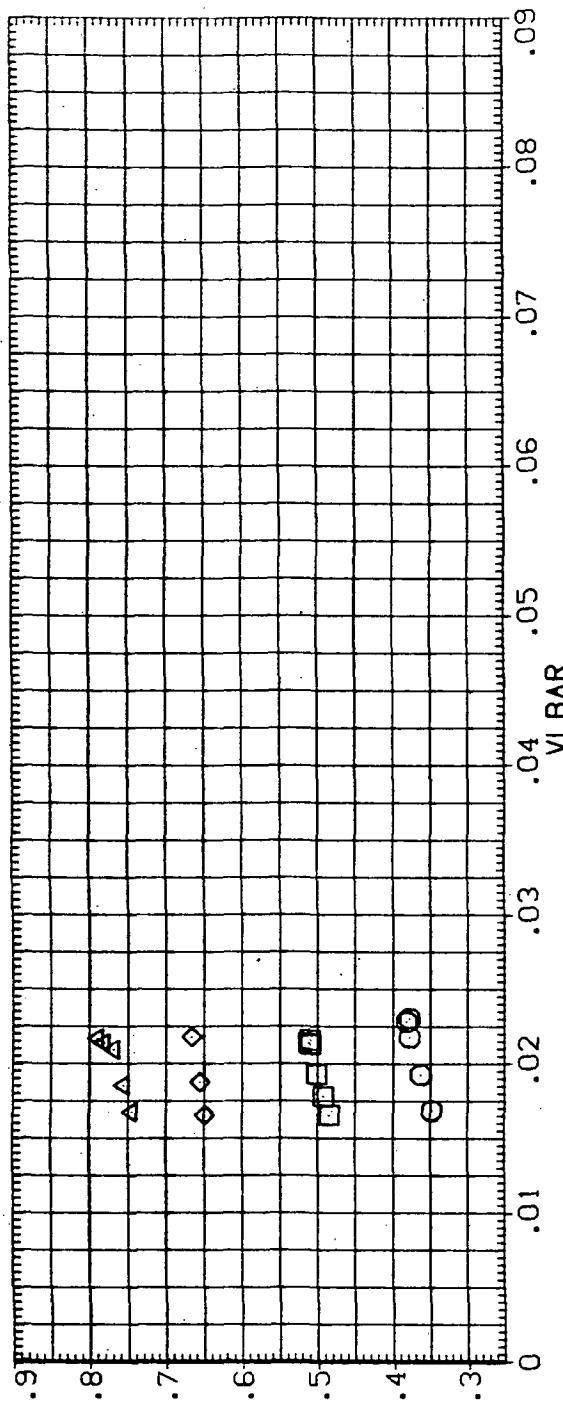


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

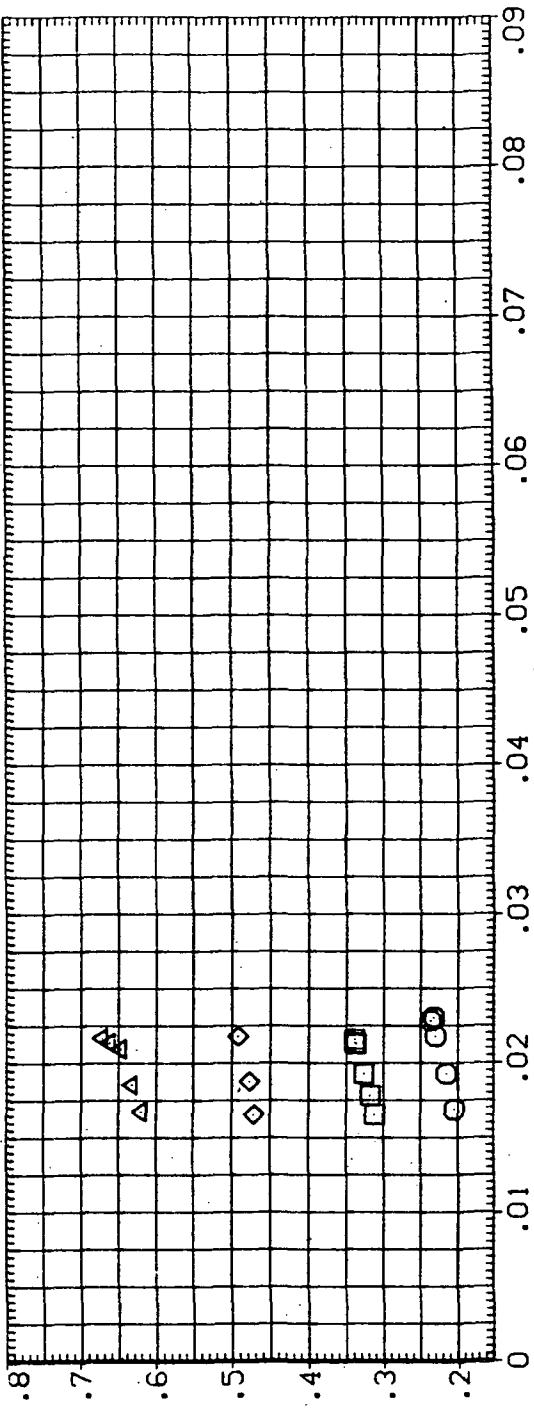
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AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5) (FT0007)

SYMBOL	ALPHA	BETA	PARAMETRIC VALUES	REFERENCE INFORMATION
○	20.000	.000	RNL .500	SREF 2690 .0000 LREF 474 .8000 BREF 936 .7000 XMRP 1076 .7000 YMRP 375 .0000 ZMRP .0100
□	25.000	.000	AIRON .000	INCHES
◊	30.000	.000	SPDBRK 55.000	INCHES
△	35.000	.000	MACH 16.000	INCHES



CL



CD

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5) (FT0008)

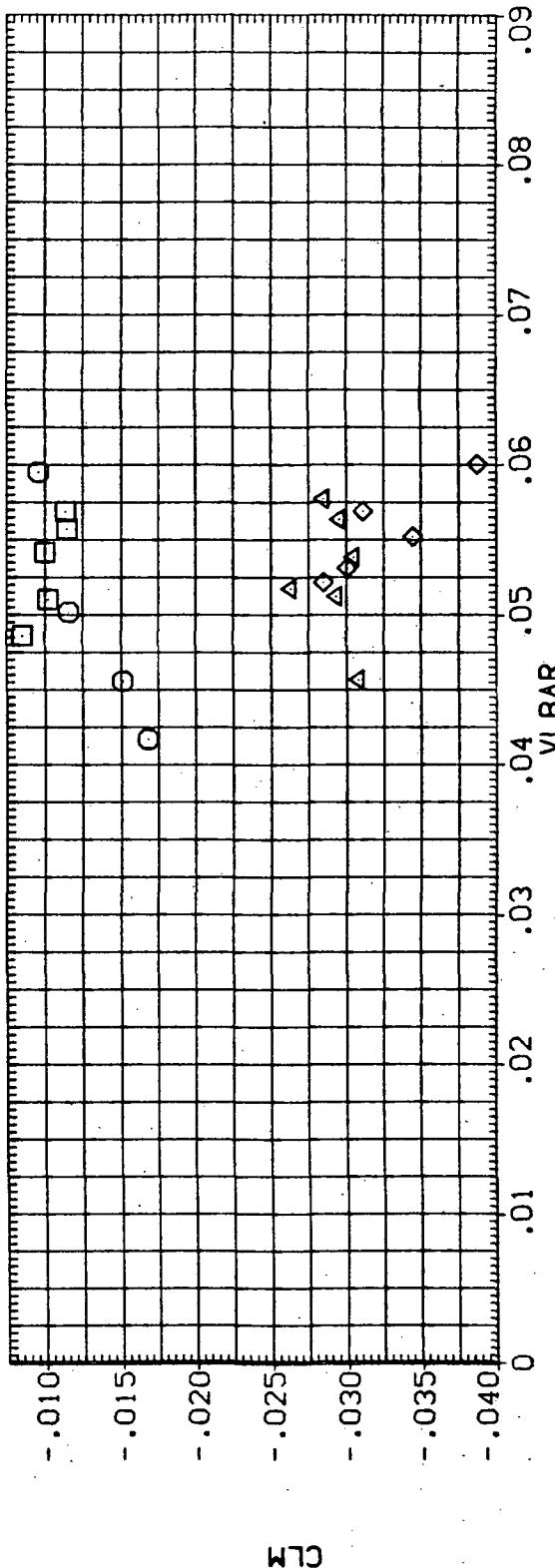
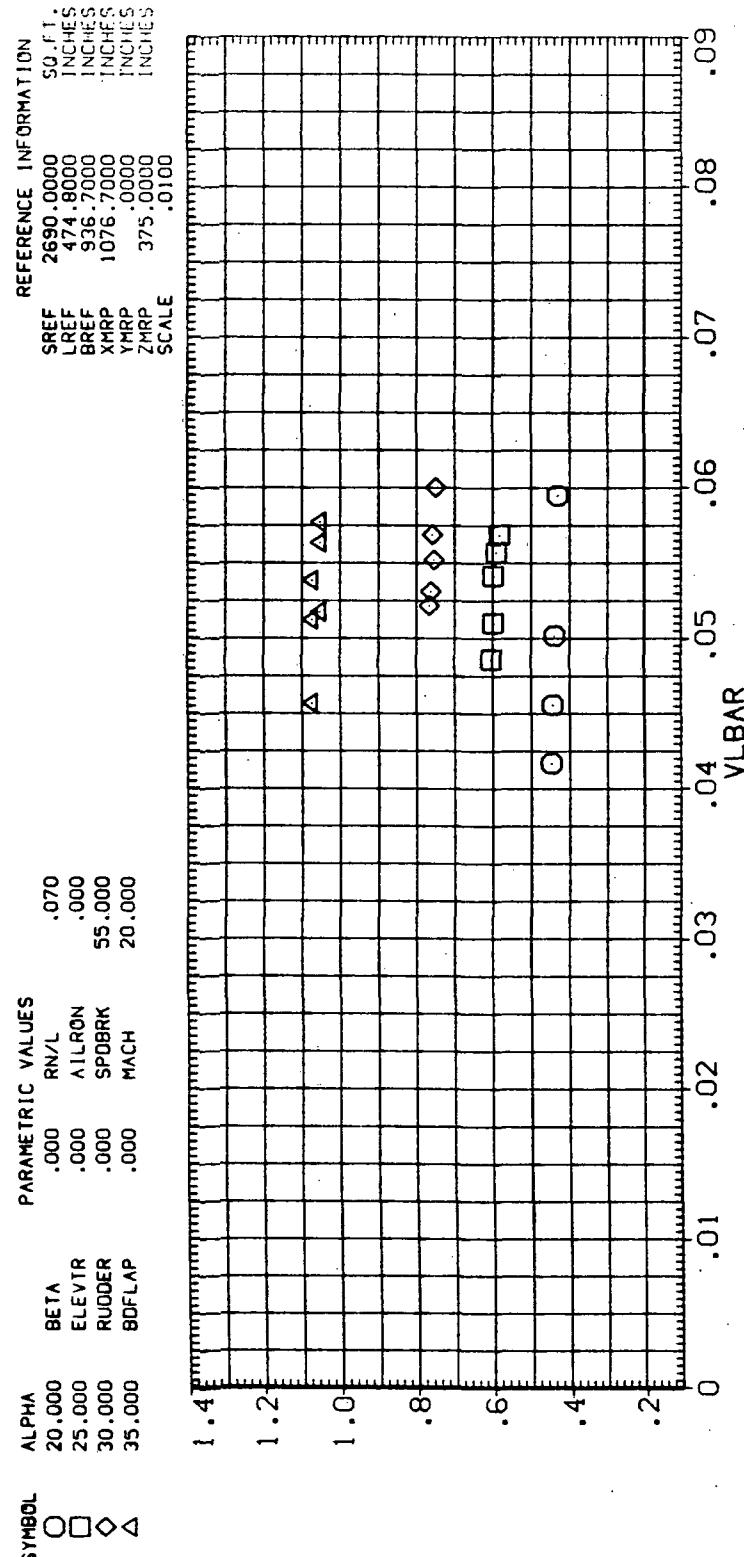
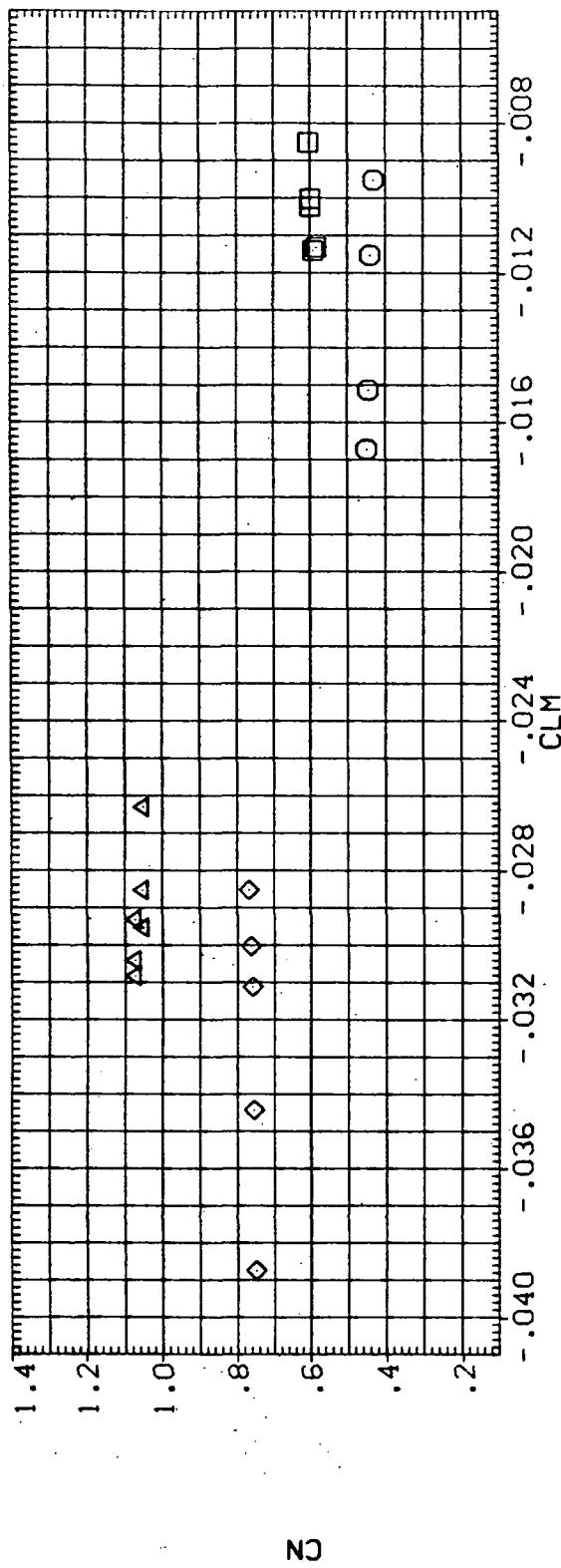


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

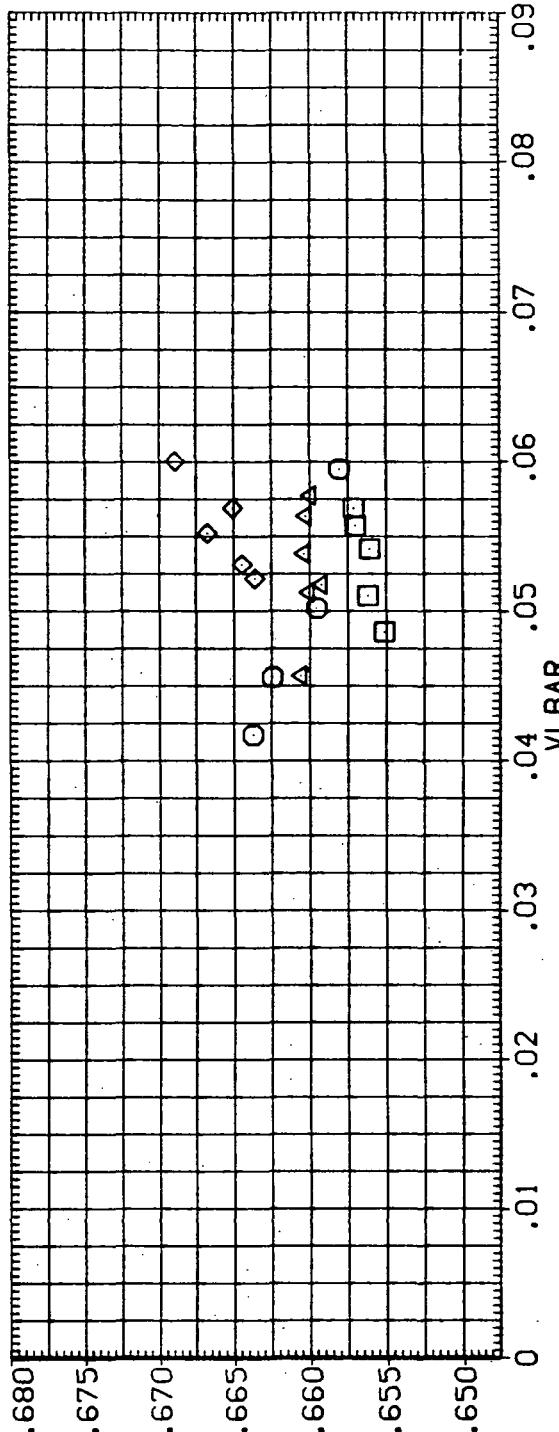
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AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT00008)

SYMBOL	ALPHA	PARAMETRIC VALUES			CLM	REFERENCE INFORMATION
		BETA	RNL	.070		
○	20.000	.000	AIRON	.000		SREF 2690.0000 SQ.FT.
□	25.000	.000	SPOBRK	.000		LREF 474.8000 INCHES
◊	30.000	.000	MACH	20.000		BREF 936.7000 INCHES
△	35.000	.000				XMRP 1076.7000 INCHES
						YMRP .375.0000 INCHES
						ZMRP .0100 INCHES
						SCALE



Cn



XCP/L

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

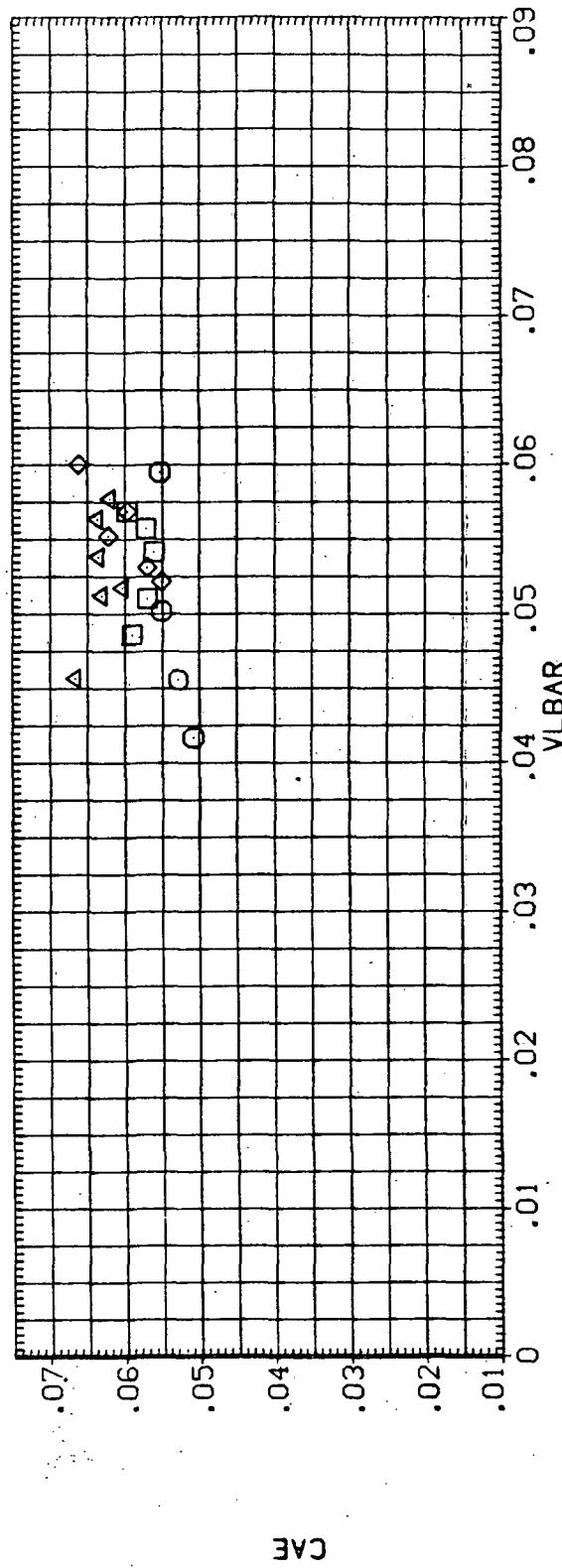
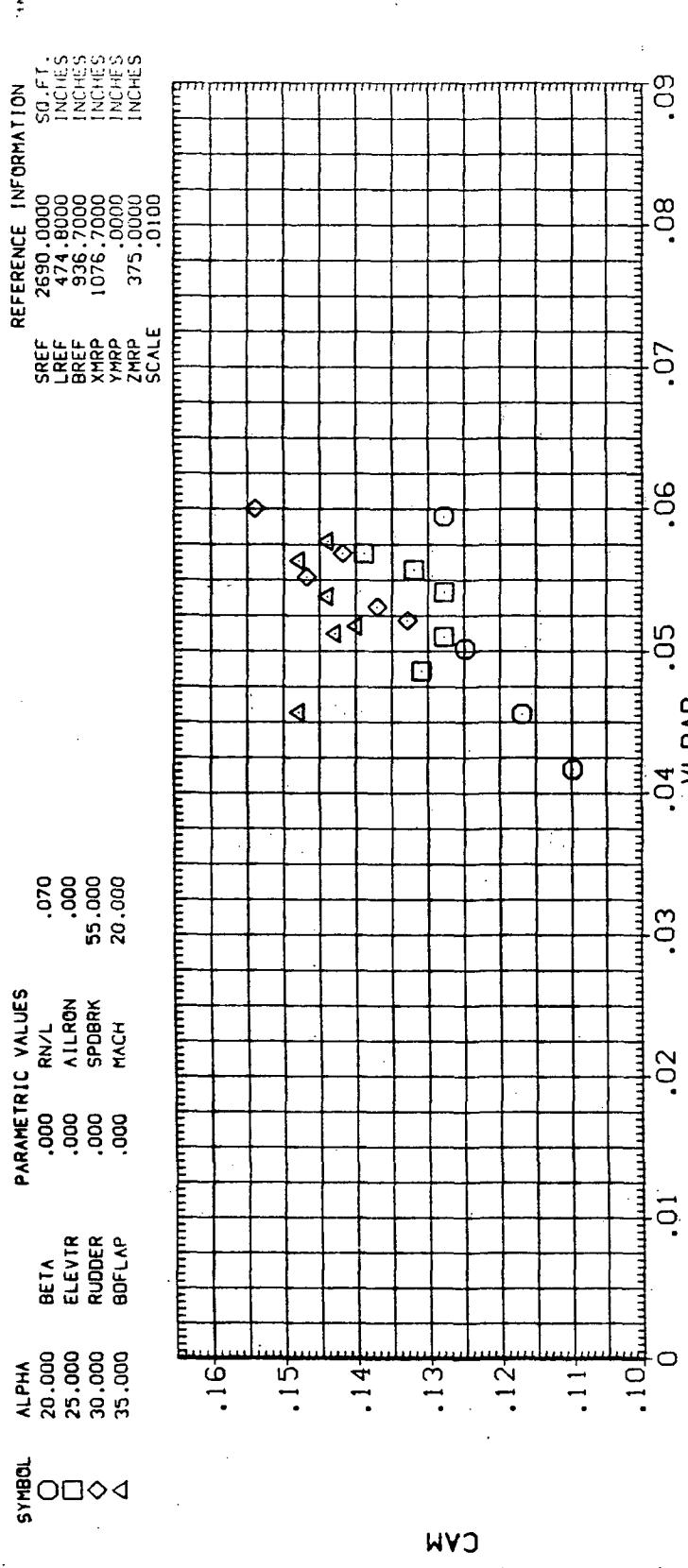
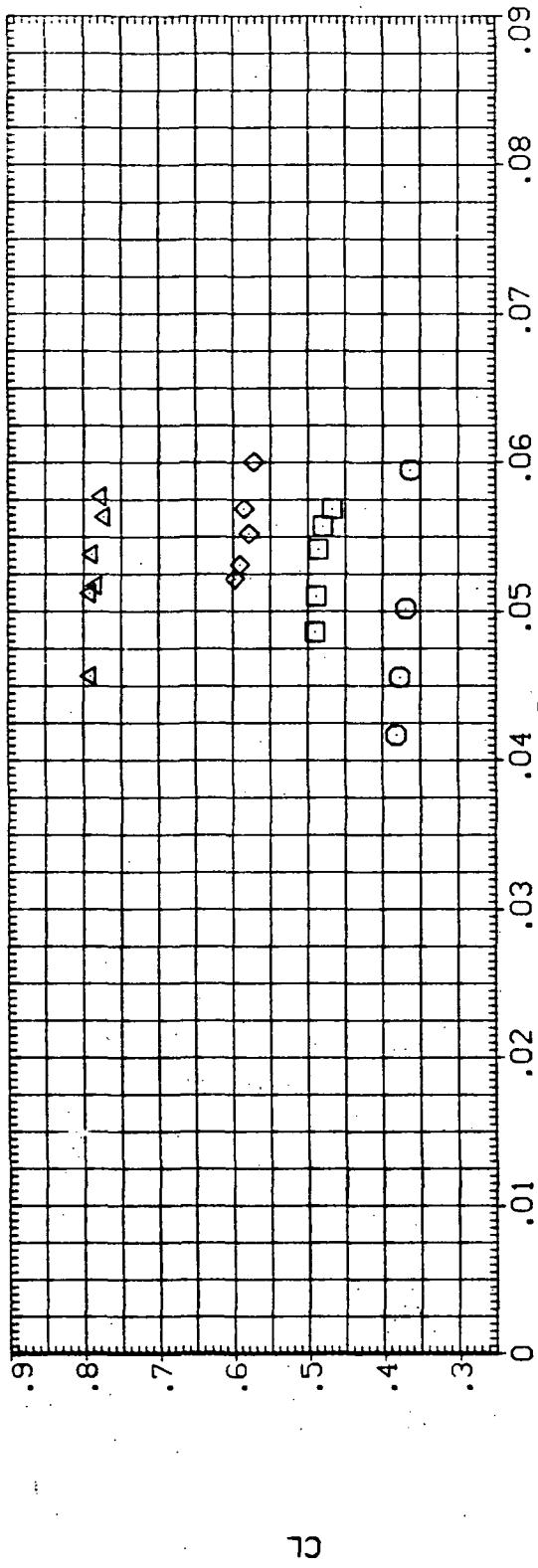


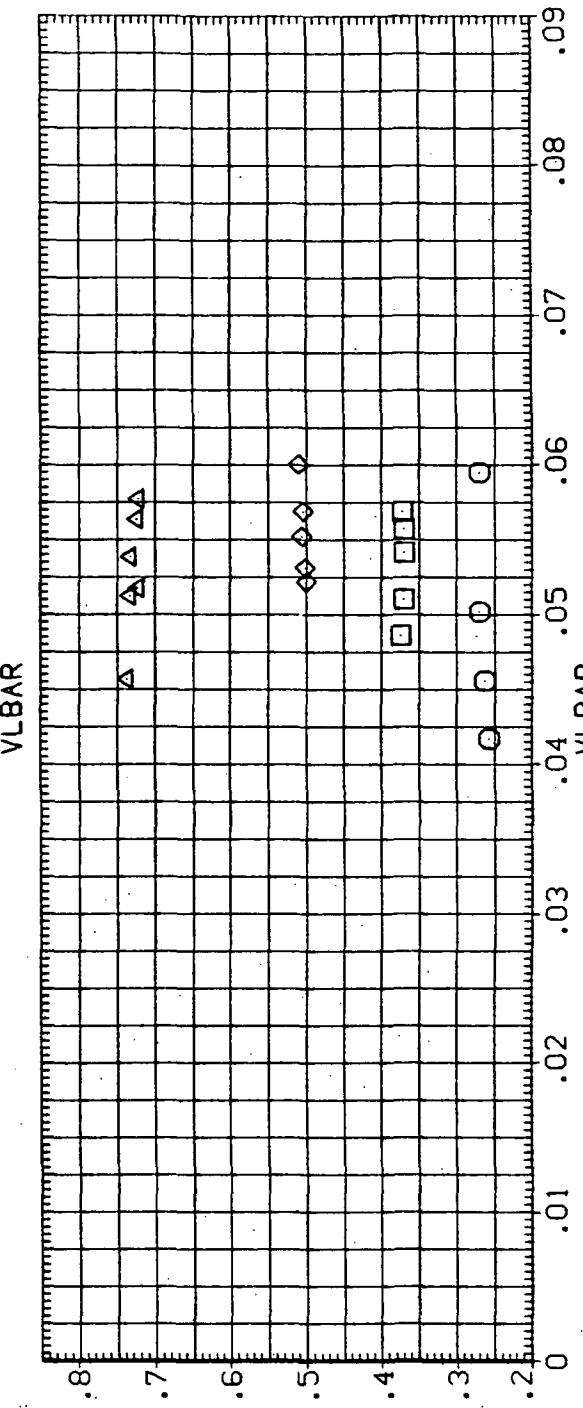
FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VAA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0008)

SYMBOL	ALPHA	PARAMETRIC VALUES		REFERENCE INFORMATION
		BETA	RN/L	
○	20,000	.000	.070	SREF 2690.0000 LREF 474.8000 BREF 936.7000 XMRP 1076.7000 YMRP 375.0000 ZMRP .0000 SCALE .0100
□	25,000	.000	.000	
◊	30,000	.000	.000	
△	35,000	.000	.000	
	BOFLAP	MACH	MACH	



CL



CL

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5) (FT0009)

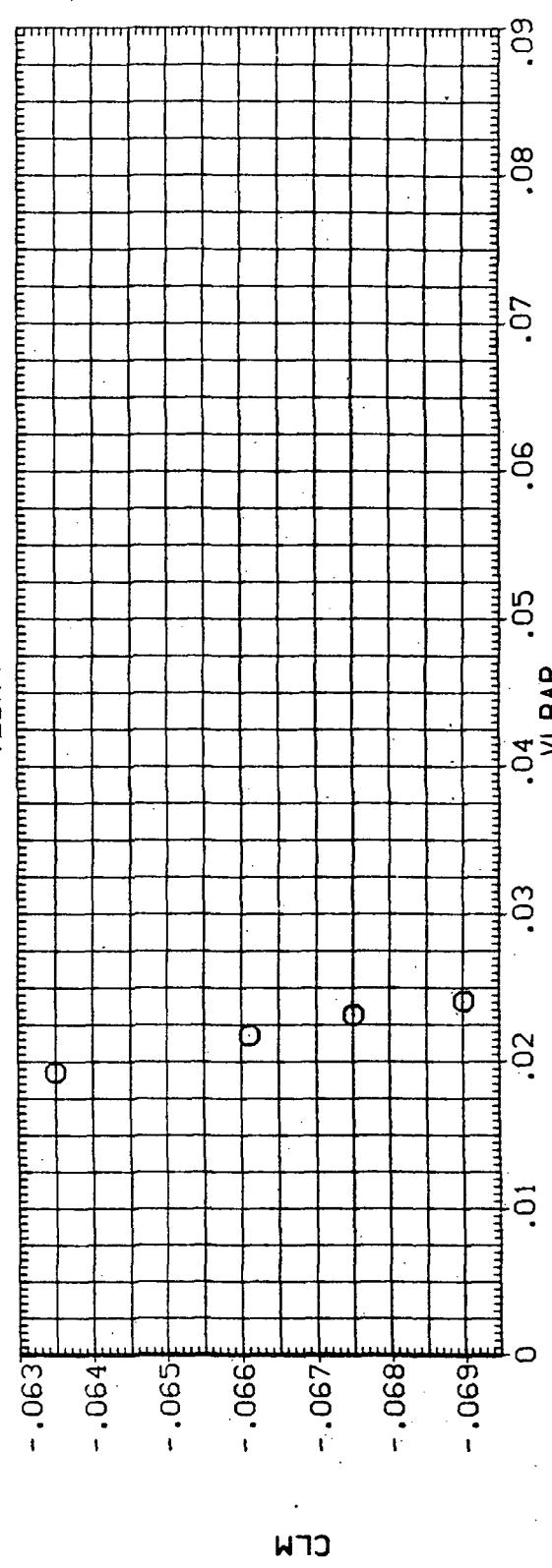
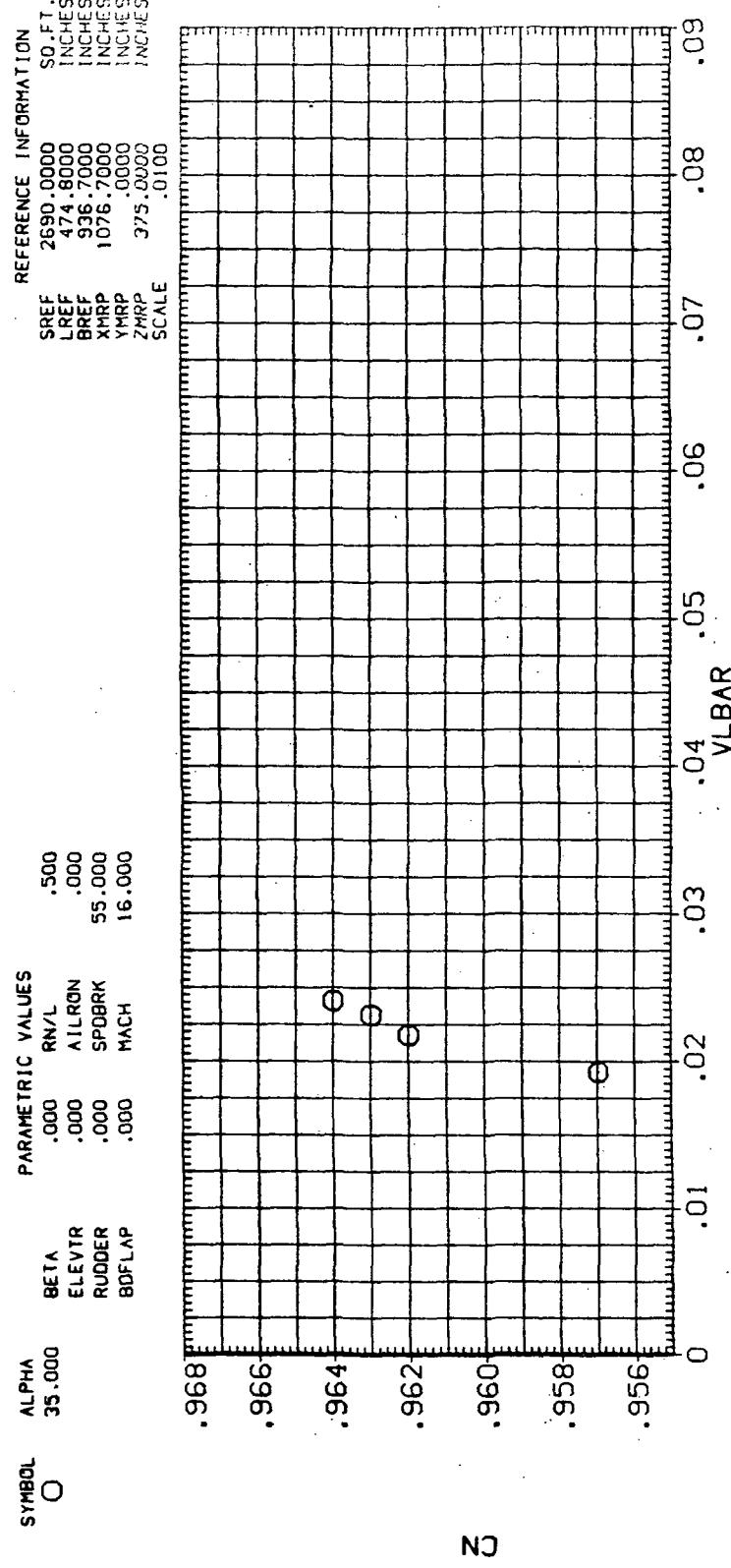


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VAA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0009)

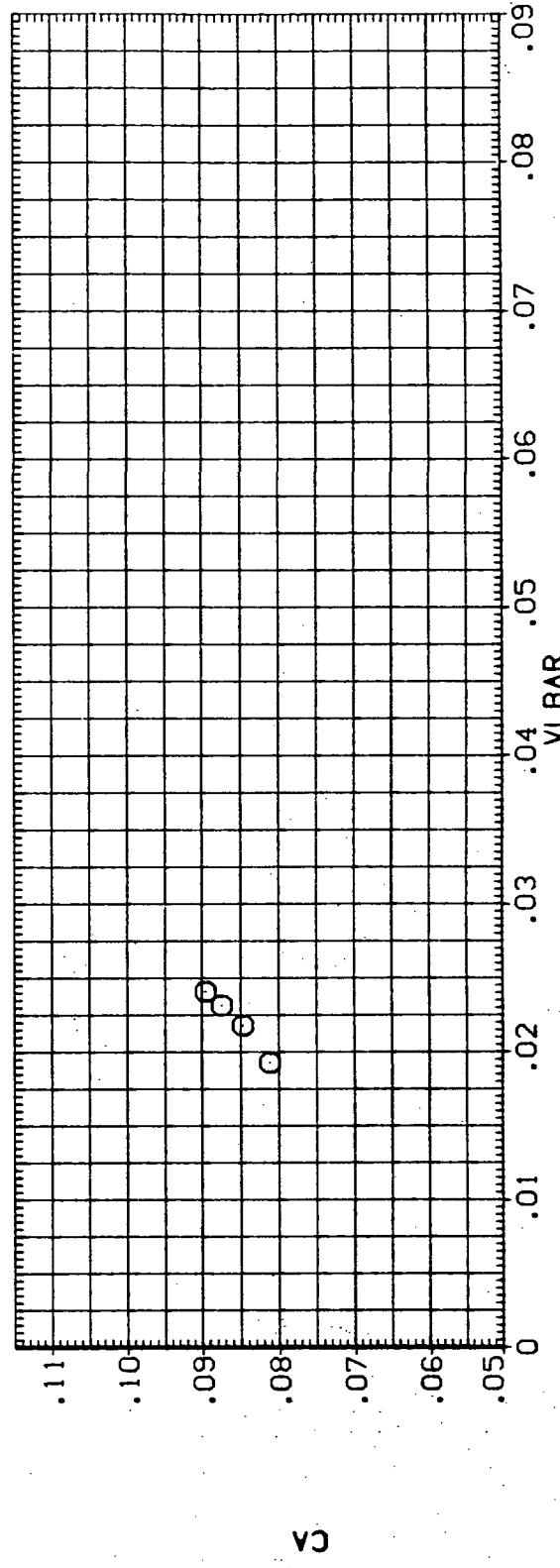
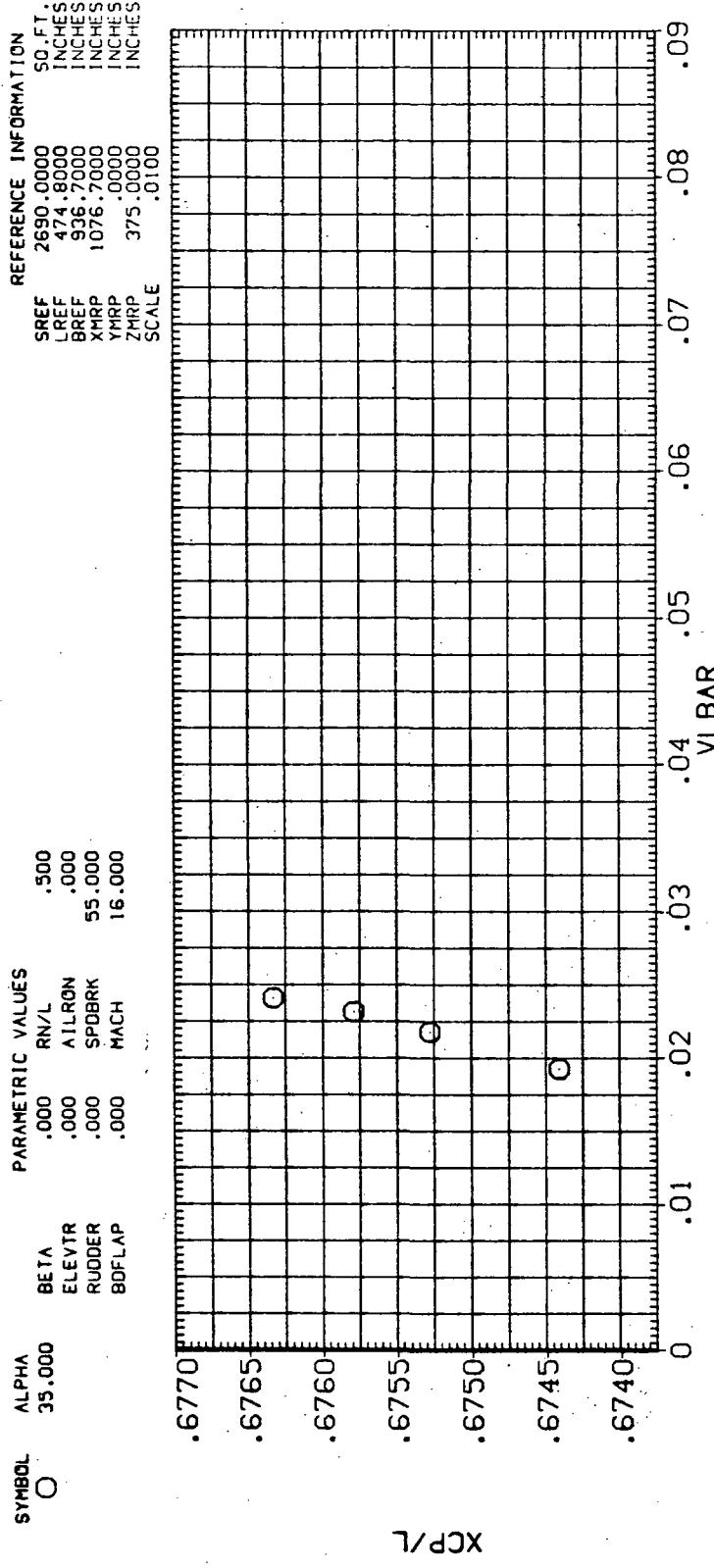


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

SYMBOL	ALPHA	PARAMETRIC VALUES		
		BETA	RNL	.500
O	35.000	.000	AILTRN	.000
	ELEVTR	.000	SPDBRK	55.000
	RUDDER	.000	MACH	16.000
	BOFLAP	.000		

REFERENCE INFORMATION
SREF 2690.00000 SO.F1:
LREF 474.80000 INCHES
BREF 936.70000 INCHES
XMRP 1076.70000 INCHES
YMRP .00000 INCHES
ZMRP 375.00000 INCHES
SCALE .0000

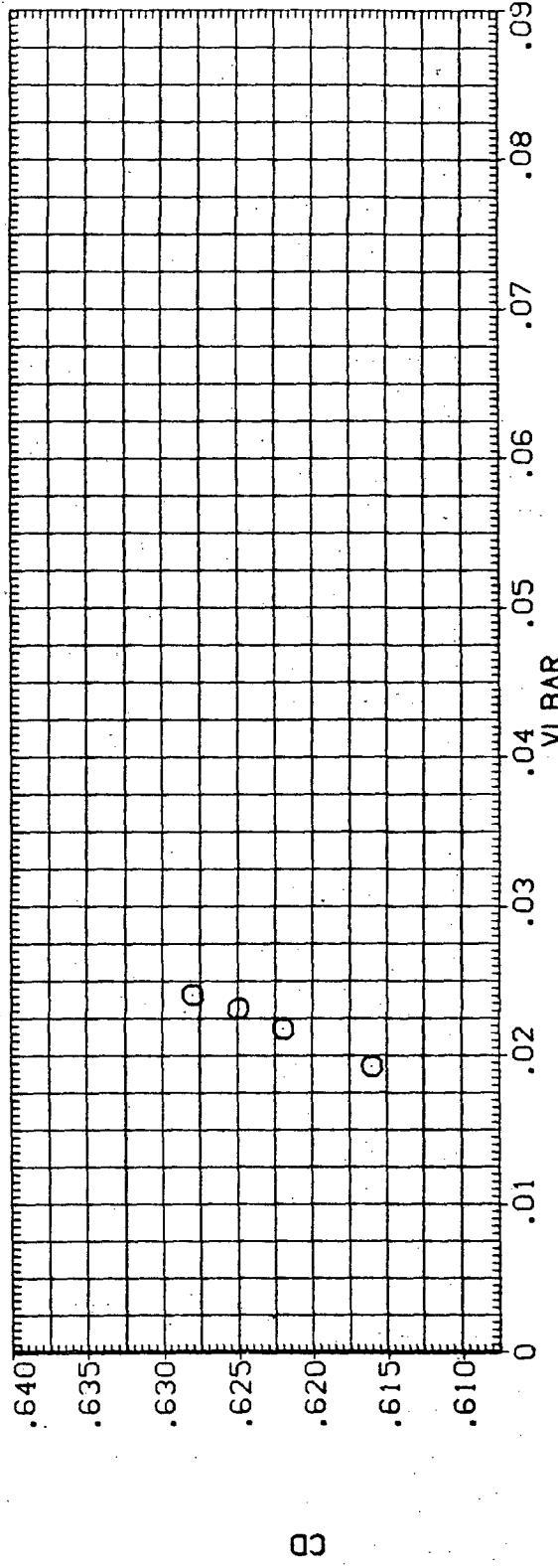
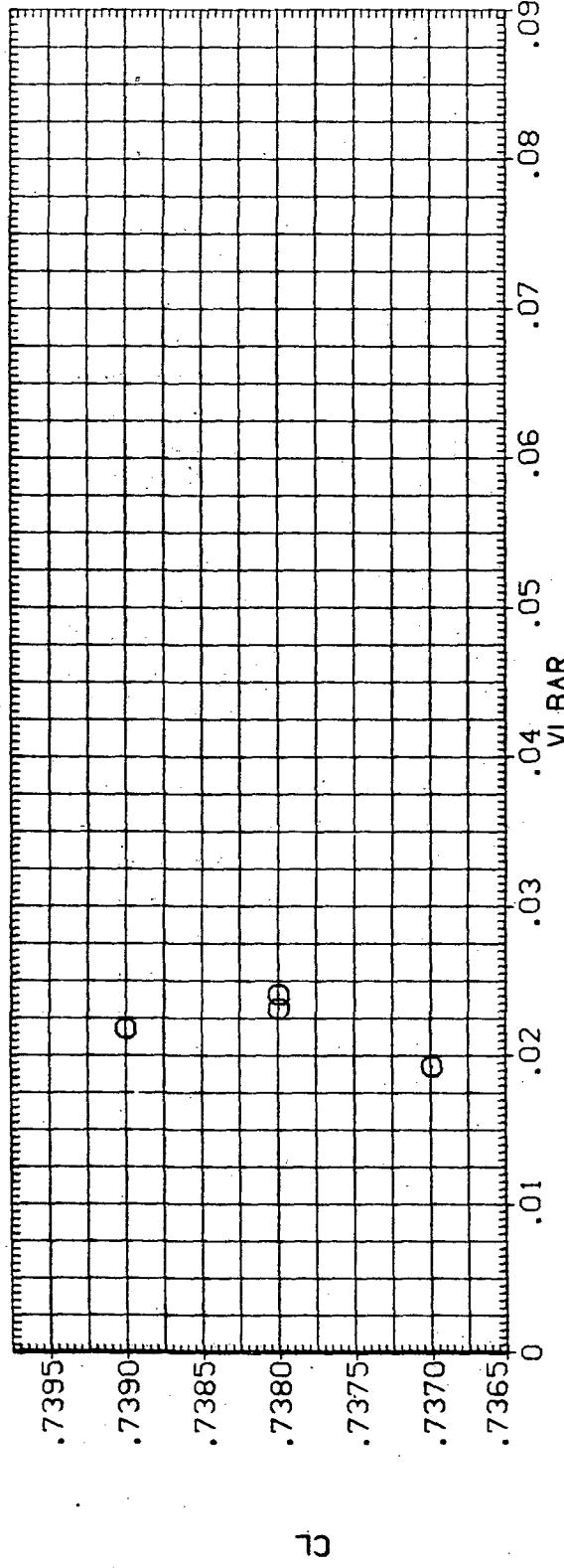
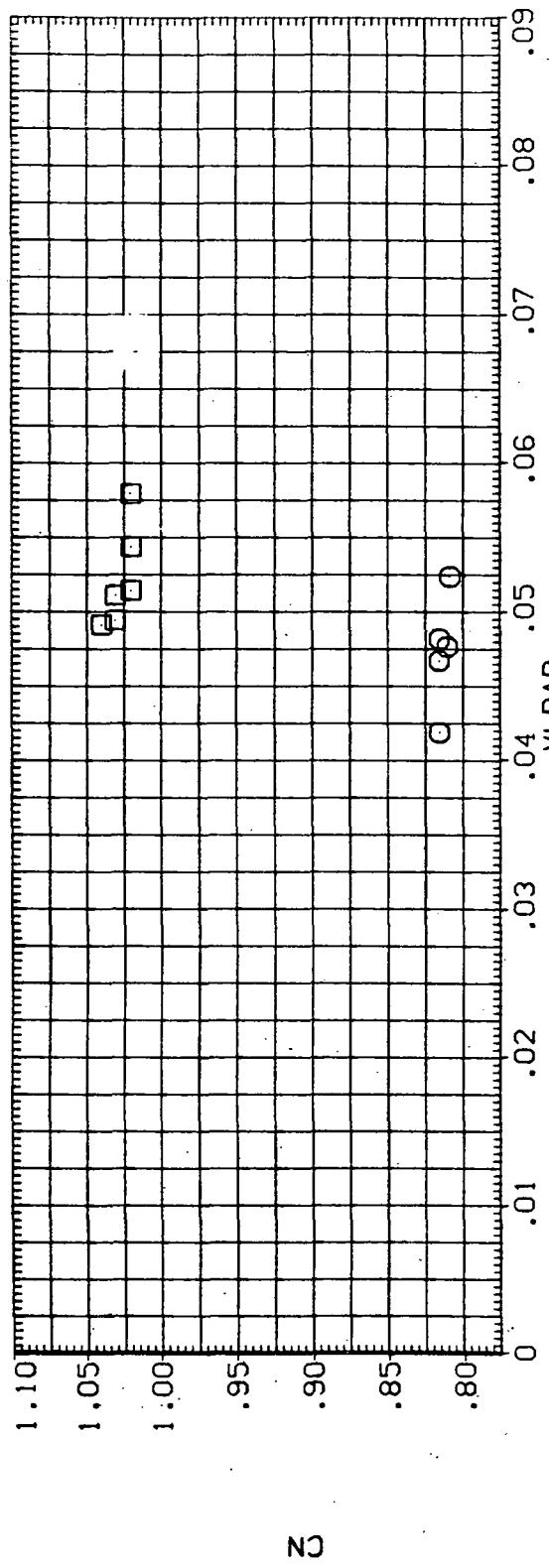


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

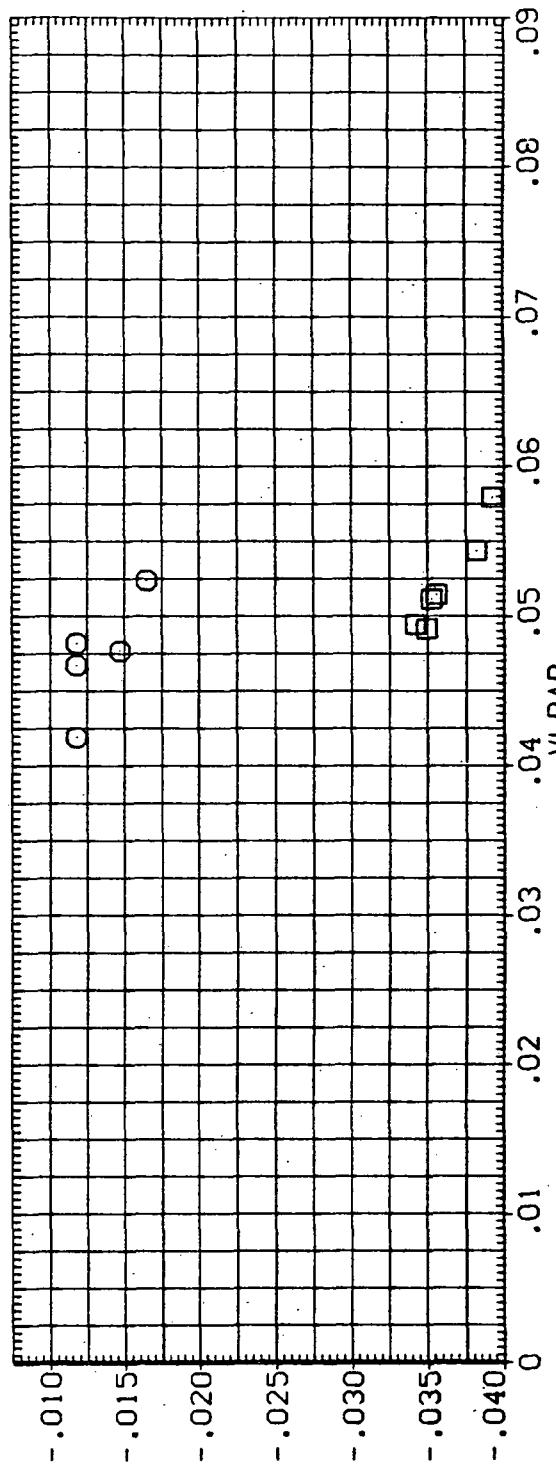
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AEDC VAA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0010)

SYMBOL	ALPHA	PARAMETRIC VALUES			REFERENCE INFORMATION
		BETA	RN/L	.070	
○	30.000	.000	AILRON	.000	SREF 2690.0000 LREF 474.8000 BREF 936.7000 XMRP 1076.7000 YMRP 375.0000 ZMRP .0000 SCALE .0100
□	35.000	.000	SPDBRK	55.000	
			MACH	20.000	
			BDFLAP		



C_n



CLM

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

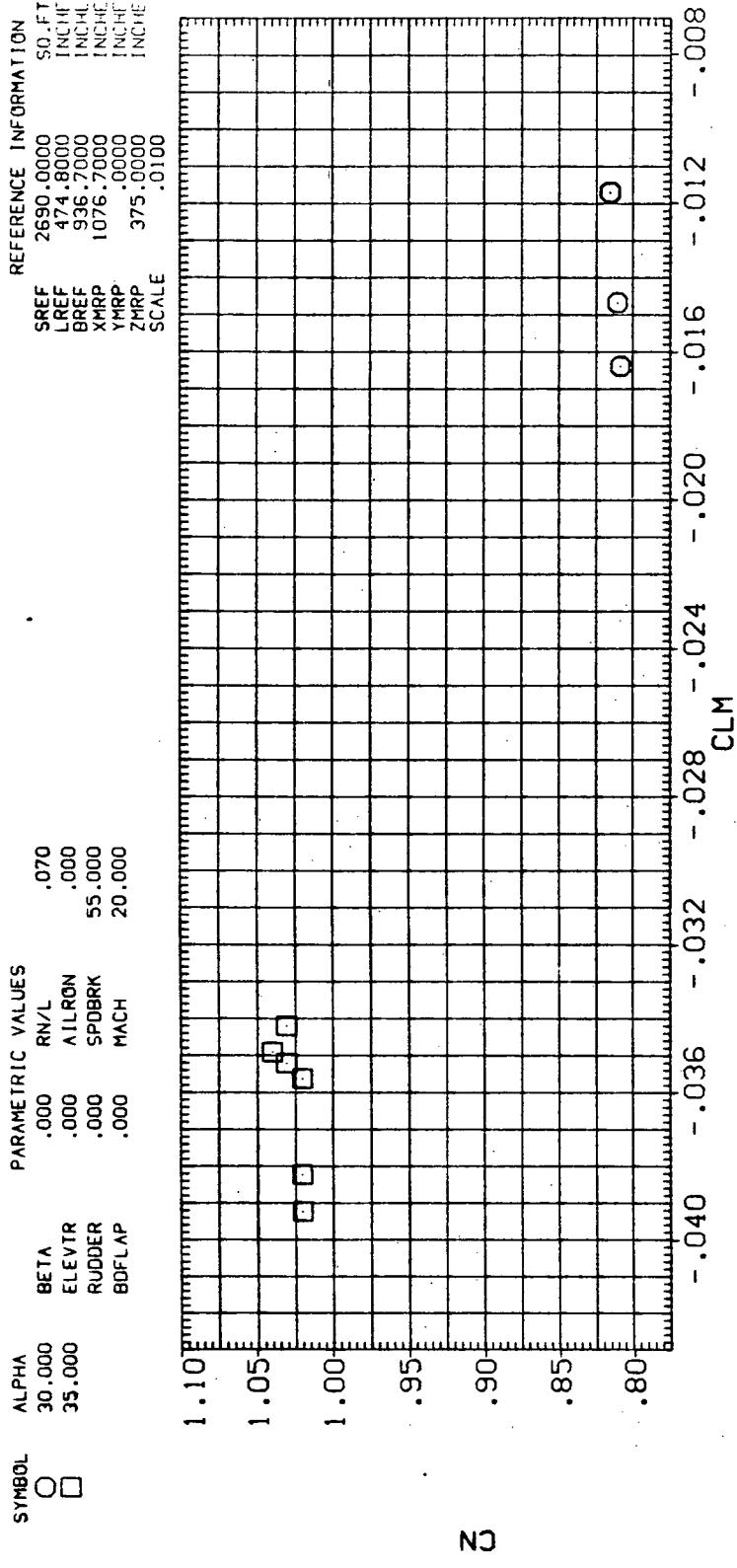
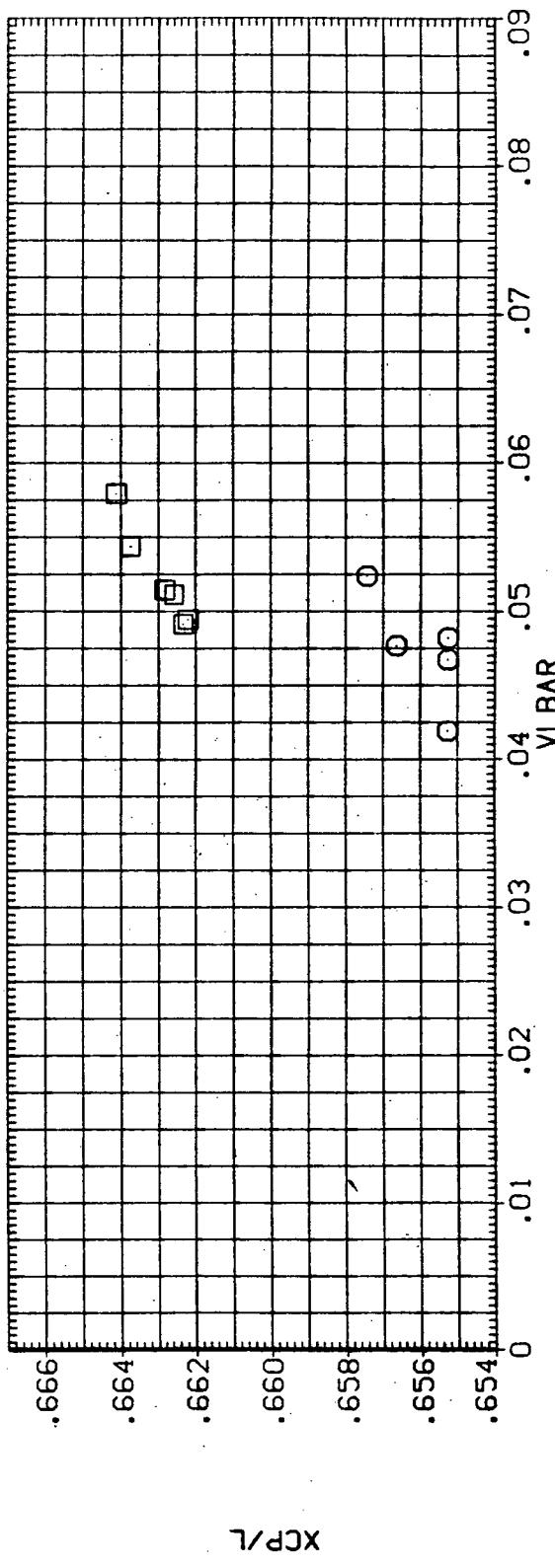
 C_D  CL_M

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VA489(0A-81). (B26C9F7M7N28)(W116E26)(V8R5)(FT0010)

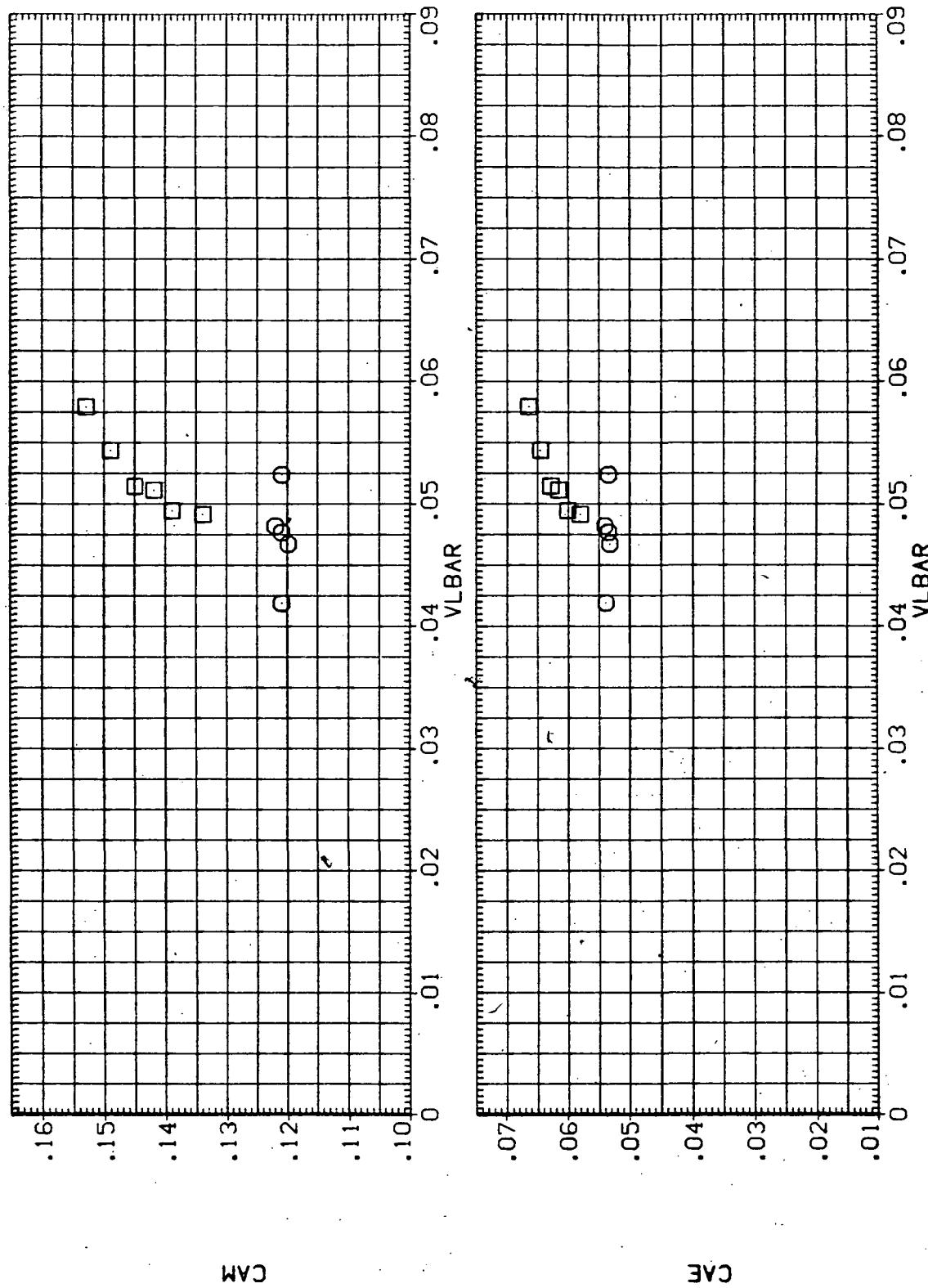
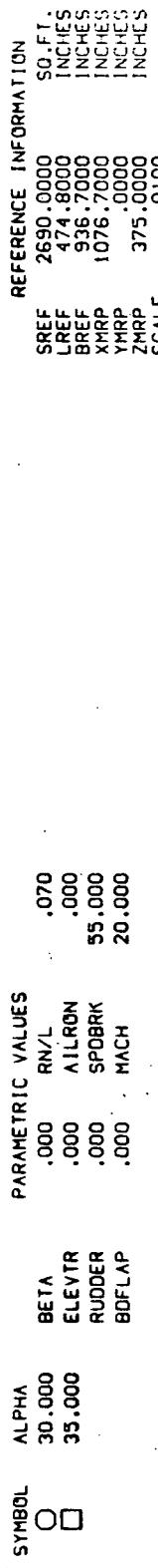


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0010)

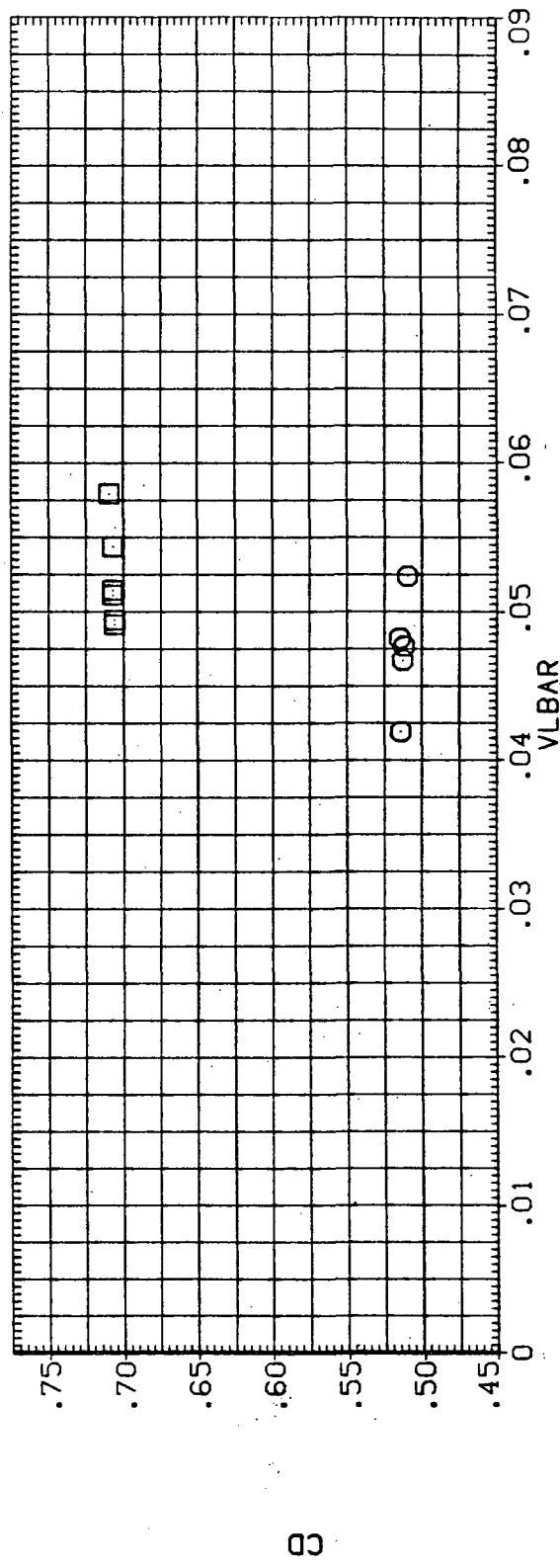
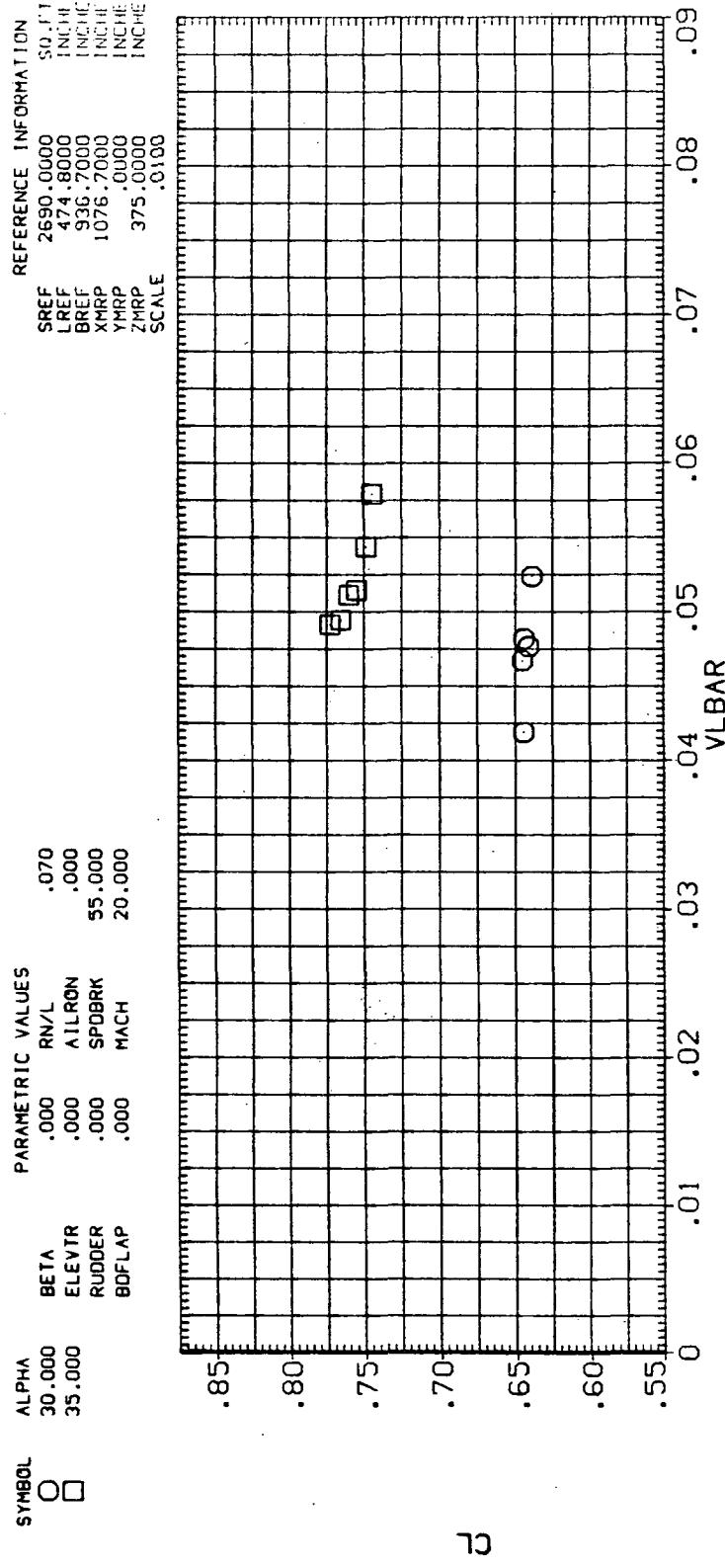


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC V489(0A-81), (826C9F7M7N28)(W116E26)(V8R5)(FT0011)

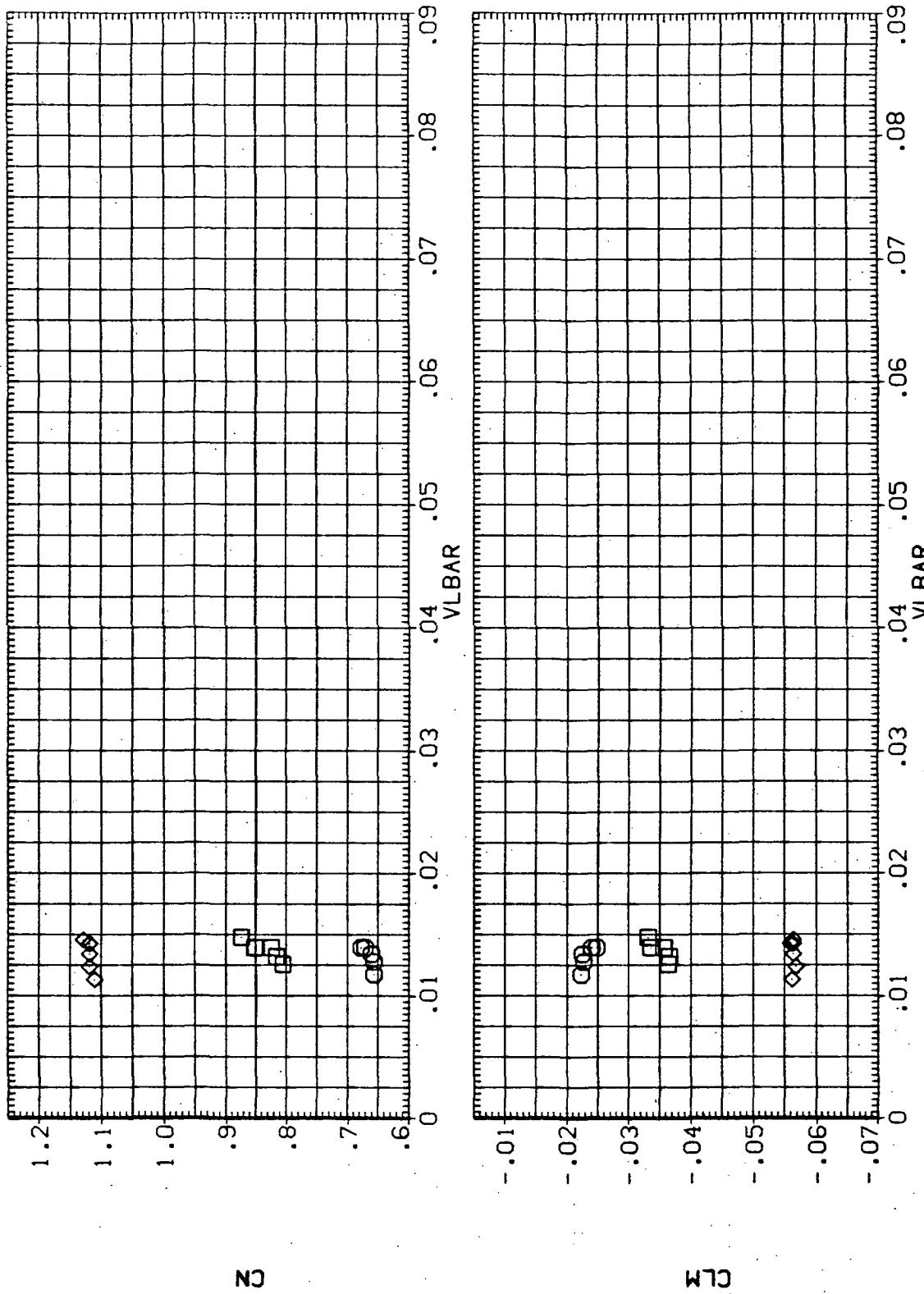
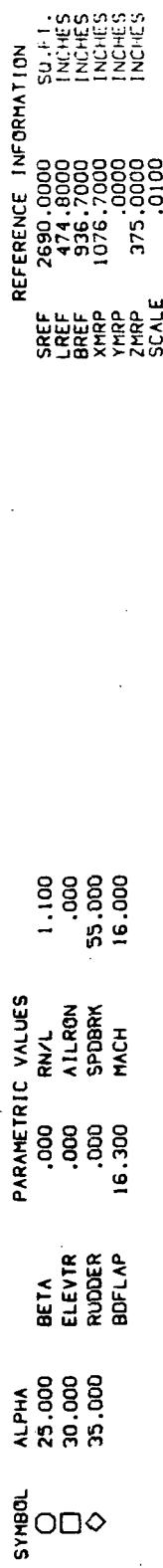


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0011)

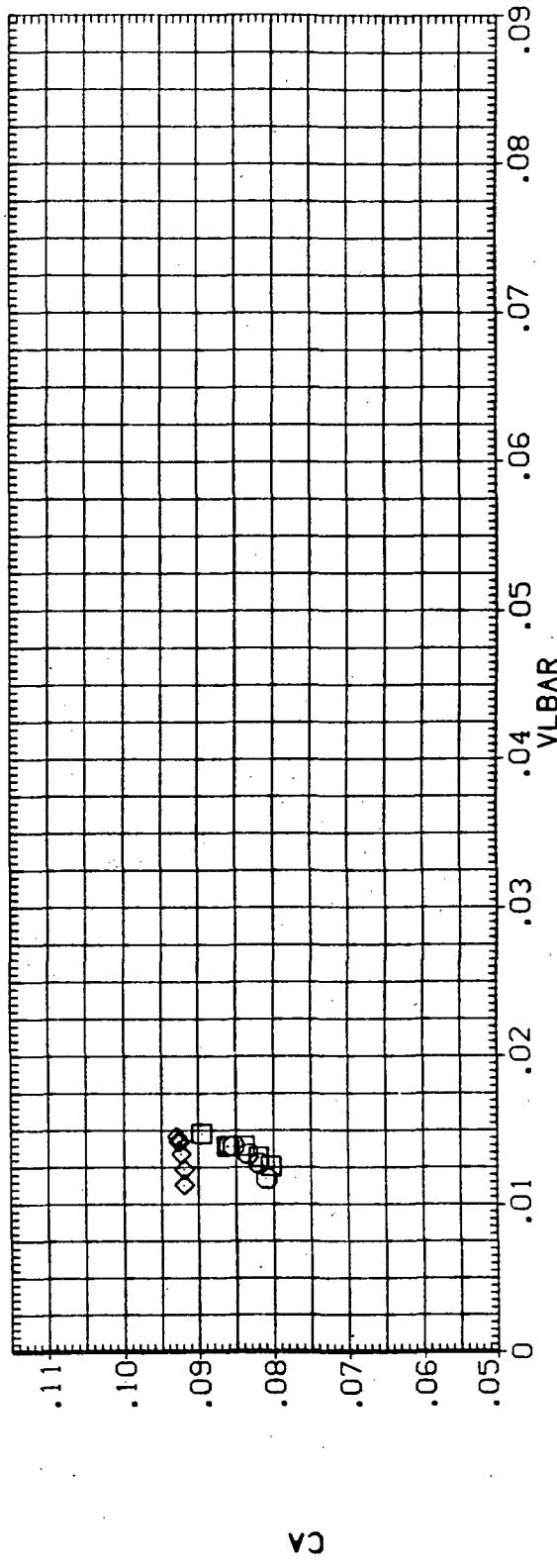
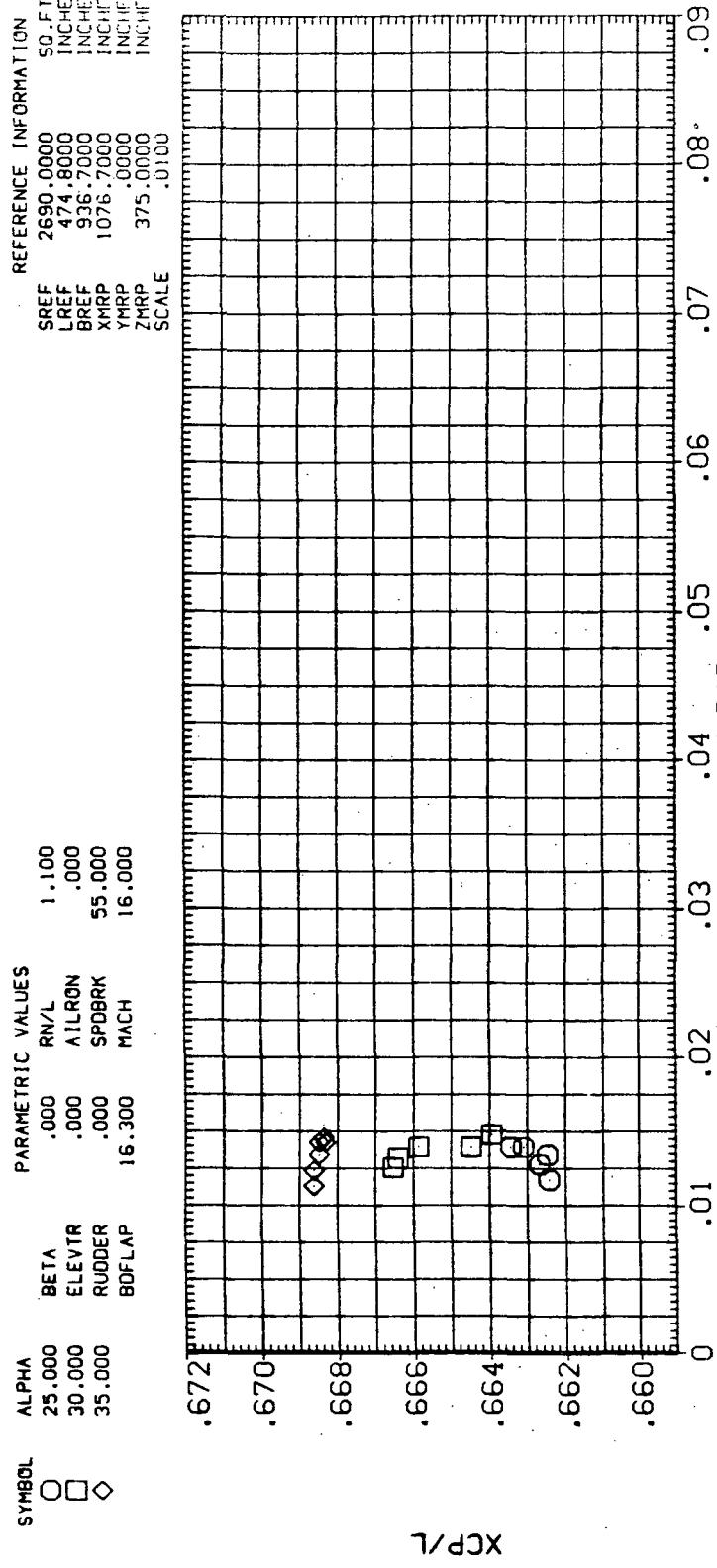


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC V489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0011)

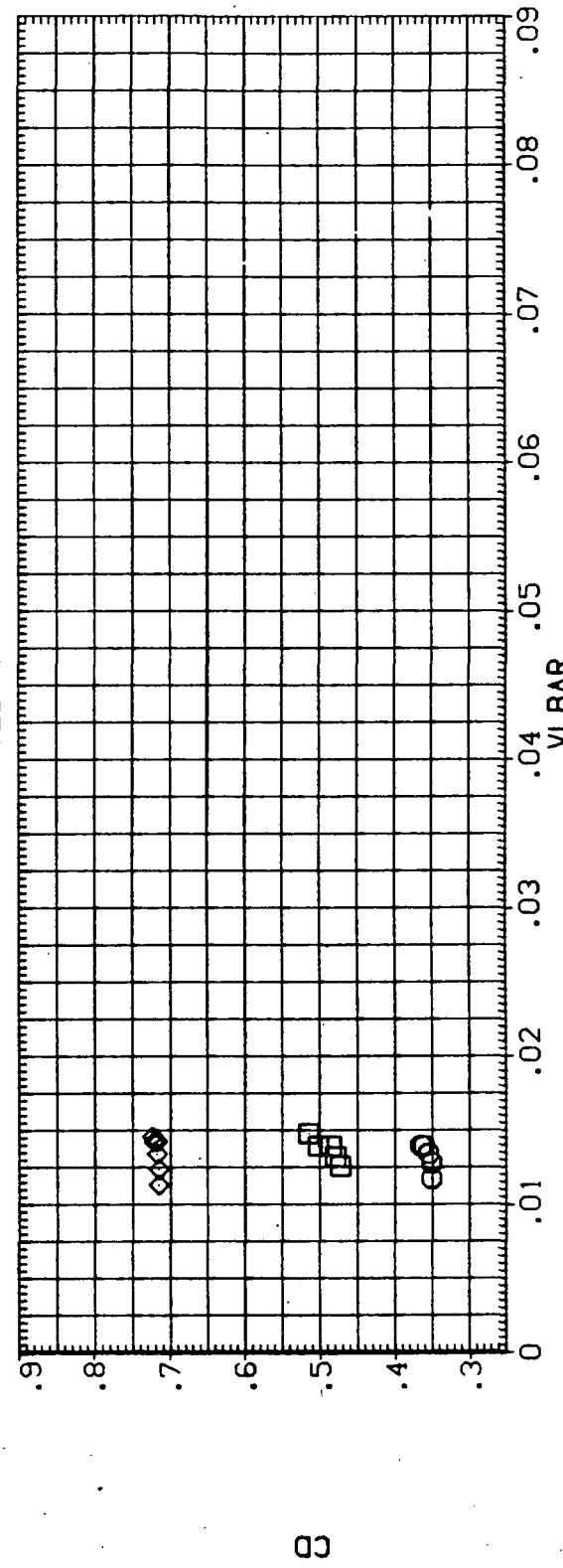
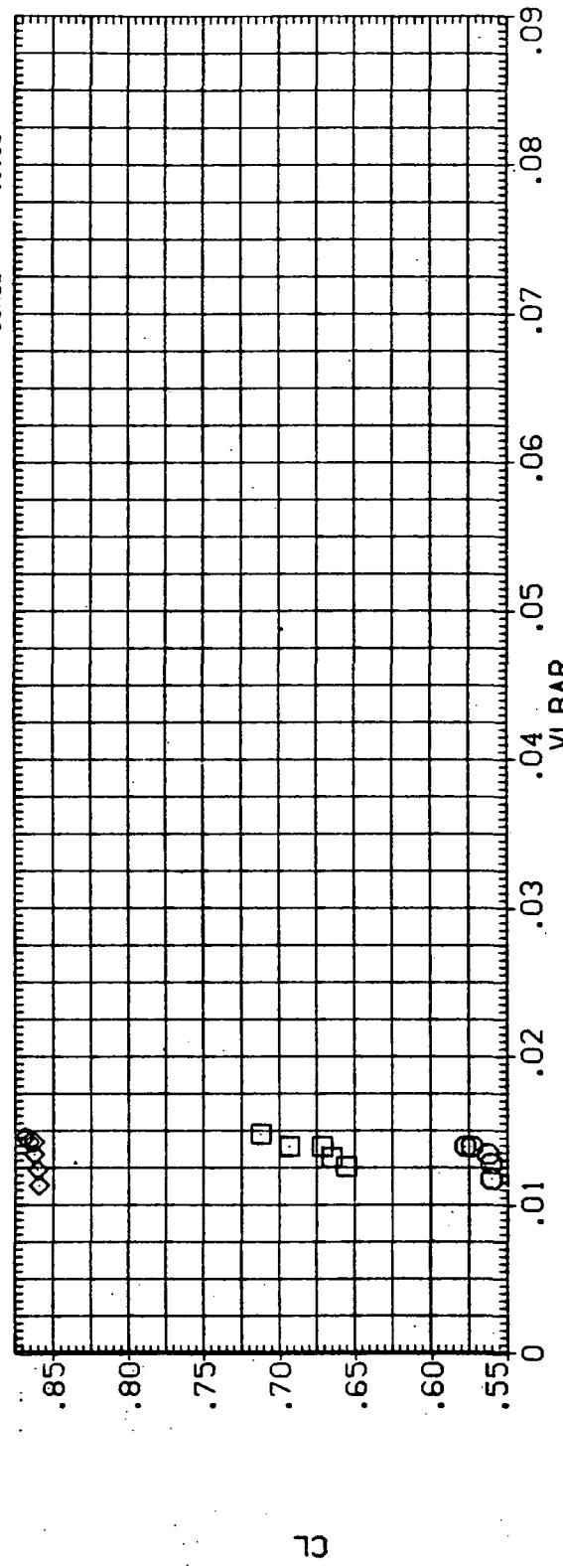


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

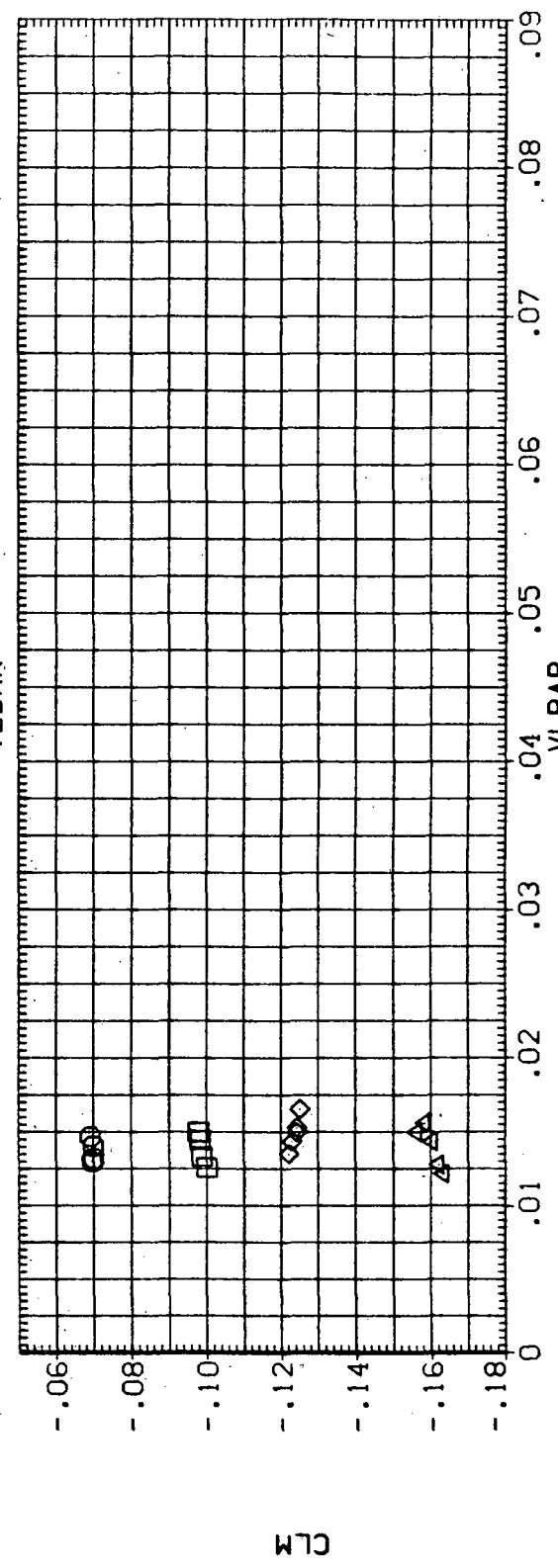
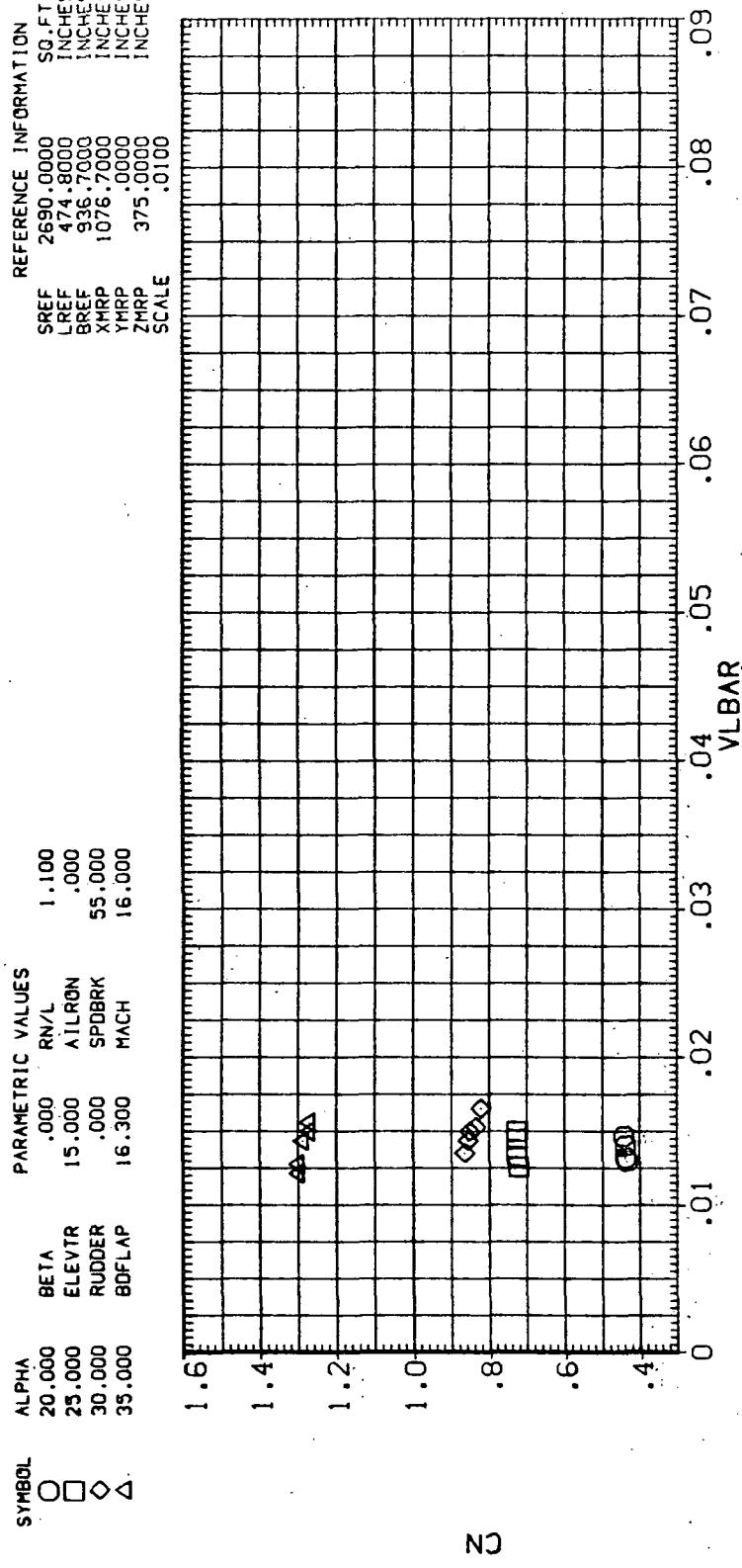


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC V4489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5) (FT0012)

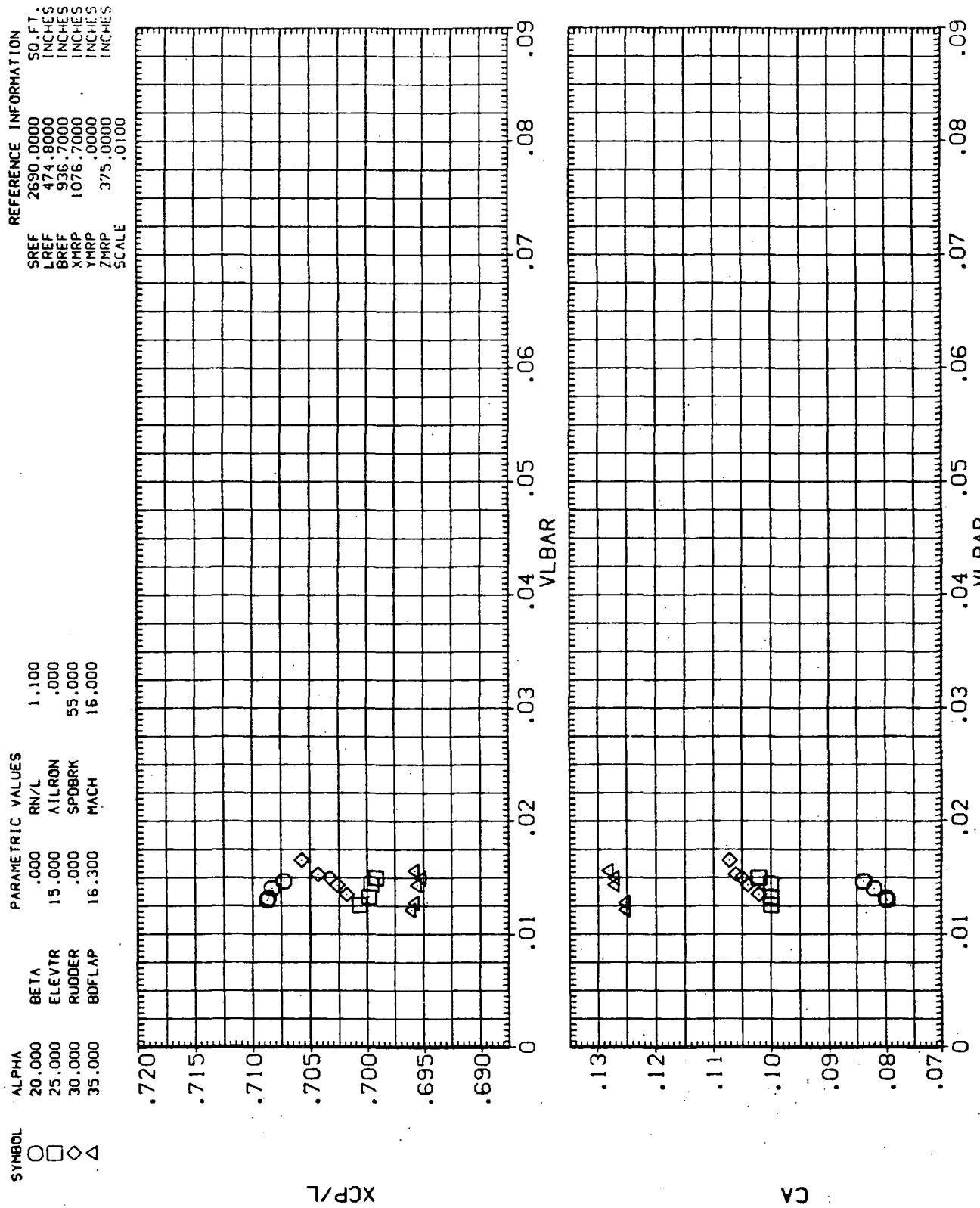
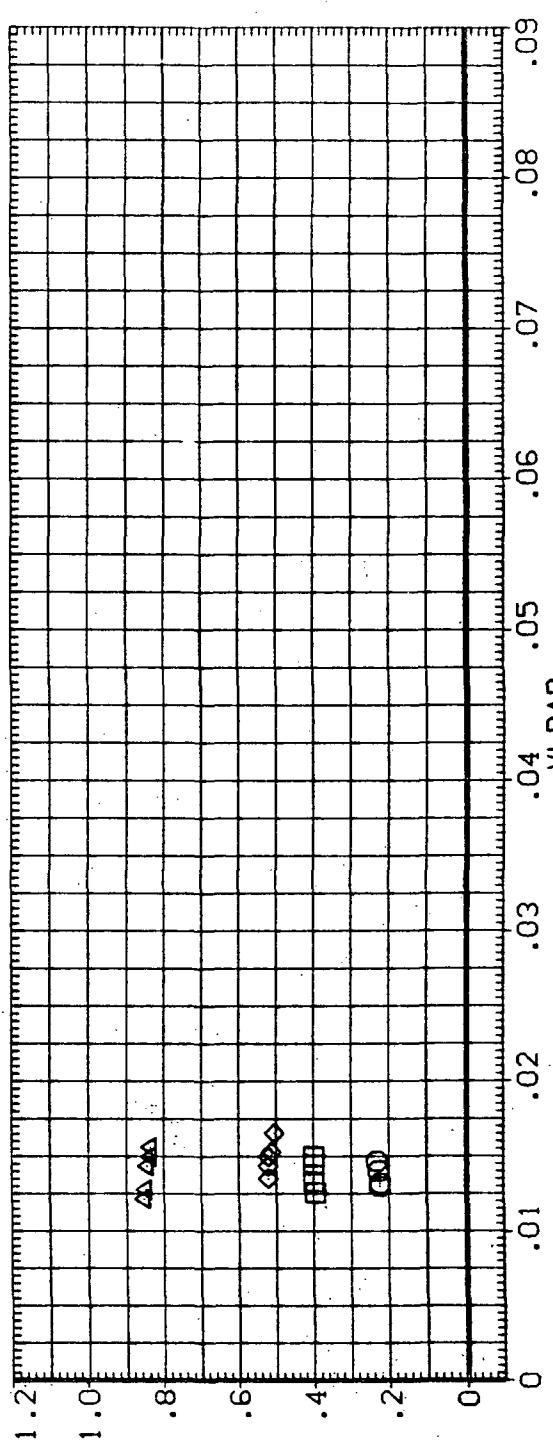


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B



CL



CQ

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0013)

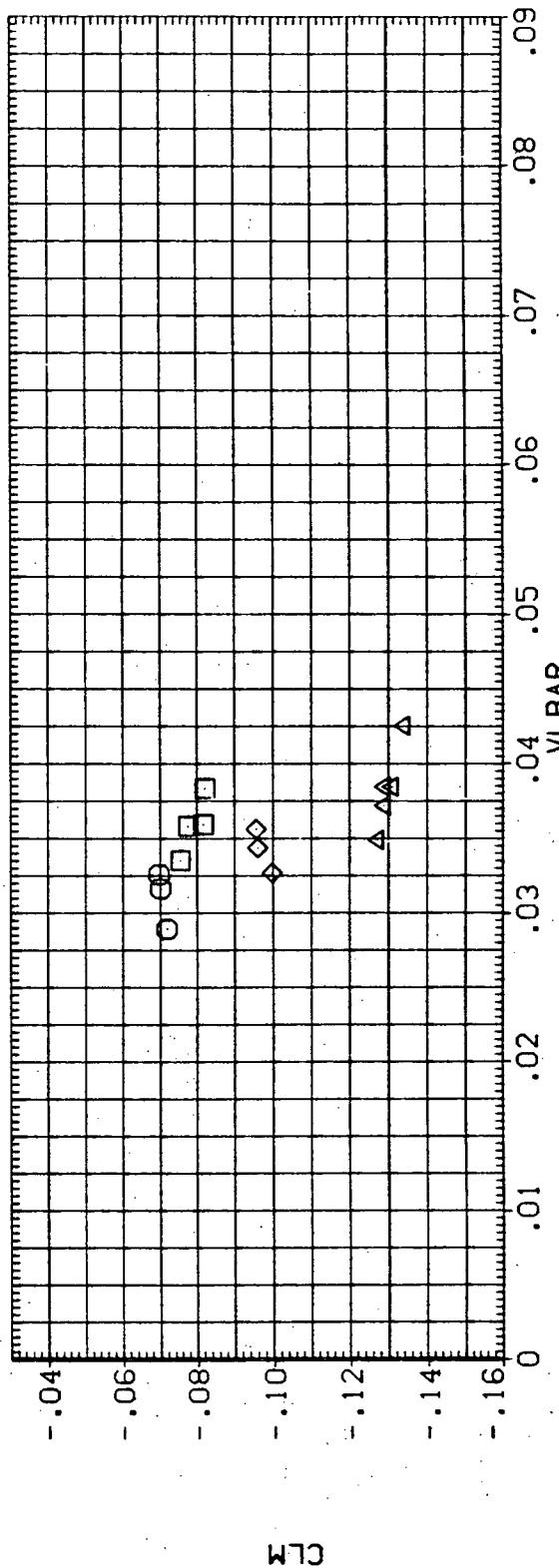
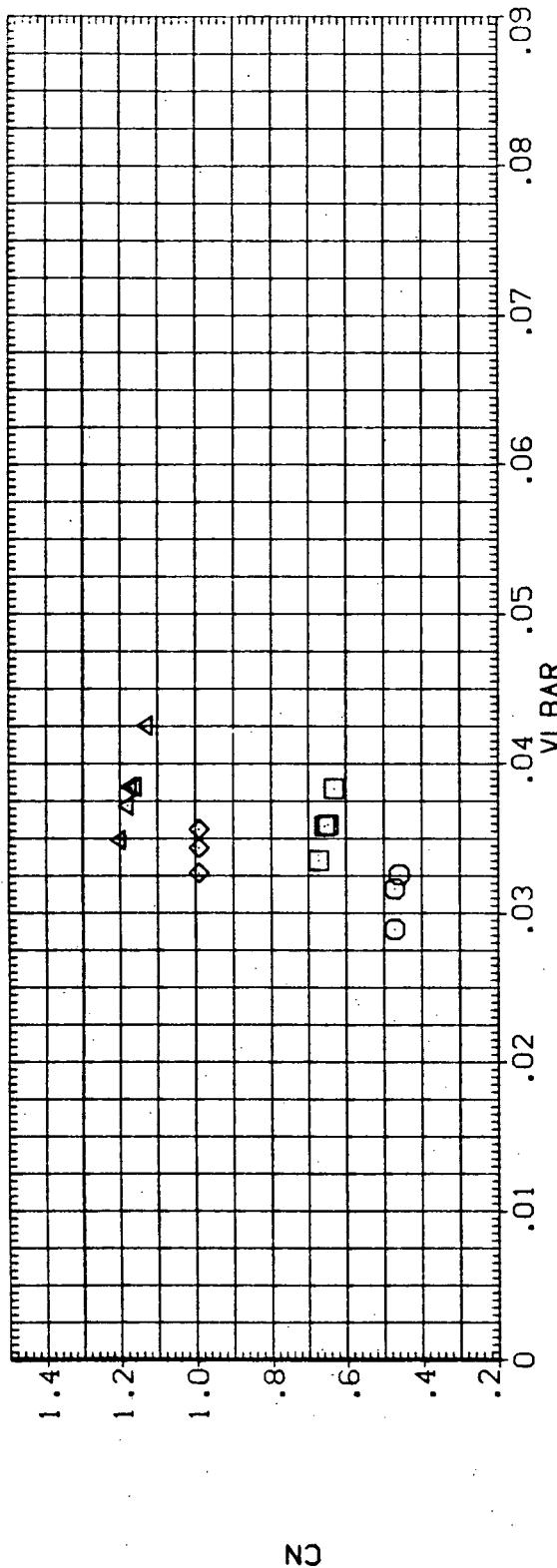


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC VAA489(0A-81), (B26C9F7M7N28)(W116E26)(V8R5)(FT0013)

SYMBOL	PARAMETRIC VALUES			REFERENCE INFORMATION
	ALPHA	BETA	RN/L	
○	20.000	.000	.250	SREF 2690.0000 LREF 474.8000 BREF 936.7000 XMRP 1076.7000 YMRP .0000 ZMRP 375.0000
□	25.000	ELEVTR 15.000	AILRON .000	INCHES
◊	30.000	RUDDER ,000	SPDBRK 55.000	INCHES
△	35.000	BDFLAP 16.300	MACH 20.000	INCHES
				SCALE .0100

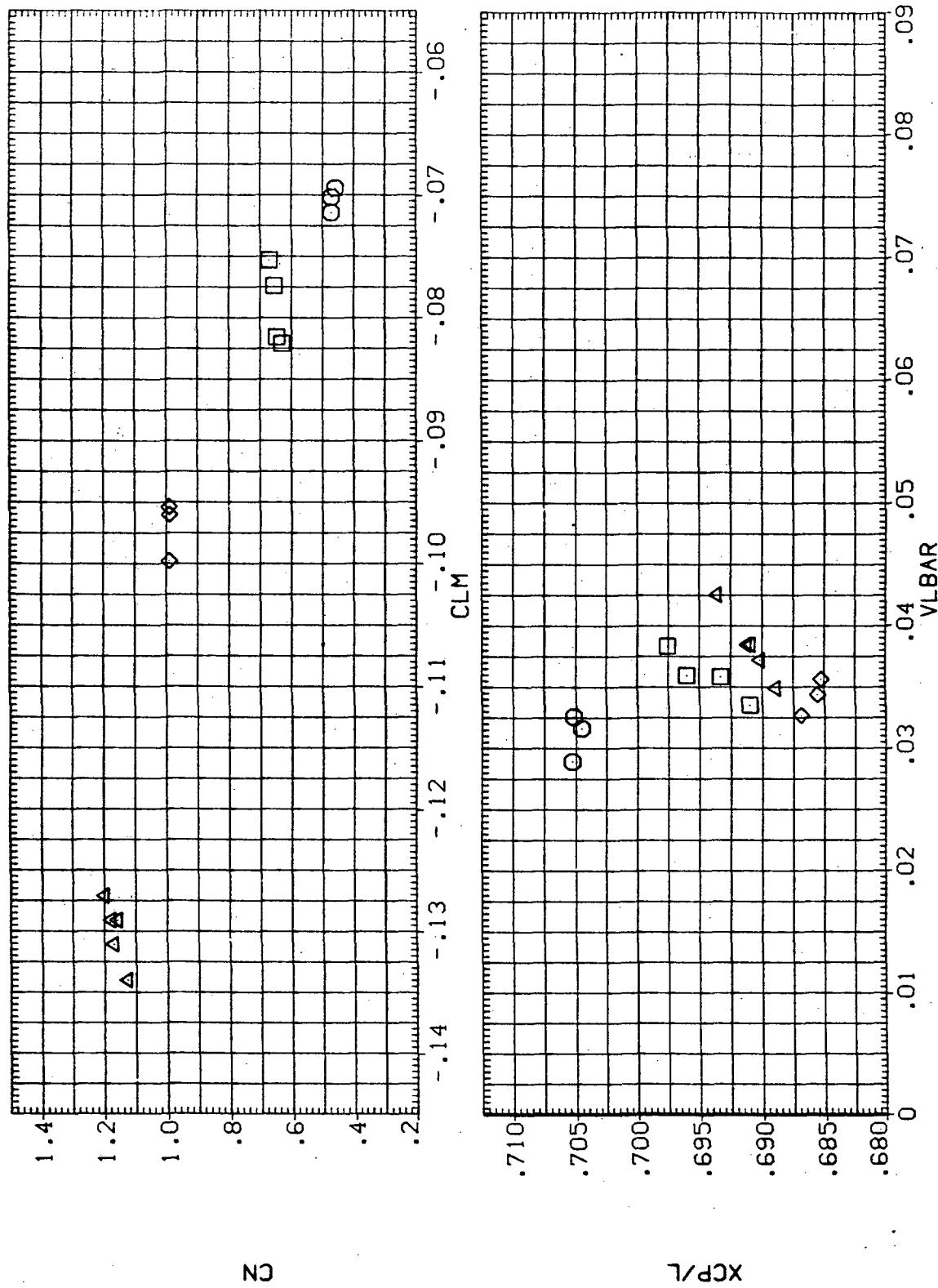


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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AEDC VAA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5) (FT0013)

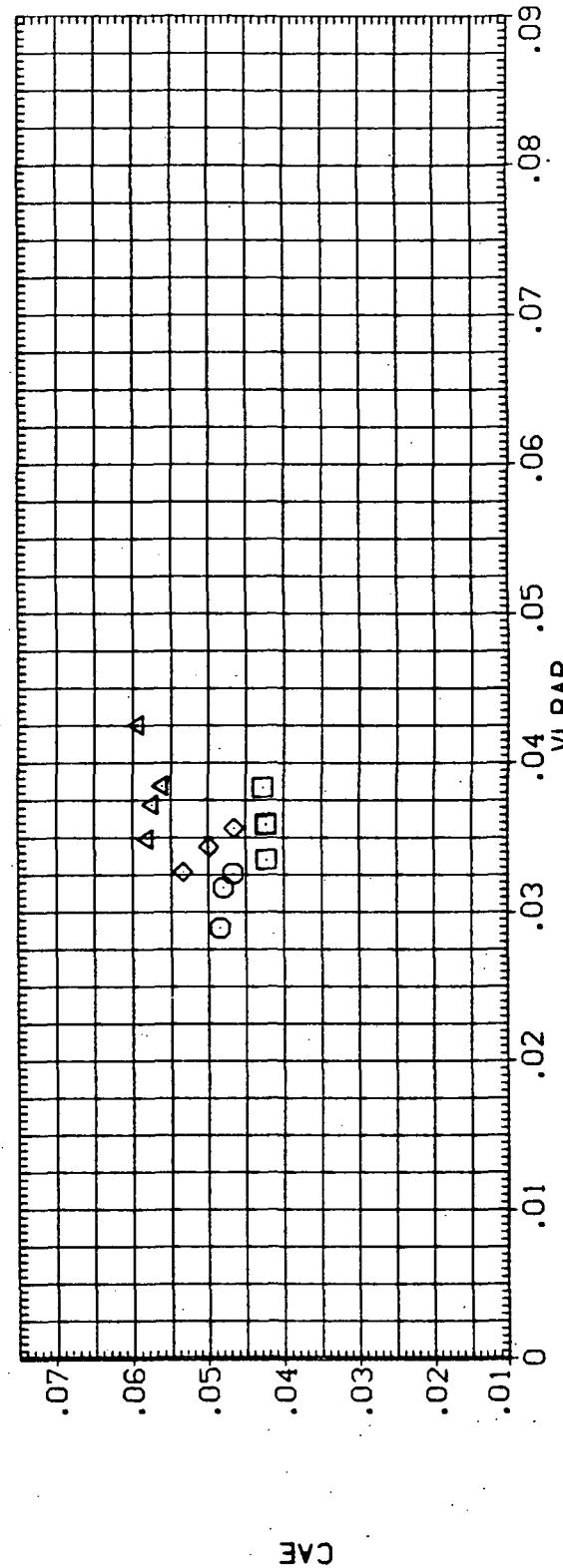
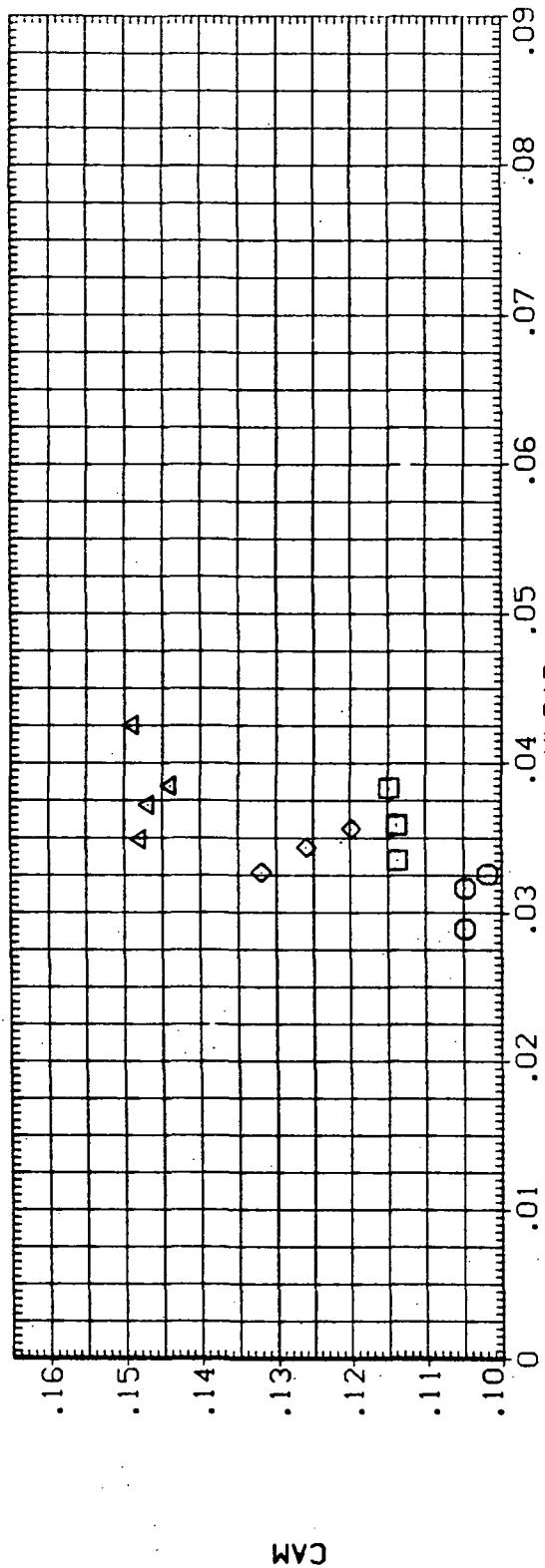
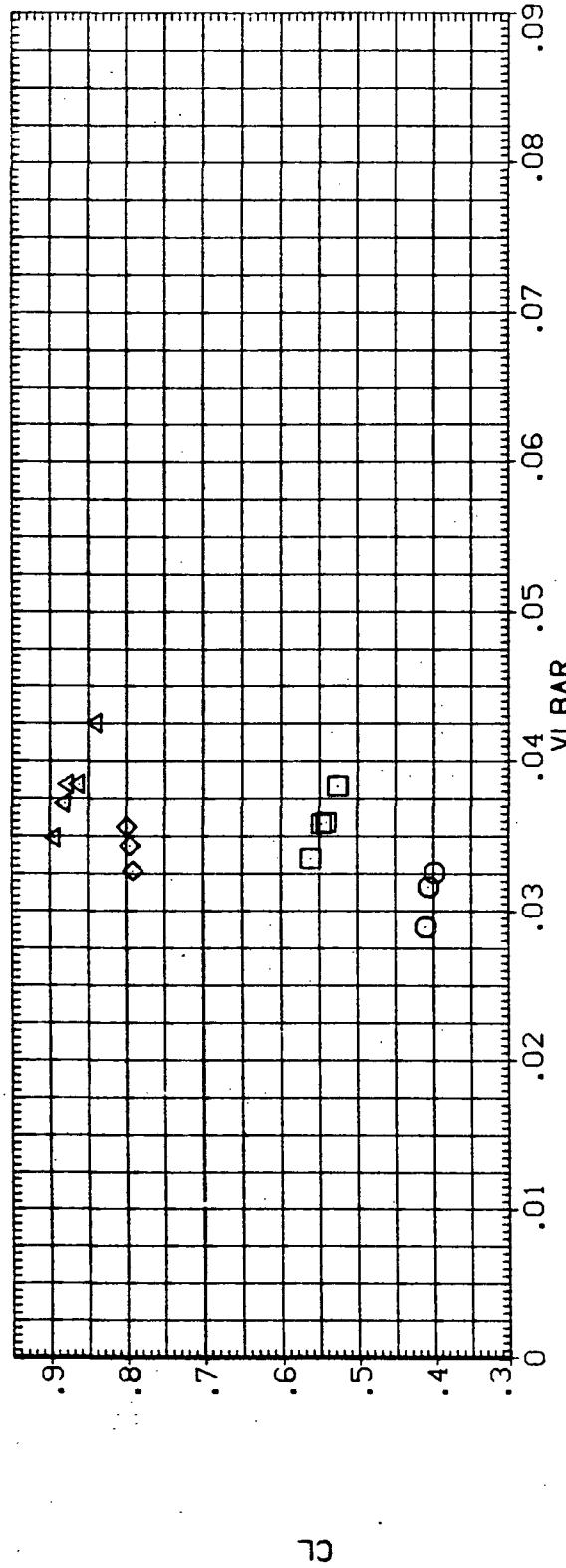
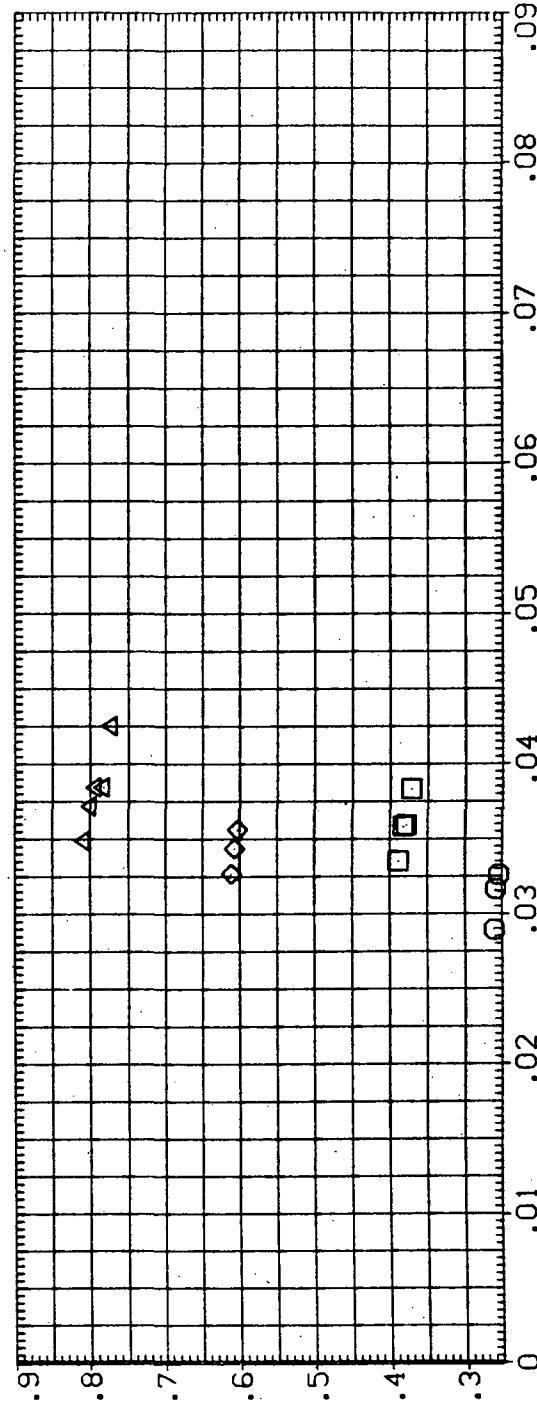


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

SYMBOL	ALPHA	PARAMETRIC VALUES		REFERENCE INFORMATION
		BETA	RNL	
O	20.000	.000	.250	SREF 2690.0000 SQ.FT.
□	25.000	15.000	.000	LREF 474.8000 INCHES
◊	30.000	.000	55.000	BREF 936.7000 INCHES
△	35.000	16.300	20.000	XMRP 1076.7000 INCHES
				YMRP .0000 INCHES
				ZMRP 375.0000 INCHES
				SCALE .0100



CL



CG

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC489(0A81) B26C9F7M7N28 W116E26 V8R5 INVERTED (FT0014)

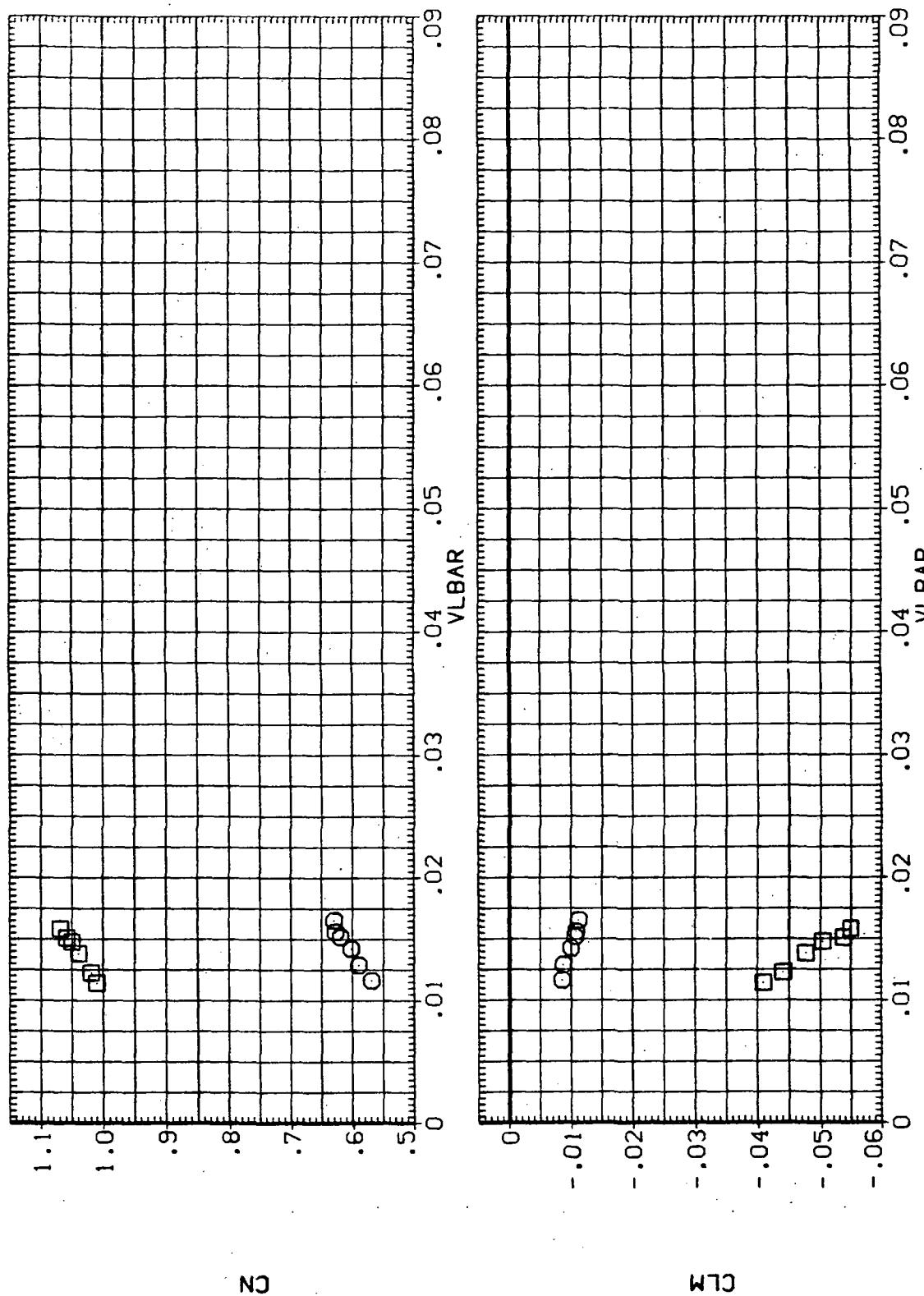


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

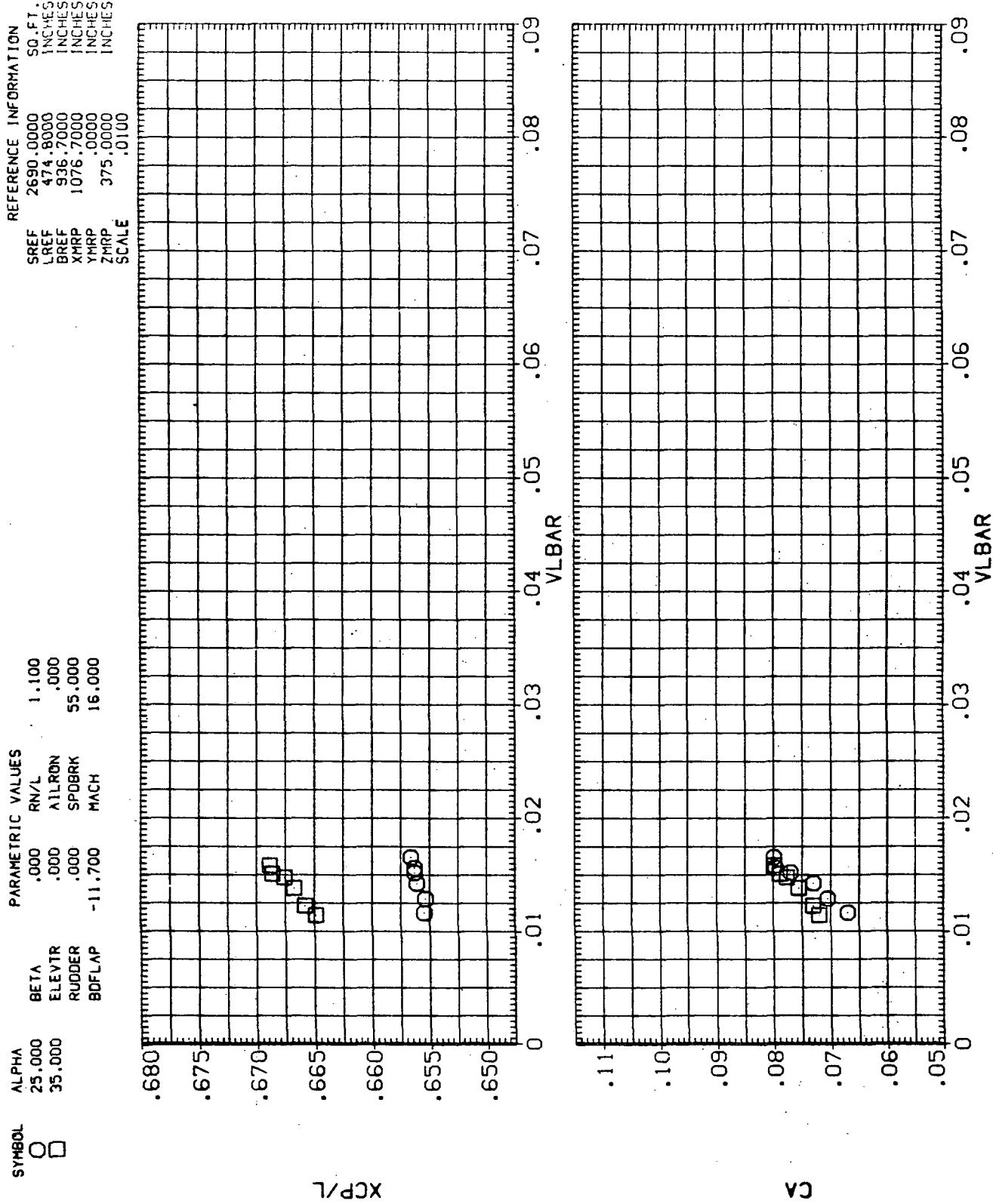
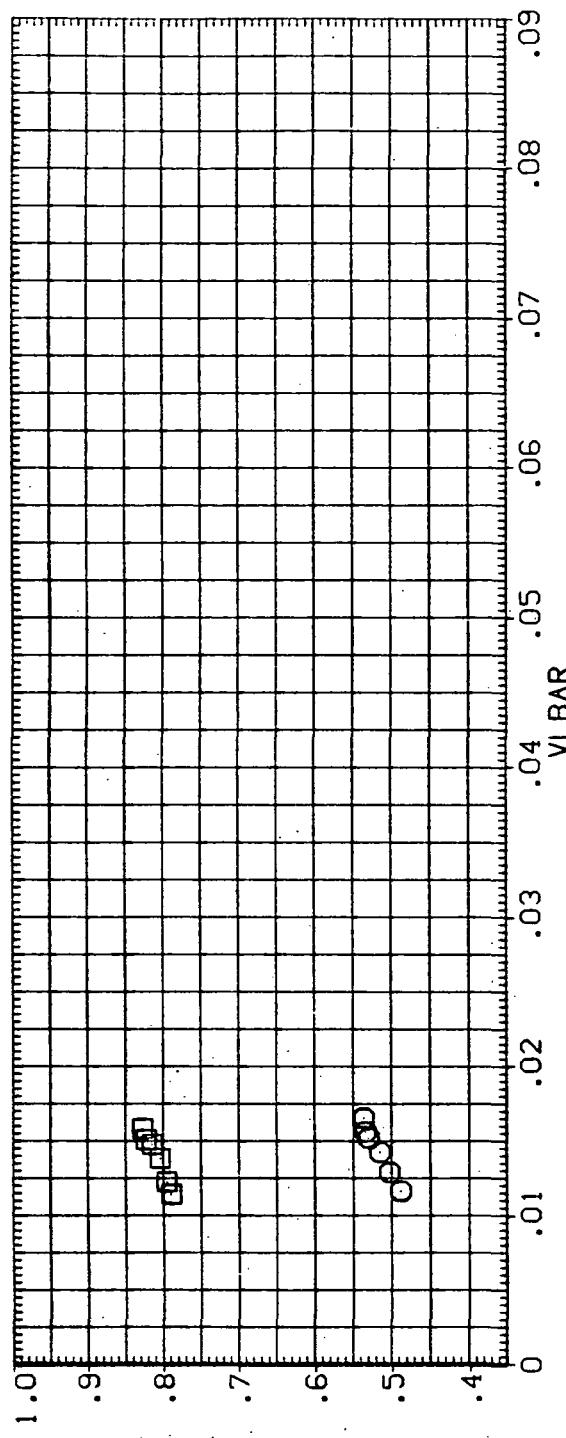


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

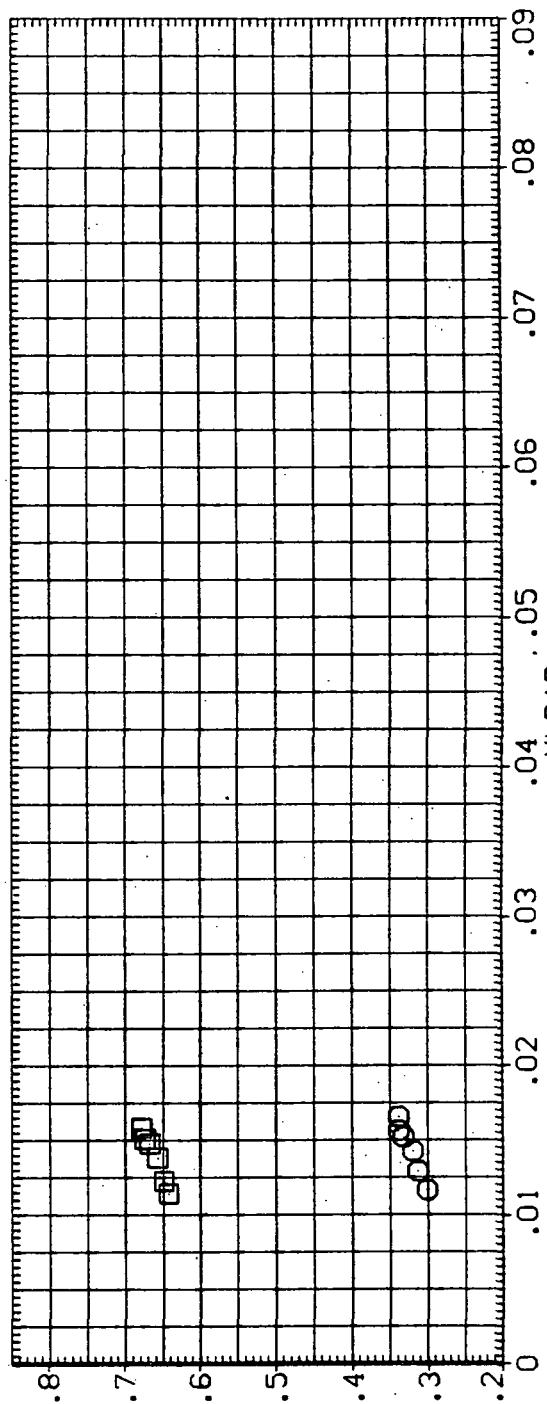
AEDC489(0A81) B26C9F7M7N28 W116E26 V8R5 INVERTED (FT0014)

SYMBOL	ALPHA	BETA	PARAMETRIC VALUES	RNL	1.100
O	25,000	.000	AIRON	.000	
□	35,000	.000	SPOBRA	55.000	
			RUDDER	16.000	
			BOFLAP	-11.700	MACH

REFERENCE INFORMATION
 SREF 2690.0000 SQ.FT.
 LREF 474.8000 INCHES
 BREF 936.7000 INCHES
 XMRP 1076.7000 INCHES
 YMRP 375.0000 INCHES
 ZMRP .0100 SCALE



C



D

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC489(0A81) B26C9F7M7N28 W116E26 V8R5 INVERTED(FT0015)

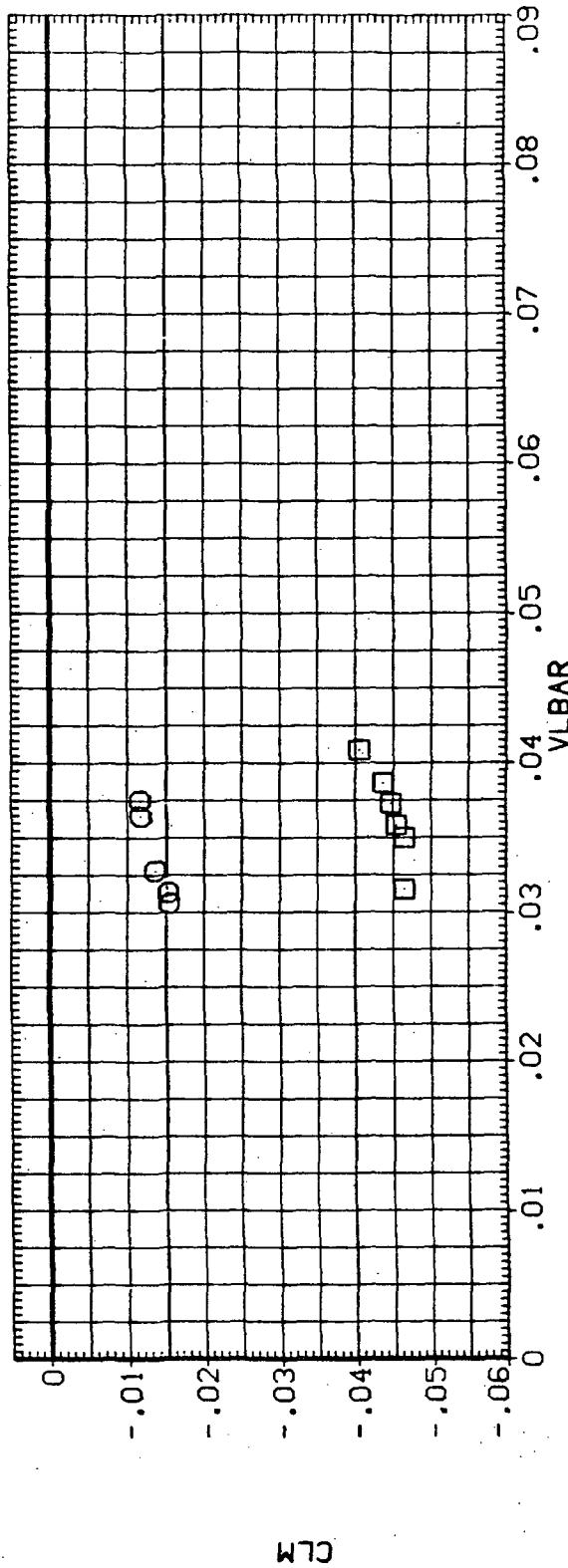
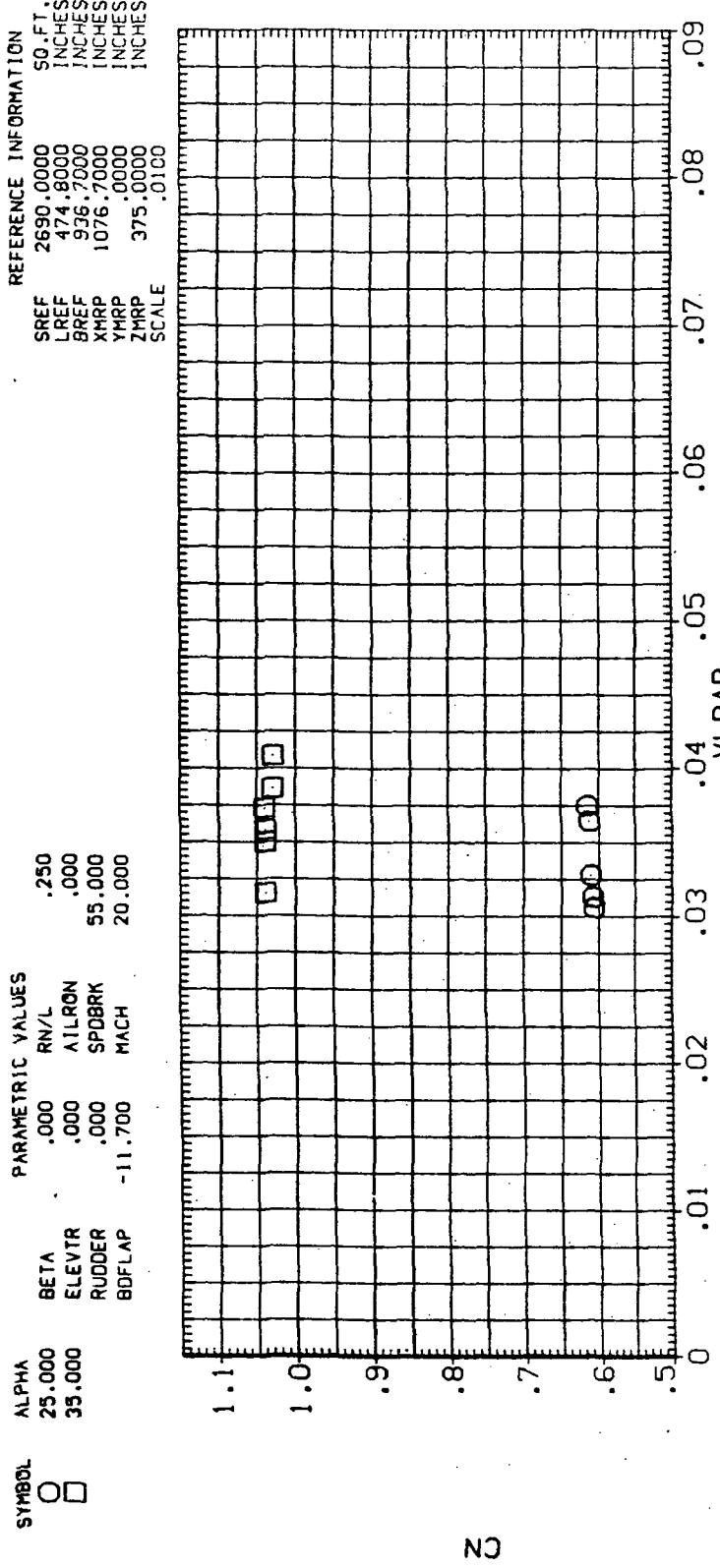


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

PAGE 82

AEDC489(0A81) B26C9F7M7N28 W116E26 V8R5 INVERTED (FT0015)

SYMBOL	ALPHA	PARAMETRIC VALUES			REFERENCE INFORMATION
		BETA	RN/L	AILRON	
○	25.000	.000	.250		SREF 2690.0000 LREF 474.8000 BREF 936.7000 XMRP 1076.7000 YMRP .0000 ZMRP 375.0000 SCALE .0100
□	35.000	.000	.000		
	ELEVTR	.000			
	RUDDER	.000			
	SPDBRK	55.000			
	BOFLAP	-11.700	MACH		

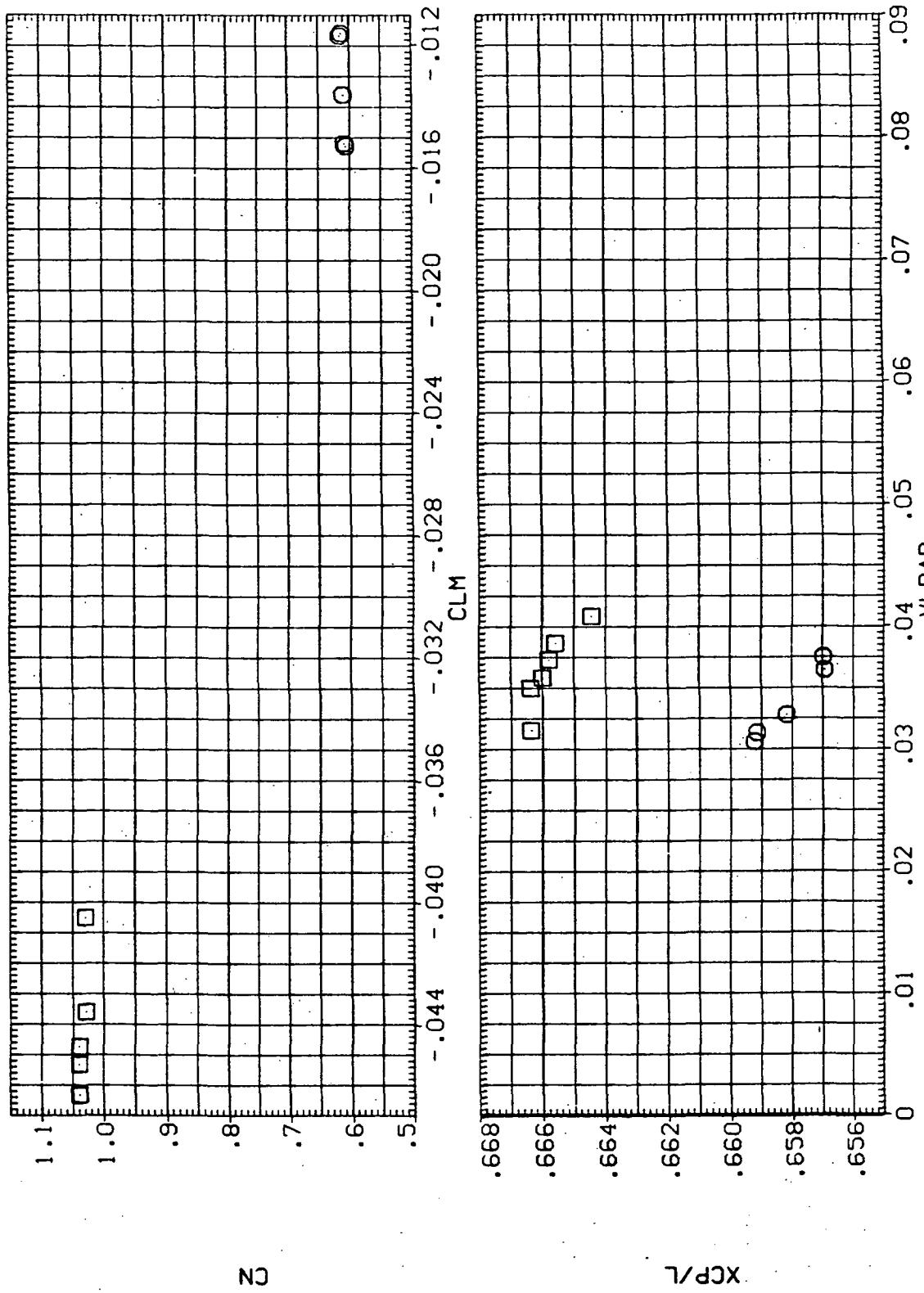
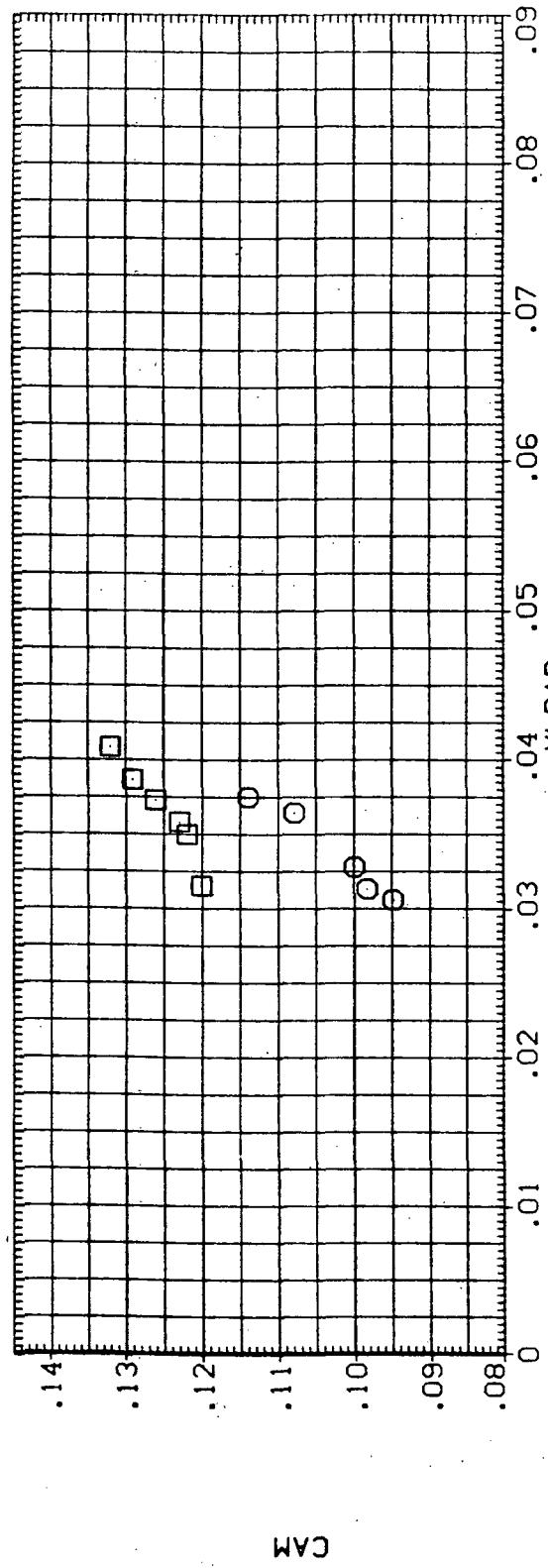


FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

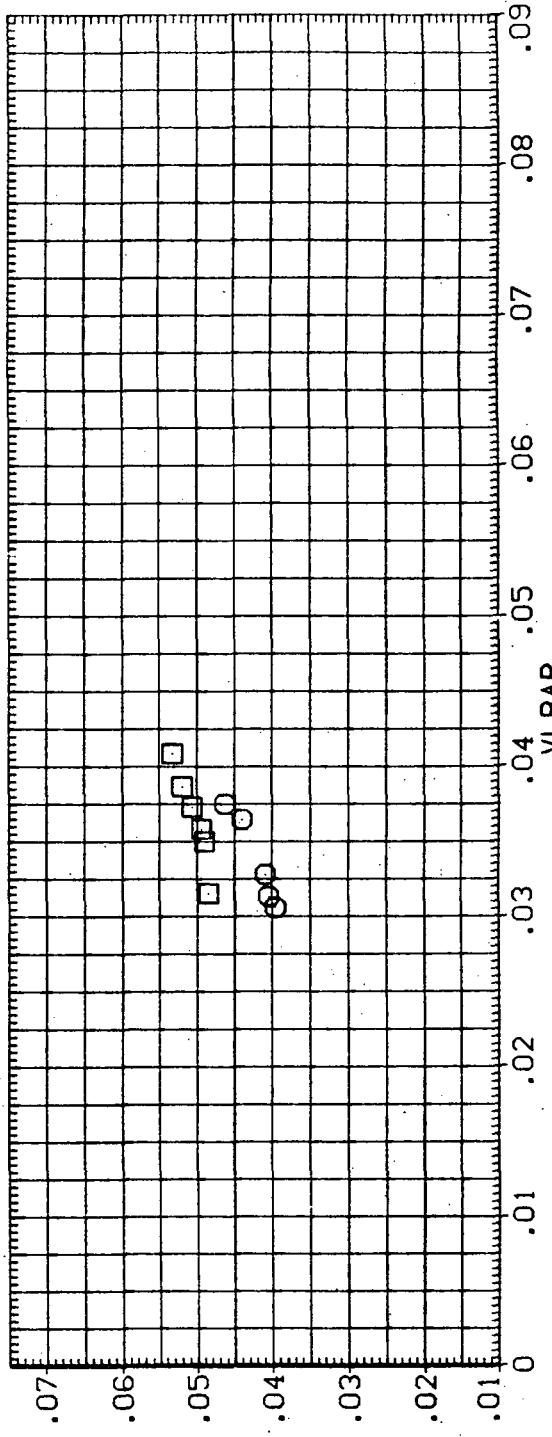
AEDC489(0A81) B26C9F7M7N28 W116E26 V8R5 INVERTED(FT0015)

REFERENCE INFORMATION
SREF 2690.0000 SQ.FT.
LREF 474.8000 INCHES
BREF 936.7000 INCHES
XHMP 1076.7000 INCHES
YMRP .0000 INCHES
ZMRP 375.0000 INCHES
SCALE .0100

PARAMETRIC VALUES
SYMBOL ALPHA .25.000 BETA .000 RNL .250
□ ELEVTR .000 AILRON .000
RUDDER .000 SPDBRK 55.000
BDFLAP -11.700 MACH 20.000



CAE



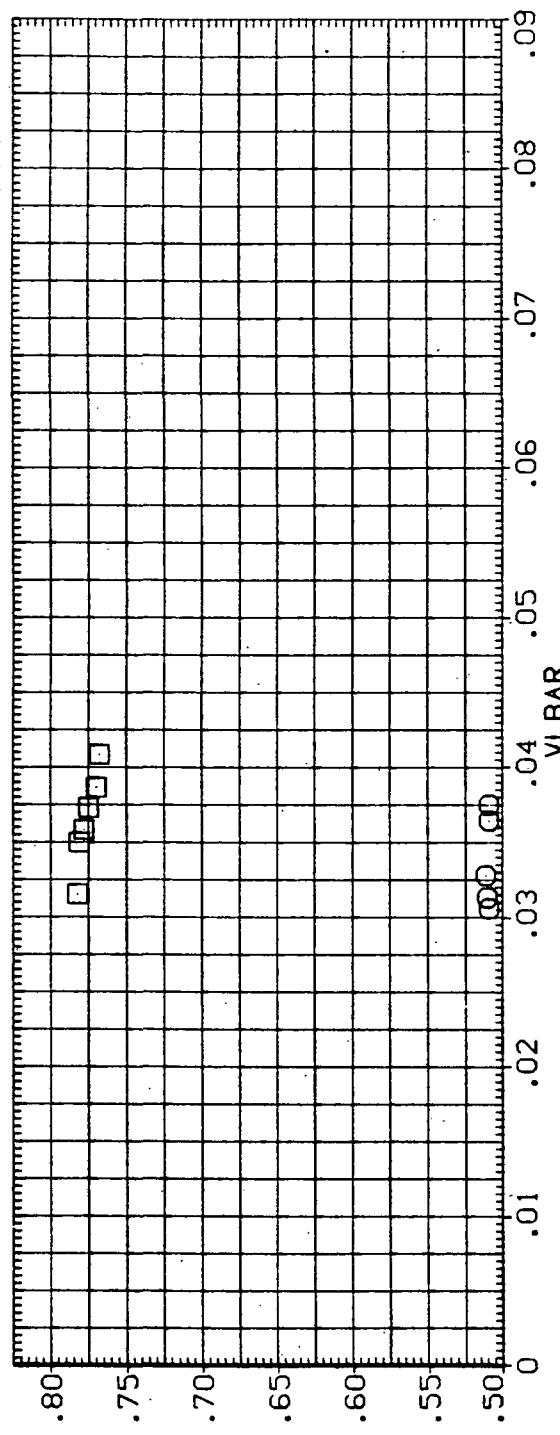
CAE

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

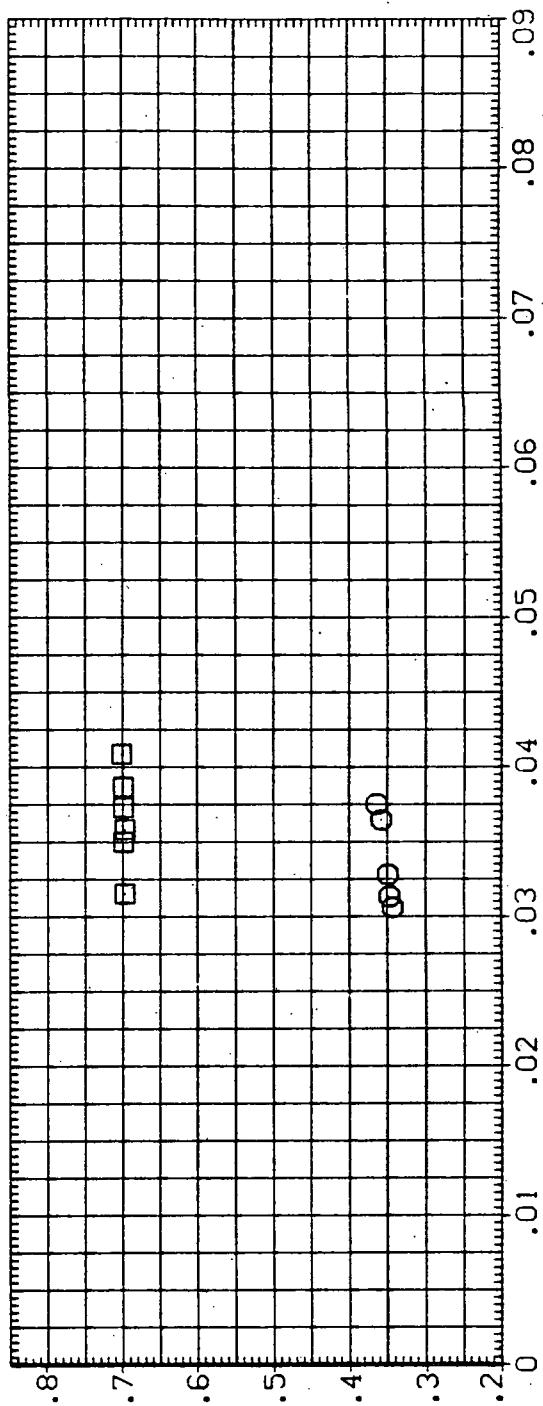
AEDC489(0A81) B26C9F7M7N28 W116E26 V8R5 INVERTED(FT0015)

PARAMETRIC VALUES	ALPHA	BETA	RN/L
.000	25,000	.000	.250
.000	35,000	AIRON	.000
.000	RUDDER	SPOBRK	55,000
-11.700	BDFLAP	MACH	20,000

REFERENCE INFORMATION	SO.FT.
SREF	2690,0000
LREF	474,8000
BREF	936,7000
XMRP	1076,7000
YMRP	375,0000
SCALE	.0100



C



C

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

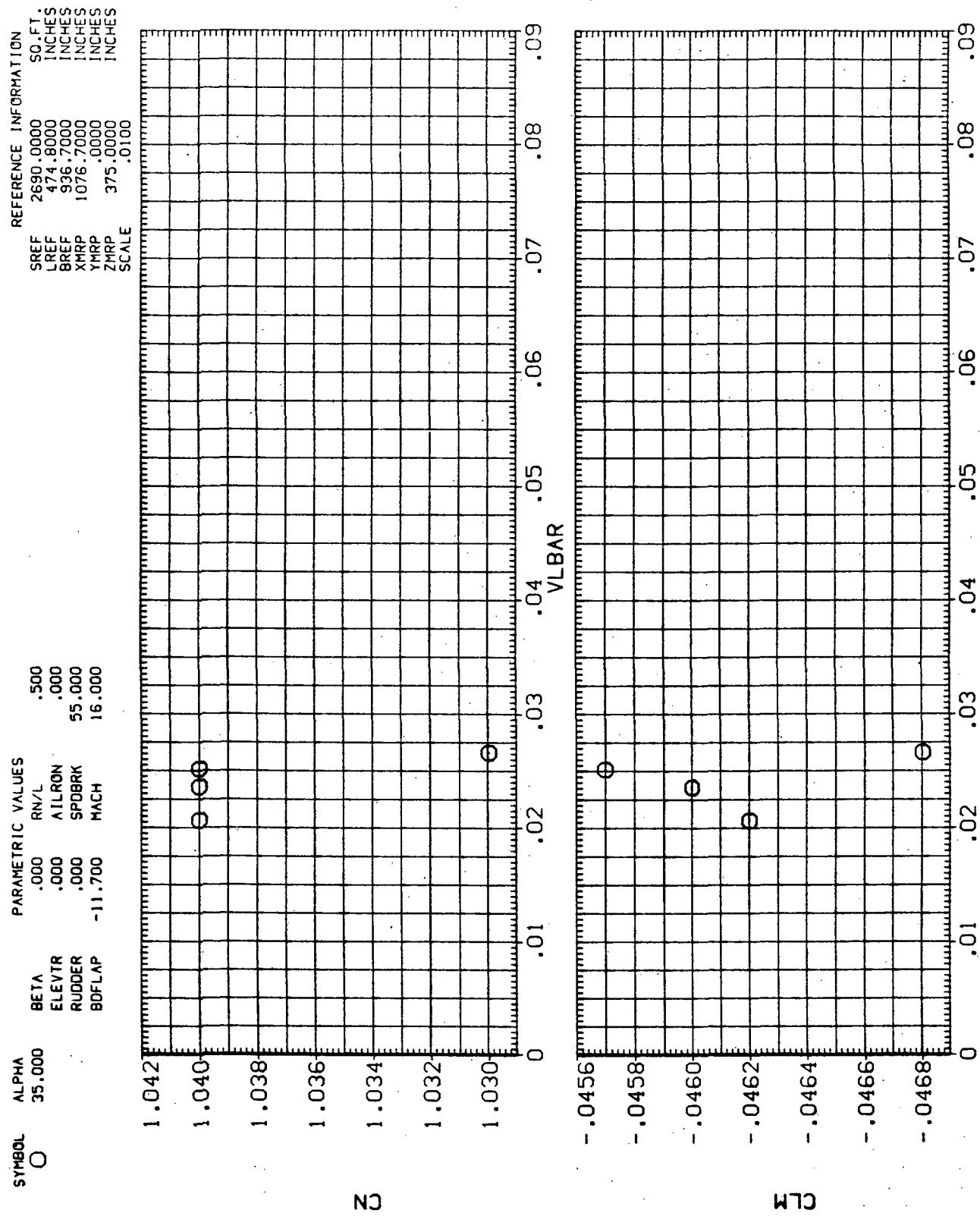
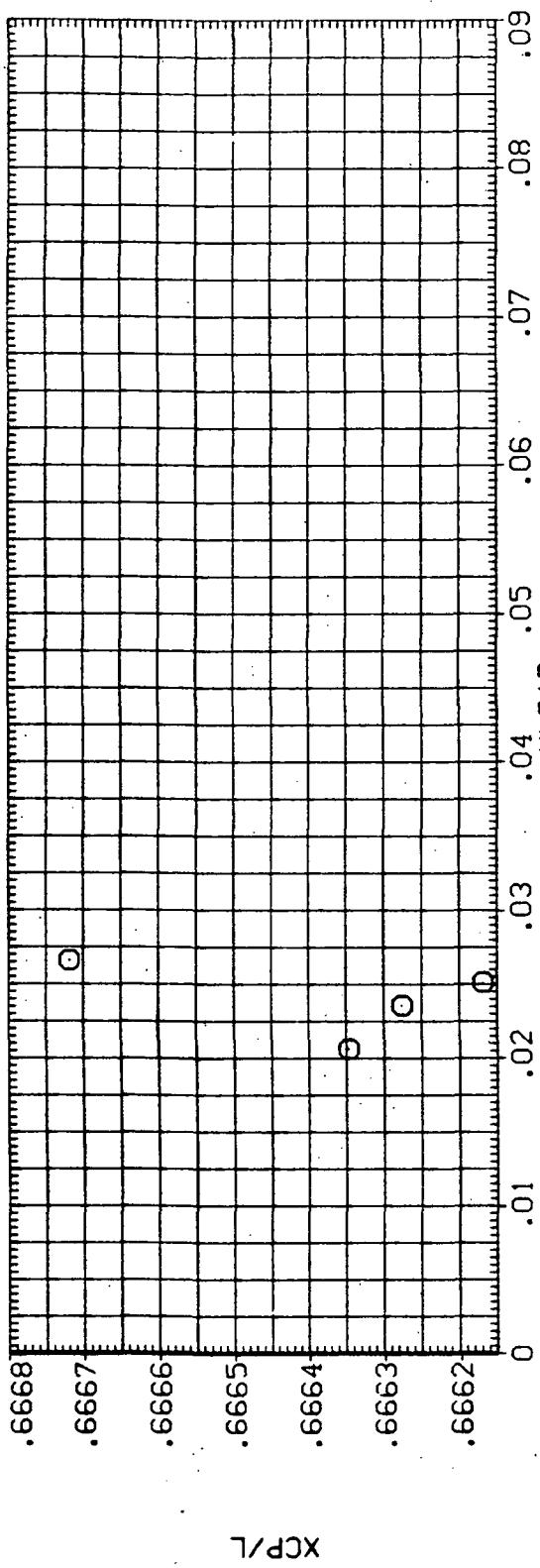


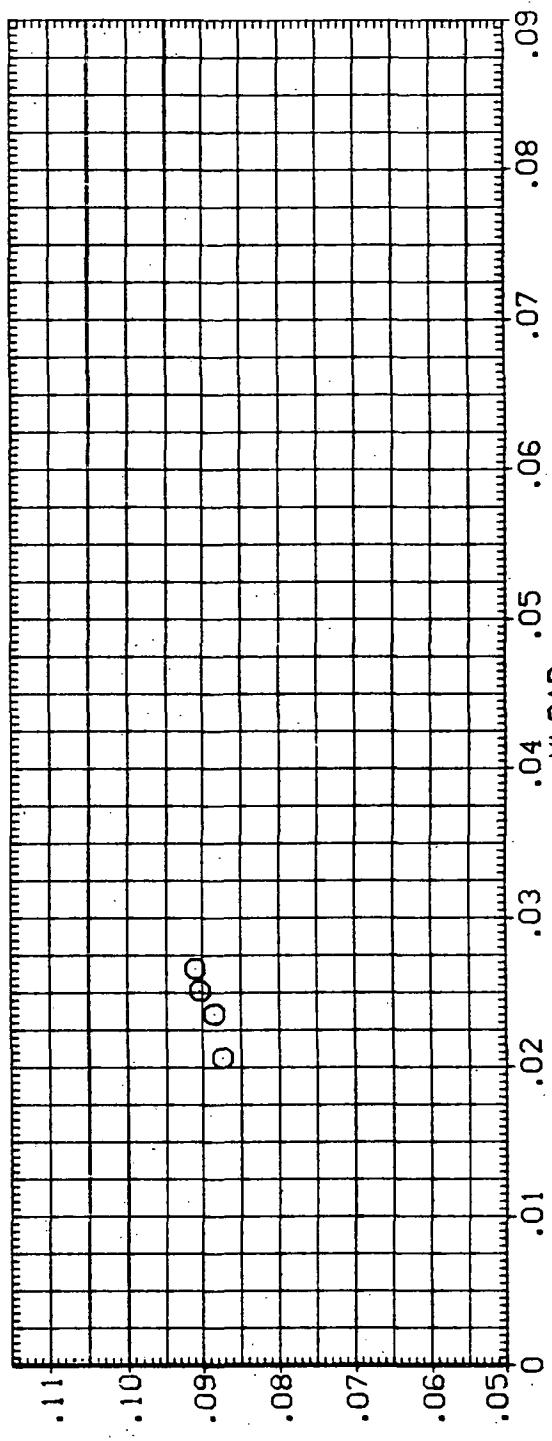
FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC489(0A81) B26C9F7M7N28 W116E26 V8R5 INVERTED (FT0016)

SYMBOL	ALPHA	PARAMETRIC VALUES			REFERENCE INFORMATION	
		BETA	RN/L	.500		
O	35.000	.000	AIRON	.000	LREF	474.8000
	ELEVTR	.000	SPDBRK	55.000	BREF	936.7000
	RUDDER	.000	MACH	16.000	XMRP	1076.7000
	BDFLAP	-11.700			ZMRP	375.0000
					SCALE	.0100



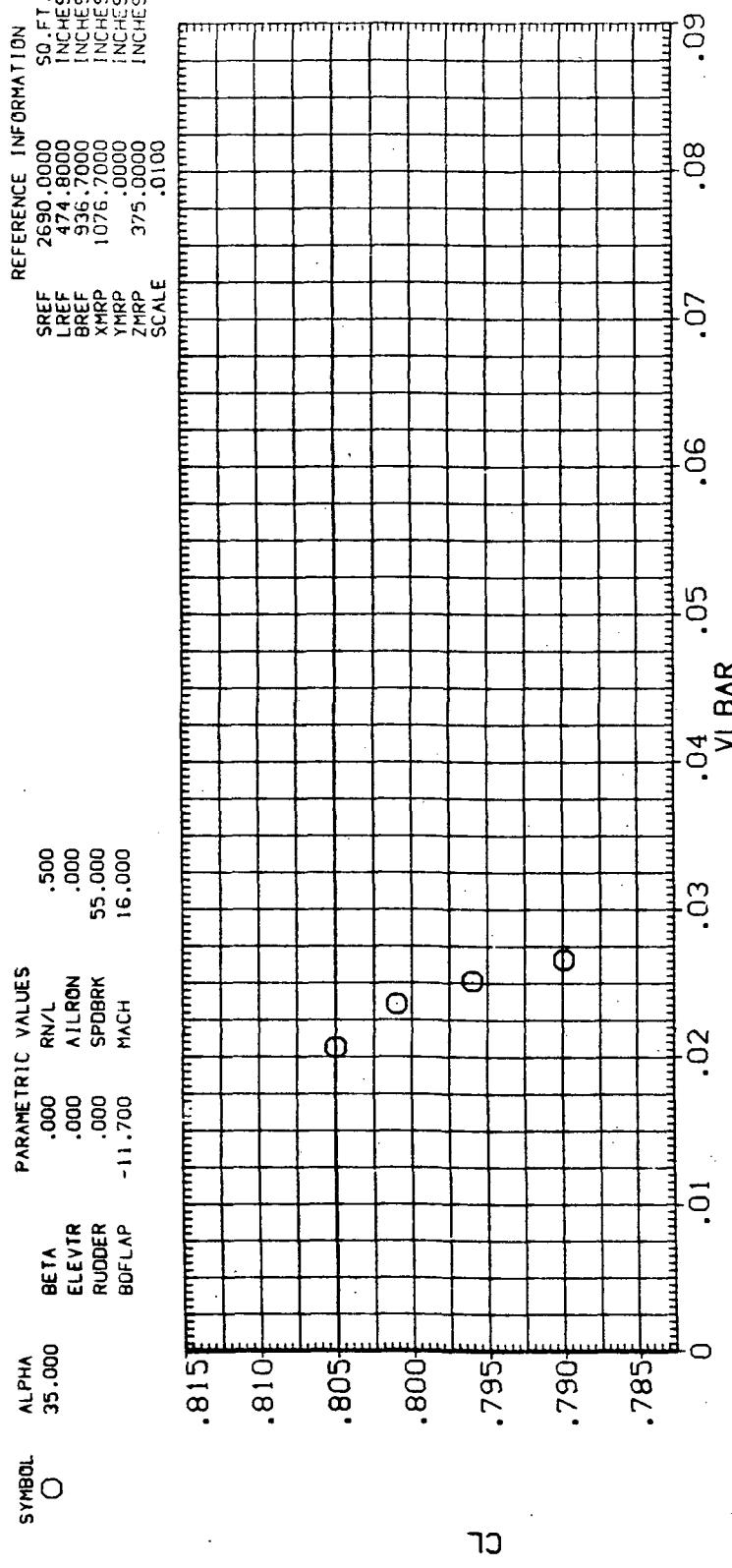
XCP/L



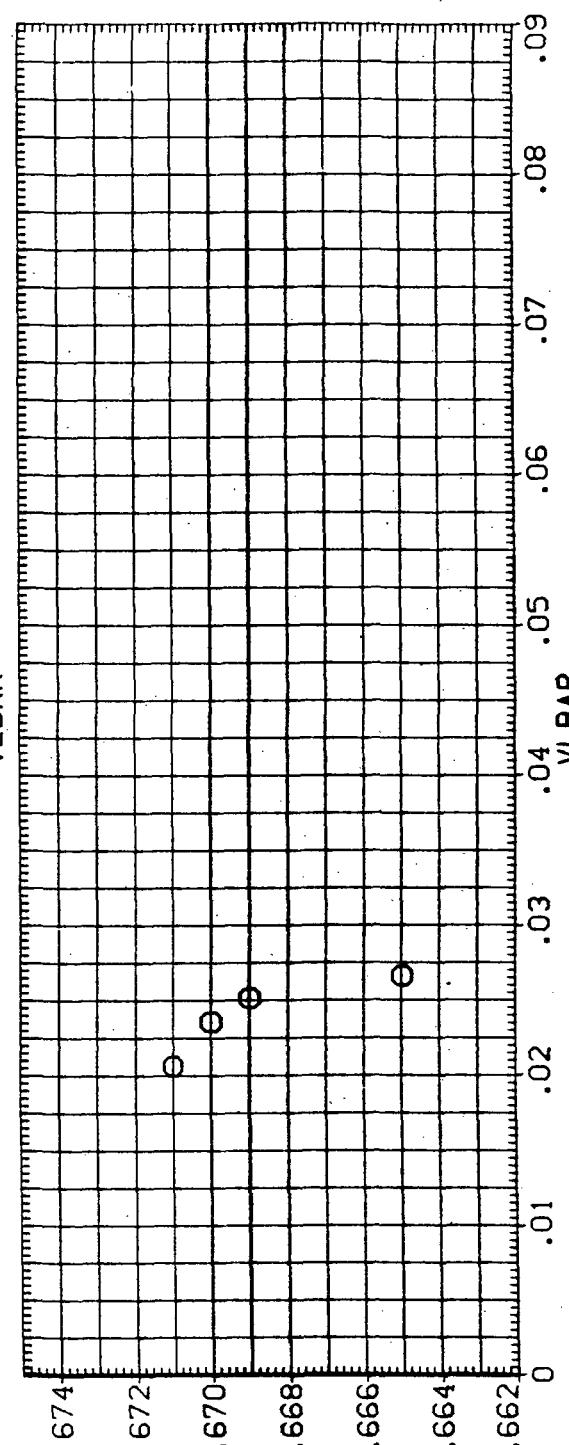
CA

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

AEDC489(0A81) B26C9F7M7N28 W116E26 V8R5 INVERTED (FT0016)



CL



CL

FIGURE 5. HYPERSONIC CHARACTERISTICS OF SPACE SHUTTLE ORBITER 140 A/B

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APPENDIX
TABULATED SOURCE DATA

DATE 04 NOV 75

TACULATED SOURCE DATA, AEDC VA 489, OABI

AEDC VA489(0A-81), (B2BC9F7M7N28) (W116E26) V8R5)

PAGE 2

REFERENCE DATA

SREF = 2690.0000 SQ.FT. XMRP = 1076.7000 INCHES
 LREF = 474.8000 INCHES YMRP = .0000 INCHES
 BREF = 936.7000 INCHES ZMRP = 375.0000 INCHES
 SCALE = .0100

MACH	ALPHA	TIME	PO	P	Q(PSI)	TO	RE-L	CPB	V-INF	VLBAR
16.000	20.000	60.0000011697.00000	.00998	1.87000	4041.00000	1.05000	-	.01450	.01284	
15.700	20.000	65.0000011035.00000	.01010	1.74000	4303.00000	.88000	-.001180	.01570	.01364	
15.700	20.000	70.0000010470.00000	.00917	1.66000	4658.00000	.72000	-.001178	.01720	.01439	
15.400	20.000	80.00000 9500.00000	.00958	1.58000	4695.00000	.69000	-.001175	.01720	.01489	
15.000	20.000	100.00000 7771.00000	.00973	1.45000	4204.00000	.69000	-.001169	.01680	.01477	
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	
	RUN NO.	4/ 0	RNL =	.87	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	3/ 0	RNL =	1.02	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	2/ 0	RNL =	.97	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	1/ 0	RNL =	.94	GRADIENT INTERVAL =	-5.00/	5.00			
MACH	ALPHA	TIME	PO	P	Q(PSI)	TO	RE-L	CPB	V-INF	VLBAR
16.400	25.000	70.0000010374.00000	.00796	1.46000	3599.00000	1.09000	-.00022	.01470	.01314	
16.400	25.000	75.0000010003.00000	.00765	1.43000	3526.00000	1.12000	-.00023	.01450	.01301	
16.400	25.000	80.00000 9557.00000	.00746	1.41000	3397.00000	1.09000	-.00025	.01470	.01325	
16.000	25.000	90.00000 8739.00000	.00734	1.34000	3860.00000	.86000	-.00025	.01610	.01431	
15.600	25.000	100.00000 7984.00000	.00740	1.23000	4433.00000	.60000	-.00026	.01870	.01631	
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	
	RUN NO.	4/ 0	RNL =	.87	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	3/ 0	RNL =	1.02	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	2/ 0	RNL =	.97	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	1/ 0	RNL =	.94	GRADIENT INTERVAL =	-5.00/	5.00			
MACH	ALPHA	TIME	PO	P	Q(PSI)	TO	RE-L	CPB	V-INF	VLBAR
16.000	30.000	60.0000011321.00000	.01010	1.80000	3710.00000	1.18000	-.00160	.01370	.01227	
15.900	30.000	70.0000010185.00000	.00861	1.52000	4160.00000	.90000	.00187	.01560	.01374	
16.100	30.000	80.00000 9293.00000	.00729	1.32000	4106.00000	.83000	.00215	.01650	.01452	
16.400	30.000	90.00000 8550.00000	.00646	1.20000	3460.00000	.82000	.00243	.01690	.01526	
16.400	30.000	100.00000 7806.00000	.00579	1.08000	3568.00000	.78000	.00274	.01730	.01558	
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	
	RUN NO.	4/ 0	RNL =	.87	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	3/ 0	RNL =	1.02	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	2/ 0	RNL =	.97	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	1/ 0	RNL =	.94	GRADIENT INTERVAL =	-5.00/	5.00			
MACH	ALPHA	TIME	PO	P	Q(PSI)	TO	RE-L	CPB	V-INF	VLBAR
16.100	35.000	60.0000011397.00000	.00963	1.80000	3744.00000	1.20000	.00390	.01370	.01223	
15.700	35.000	65.0000010666.00000	.01030	1.75000	4036.00000	.98000	.00370	.01480	.01304	
15.600	35.000	70.0000010211.00000	.00968	1.68000	4391.00000	.80000	.00358	.01630	.01417	
15.500	35.000	80.00000 9273.00000	.00870	1.53000	4481.00000	.69000	.00340	.01750	.01520	
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	
	RUN NO.	4/ 0	RNL =	.87	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	3/ 0	RNL =	1.02	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	2/ 0	RNL =	.97	GRADIENT INTERVAL =	-5.00/	5.00			
	RUN NO.	1/ 0	RNL =	.94	GRADIENT INTERVAL =	-5.00/	5.00			

PARAMETRIC DATA

(ST001) (04 NOV 75)

DATE 04 NOV 75

 TABULATED SOURCE DATA, AEDC VA 489, OAB!
 AEDC VA489(OA-81), (B26C9F7M7N28) (W116E26) (V8R5)

 PAGE 3
 (RT002) (04 NOV 75)

REFERENCE DATA

PARAMETRIC DATA									
SREF	2690.0000	SQ.FT.	XMRP	1076.7000	INCHES	BETA	RN/L	CAM	CAFM
LREF	474.8000	INCHES	YMRP	.0000	INCHES	ELEVTR	AIRRON	CAM	CAFM
BREF	936.7000	INCHES	ZMRP	375.0000	INCHES	RUDDER	SPDBRK	CAM	CAFM
SCALE	.0100					BDFLAP	-11.700	.000	.000
MACH	ALPHA	TIME	CN	CLM	XCP/L	.24	GRADIENT INTERVAL = -5.00 / 5.00		
20.300	20.000	60.00000	.38400	.00869	.64167	.33100	L/D	CAE	CAF
	20.200	70.00000	.38400	.00869	.64167	.32900	.21200	.03837	.08631
	19.300	20.000	.38400	.01080	.63965	.33200	.21800	.09200	.09250
	19.100	20.000	.38300	.01070	.63972	.32800	.21000	.03492	.08360
	19.000	20.000	.38200	.01180	.63963	.32600	.22000	.09150	.09200
	19.200	20.000	.38000	.01170	.63867	.32500	.21800	.03848	.09259
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.03750	.09417
								.00000	.00000
MACH	ALPHA	TIME	CN	CLM	XCP/L	.25	GRADIENT INTERVAL = -5.00 / 5.00		
19.900	25.000	90.00000	.52200	.01750	.63766	.43800	L/D	CAE	CAF
	19.600	25.000	.51800	.01590	.63870	.43200	.29800	.03227	.08224
	19.600	25.000	.51800	.01590	.63870	.43100	.30100	.08740	.08764
	19.500	25.000	.51300	.01440	.63967	.42500	.30100	.03536	.09114
	19.700	25.000	.51100	.01430	.63970	.42400	.30000	.03583	.09280
	19.800	25.000	.50600	.01150	.64164	.42000	.41000	.03465	.09243
	19.600	25.000	.50300	.01140	.64166	.41700	.29800	.03393	.09222
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.03377	.09220
								.00000	.00000
MACH	ALPHA	TIME	CN	CLM	XCP/L	.25	GRADIENT INTERVAL = -5.00 / 5.00		
19.900	30.000	80.00000	.72500	.02230	.63868	.58300	L/D	CAE	CAF
	19.800	30.000	.72300	.01640	.64165	.57800	.44000	.03334	.08981
	19.800	30.000	.70900	.01410	.64268	.56200	.44500	.03619	.09610
	19.500	30.000	.70200	.01020	.64465	.55700	.44000	.03937	.10400
	19.400	30.000	.69400	.00816	.64567	.55100	.43300	.03716	.10200
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.03431	.09880
								.00000	.00000
MACH	ALPHA	TIME	CN	CLM	XCP/L	.23	GRADIENT INTERVAL = -5.00 / 5.00		
20.200	35.000	90.00000	.90800	.02800	.63865	.68600	L/D	CAE	CAF
	19.900	35.000	.90700	.02550	.63965	.68300	.60400	.04294	.10129
	19.200	35.000	.90400	.02290	.64068	.67200	.61600	.04451	.10525
	18.800	35.000	.90400	.02540	.63966	.67600	.61000	.05179	.11823
	18.900	35.000	.89500	.02510	.63968	.66800	.60700	.04717	.11700
	18.800	35.000	.88700	.02730	.63867	.66000	.60400	.04743	.11219
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.04747	.11414
								.00000	.00000

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OAB1

PAGE 4 NOV 14 1975

AEFDC YAK89(0A-01) (026C9E7M7N28) (W116F26) (Y885)

REFERENCE DATA

	SREF	LREF	BREF	SCALE	SO.FT.	XMRP	YMRP	ZMRP	INCHES	INCHES	INCHES	INCHES
	2690.0000	474.8000	936.7000	.0100	1.076.	7000			INCHES	INCHES	INCHES	INCHES
RUN NO.	8 / 0											
MACH	ALPHA	TIME	PO	P		Q(PSI)	T0		RE-L		CPB	V- INF
20.300	20.000	60.0000012189	.00000	.00185	.54000	4610.0000		.37000		.00330	.03130	.02694
20.200	20.000	70.0000011672	.00000	.00176	.53000	4934.0000		.28000		.00325	.03560	.03050
19.300	20.000	80.0000011458	.00000	.00198	.51000	6269.0000		.24000		.00325	.03700	.03023
18.100	20.000	100.0000010578	.00000	.00185	.48000	6634.0000		.18000		.00320	.04190	.03422
19.000	20.000	120.0000009912	.00000	.00175	.44000	6837.0000		.15000		.00315	.04570	.03708
19.200	20.000	140.0000009427	.00000	.00154	.40000	6747.0000		.13000		.00300	.04990	.04038
GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000		.00000		.00000	.00000	.00000

PARAMETRIC DATA

RUN NO.: 6 / 0 RN/L = .25 GRADIENT INTERVAL = -5.00 / 5.00

23 GRADIENT INTERVAL = -5.00 / 5.00

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OAB1

AEDC VA489(0A-81) : (B26C9F7M7N28) (W116E26) (V8R5)

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES
BREF	=	935.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

PARAMETRIC DATA

BETA	.000	RNL	1.100
ELEVTR	.000	AIRRON	.000
RUDDER	.000	SPDBRK	55.000
BDFLAP	-11.700		

	V-LBAR
V-INF	.01179
.01320	.01179
.01500	.01316
.01660	.01445
.01680	.01482
.01740	.01533
.00000	.00000

PAGE 6
04 NOV 75)

15T0003 (04 NOV 75)

ANALYTIC DATA

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, CAB1

AEDC VA489(OA-81), (B26C9F7M7N2B) (W116E26) (V8R5)

REFERENCE DATA
(RT0004) (04 NOV 75)

REFERENCE DATA

SREF	2690.0000	SQ.FT.	XMRP	= 1076.7000 INCHES		BETA	= .000	RNL	= 1.100
LREF	474.8000	INCHES	YMRP	= .0000 INCHES		ELEVTR	= .000	AIRRON	= .000
BREF	936.7000	INCHES	ZMRP	= 375.0000 INCHES		RUDDER	= .000	SPDBRK	= 55.000
SCALE	.0100					BOFLAP	= .000		

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	CAF	VLBAR
15.800	20.000	50.00000	.43300	-.00347	.65295	.38200	.21500	1.7000	.07200	.07160	.01083
15.800	20.000	60.00000	.44300	-.00119	.65099	.39000	.22300	1.75000	.07600	.07590	.01342
16.300	20.000	70.00000	.44700	-.00136	.65112	.39300	.22600	1.74000	.07800	.07790	.01440
16.200	20.000	80.00000	.45100	-.00161	.65050	.39700	.22900	1.73000	.08000	.07900	.01439
16.000	20.000	90.00000	.45300	-.00065	.65052	.39800	.23200	1.71000	.08200	.08190	.01499
					.000000	.000000	.000000		.000000	.000000	.00000
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	CAF	VLBAR
16.300	25.000	60.00000	.56200	-.01360	.65890	.48100	.29900	1.61000	.06800	.06800	.01227
16.300	25.000	65.00000	.56500	-.01240	.65808	.48300	.30100	1.60000	.06900	.06910	.01342
16.400	25.000	70.00000	.56600	-.01280	.65832	.48300	.30300	1.60000	.07000	.07010	.01415
16.200	25.000	80.00000	.57400	-.01140	.65731	.49000	.30800	1.59000	.07200	.07210	.01458
16.400	25.000	90.00000	.58000	-.01030	.65653	.49400	.31200	1.58000	.07400	.07410	.01519
					.000000	.000000	.000000		.000000	.000000	.00000
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	CAF	VLBAR
16.200	30.000	50.00000	.80700	-.00811	.65370	.66200	.46600	1.42000	.07200	.07240	.01174
16.100	30.000	55.00000	.81200	-.01060	.65480	.66500	.47200	1.41000	.07600	.07640	.01279
16.100	30.000	60.00000	.81400	-.01067	.65437	.66500	.47500	1.40000	.07800	.07850	.01395
15.800	30.000	70.00000	.82100	-.01050	.65471	.67100	.48000	1.40000	.08000	.08050	.01538
15.900	30.000	80.00000	.82900	-.01010	.65448	.67700	.48600	1.39000	.08200	.08250	.01572
16.000	30.000	90.00000	.83500	-.01040	.65458	.68100	.49100	1.39000	.08400	.08460	.01623
					.000000	.000000	.000000		.000000	.000000	.00000
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	CAF	VLBAR
16.200	35.000	55.00000	.1.03000	-.03180	.66136	.80100	.65300	1.23000	.07650	.07570	.01237
16.400	35.000	60.00000	.1.03100	-.03100	.66108	.80200	.65500	1.22000	.07750	.07660	.01360
16.500	35.000	70.00000	.1.04000	-.03070	.66086	.80400	.65800	1.22000	.07900	.07810	.01445
16.700	35.000	80.00000	.1.04000	-.03070	.66086	.80500	.66000	1.22000	.08000	.07910	.01423
16.700	35.000	90.00000	.1.04000	-.02990	.66058	.80600	.66200	1.22000	.08050	.07960	.01424
					.000000	.000000	.000000		.000000	.000000	.00000

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, 0AB1

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AEDC VA469 (OA-81), (B26C 9F7M7N28) (W116E26) (VBR5) (ST0004) (04 NOV 75)

REFERENCE DATA

SREF	=	2690.000	SQ.FT.	XMRP	=	1076.7000	INCHES	BETA	=	.000	RNL	=	1.100
LREF	=	474.800	INCHES	YMRP	=	.0000	INCHES	ELEVTR	=	.000	ALTRON	=	.000
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES	RUDDER	=	.000	SPDBRK	=	55.000
SCAF	=	0.100						BDFLAP	=	.000			

PARAMETRIC DATA

RUN NO.	12/ 0	RNL =	1.07	GRADIENT INTERVAL =	-5.00/	5.00
MACH	TIME	PO	P	Q(PSL)	TO	RE-L
ALPHA	55.0000012566.00000	.01010	1.89000	3824.	00000	1.18000
.200	60.0000012332.00000	.00895	1.68000	4084.	00000	.98000
.400	70.0000011047.00000	.00746	1.43000	4094.	00000	.88000
.500	80.0000010165.00000	.00662	1.32000	3681.	00000	.96000
.700	90.000009344.00000	.00636	1.24000	3522.	00000	.97000
.700	GRADIENT .00000	.00000	.00000	.00000	.00000	.00000

	AACH	ALPHA	TIME	P0	P	Q(PSI)	T0	RE-L	CPB	V- INF	VLBAR
.200	55.00000	12566.00000	.01010	1.89000	3824.00000		1.18000		.00545	.01390	.01237
.400	60.00000	12332.00000	.00895	1.68000	4084.00000		.98000		.00550	.01540	.01360
.500	70.00000	11047.00000	.00746	1.43000	4094.00000		.88000		.00563	.01640	.01445
.700	80.00000	10165.00000	.00652	1.32000	3681.00000		.96000		.00575	.01600	.01423
35.000	90.00000	9344.00000	.00636	1.24000	3522.00000		.97000		.00590	.01580	.01424
GRADIENT	.00000	.00000	.00000	.00000	.00000		.00000		.00000	.00000	.00000

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OA81

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AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5)

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

REFERENCE DATA								PARAMETRIC DATA					
								(RT0005) (04 NOV 75)					
								(RT0005) (04 NOV 75)					
RUN NO.	TIME	CN	CLM	XCP/L	CL	CD	L/D	BETA	ELEVTR	RUDDER	AILRON	SPDBRK	CAF M
19/ 0	RNL =	.23	GRADIENT INTERVAL =	-5.00/	5.00			CAE	CAM	CAF M			
MACH	ALPHA												
20.000	20.000	90.00000	.41400	-.01310	.35600	.23300	.53000	.04106	.09670	.09717			
20.000	20.000	100.00000	.41400	-.01200	.66067	.23000	.55000	.03828	.09330	.09377			
20.000	20.000	110.00000	.41400	-.00975	.65987	.23200	.54000	.03899	.09540	.09586			
20.300	20.000	120.00000	.41400	-.00975	.65867	.23400	.49000	.04181	.10100	.10146			
19.800	20.000	130.00000	.41400	-.01090	.65969	.23700	.45000	.04550	.10600	.10846			
19.700	20.000	140.00000	.41400	-.01090	.65969	.24300	.43000	.04655	.11000	.11045			
GRADIENT		.000000		.000000	.000000		.000000						
RUN NO.	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE	CAM	CAF M			
18/ 0	RNL =	.22	GRADIENT INTERVAL =	-5.00/	5.00								
MACH	ALPHA												
20.000	25.000	84.00000	.60000	-.01900	.66165	.49900	.35000	.43000	.03987	.10500	.10584		
20.000	25.000	90.00000	.59800	-.01900	.66169	.49700	.35000	.42000	.04027	.10600	.10681		
19.700	25.000	100.00000	.59600	-.01730	.66068	.49400	.35000	.41000	.04095	.10800	.10870		
20.000	25.000	110.00000	.59000	-.01550	.65967	.48800	.34800	.40000	.04080	.10800	.10867		
19.500	25.000	120.00000	.58800	-.01380	.65864	.48600	.34900	.39000	.04158	.11000	.11066		
19.700	25.000	140.00000	.57700	-.01200	.65765	.47400	.35000	.35000	.04441	.11600	.11662		
GRADIENT		.000000		.000000	.000000		.000000						
RUN NO.	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE	CAM	CAF M			
17/ 0	RNL =	.20	GRADIENT INTERVAL =	-5.00/	5.00								
MACH	ALPHA												
19.900	30.000	90.00000	.80700	-.02120	.65967	.64500	.49000	.30000	.04068	.10800	.10887		
19.900	30.000	100.00000	.80700	-.02120	.65967	.64200	.50200	.28000	.04278	.11300	.11384		
19.500	30.000	110.00000	.79700	-.01660	.65766	.63100	.50100	.26000	.04450	.11700	.11781		
19.300	30.000	120.00000	.78500	-.01640	.65769	.61900	.49800	.24000	.04641	.12100	.12179		
19.600	30.000	140.00000	.77400	-.01400	.65666	.60700	.49600	.23000	.04875	.12500	.12576		
19.200	30.000	160.00000	.76200	-.01380	.65666	.59600	.49200	.21000	.05035	.12700	.12774		
GRADIENT		.000000		.000000	.000000		.000000						
RUN NO.	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE	CAM	CAF M			
16/ 0	RNL =	.24	GRADIENT INTERVAL =	-5.00/	5.00								
MACH	ALPHA												
20.200	35.000	100.00000	1.02000	-.04360	.66573	.77000	.68400	.13000	.05025	.11700	.11809		
19.900	35.000	110.00000	1.02000	-.04360	.66573	.76700	.68900	.11000	.05286	.12300	.12409		
19.700	35.000	120.00000	1.02000	-.03800	.66371	.76300	.69300	.10000	.05457	.12900	.13009		
19.700	35.000	130.00000	1.02000	-.03790	.66367	.75800	.69700	.09000	.05808	.13500	.13609		
GRADIENT		.000000		.000000	.000000		.000000						

MEDEC VAI89(0A-81). (B26C9F7M7N28)(W16E26) (V8R5)

S100005) (04 NOV 75)

REFERENCE DATA

SREF		2690 .0000	SQ.FT.	XMRP	1076 .7000	INCHES	BETA	.000	RN/L	.250
LREF		474 .8000	INCHES	YMRP	.0000	INCHES	ELEVTR	.000	AIRRON	.000
BREF		936 .7000	INCHES	ZMRP	375 .0000	INCHES	RUDDER	.000	SPDBRK	55 .0000
SCALE		.0100					BDFLAP	.000		
GRADIENT										
RUN NO.		19 / 0		RN/L			GRADIENT INTERVAL =	-5.00/	5.00	
TIME		P0		P			Q (PSI)	T0	RE-L	CPB
MACH	ALPHA									V-1NF
20.000	20.000	90.0000011471 .00000		.00175			.47000	.5382 .00000	.26000	.03630
20.000	20.000	100.0000010685 .00000		.00160			.45000	.5586 .00000	.24000	.03840
20.000	20.000	110.0000010703 .00000		.00148			.44000	.5994 .00000	.22000	.02999
20.300	20.000	120.0000010306 .00000		.00139			.42000	.5440 .00000	.20000	.0298
19.800	20.000	130.0000010116 .00000		.00151			.41000	.5929 .00000	.19000	.0295
19.700	20.000	140.000020 9789 .00000		.00152			.40000	.5975 .00000	.18000	.0295
GRADIENT		.000000		.000000			.00000	.000000	.00000	.00000

MACH	ALPHA	TIME	P0	P	Q (PSI)	T0	RE-L	CPB
19.900		90.00000	11579.00000	.00176	.45000	5659.00000	.24000	.00560
19.900		100.00000	1066.00000	.00164	.44000	5843.00000	.20000	.00540
19.900		110.00000	10877.00000	.00167	.43000	5654.00000	.18000	.00525
19.500		120.00000	10552.00000	.00163	.42000	7081.00000	.17000	.00510
19.300		130.00000	10339.00000	.00143	.40000	6757.00000	.15000	.00490
19.600		140.00000	9269.00000	.00149	.39000	6924.00000	.15000	.00480
19.200		150.00000	9269.00000	.00152	.39000	6924.00000	.15000	.00480

PARAMETRIC DATA

BETA =	.000	RNL =	.250
ELEVTR =	.000	AIRRON =	.000
RUDDER =	.000	SPDBRK =	55.000
BDFLAP =	.000		
			5.00
/			
CPB	V- INF	VLR	C-FCT
.00301	.03630	.03090	1.34000
.00300	.03840	.03197	1.42000
.00299	.04010	.03301	1.43000
.00298	.04210	.03572	1.40000
.00295	.04250	.03523	1.36000
.00291	.04380	.03595	1.35000
.00000	.00000	.00000	0.00000

	C-FCTR	VLBAR	INF
	1.60000	.03038	33600
	1.60000	.03257	33870
	1.61000	.03364	+100
	1.62000	.03679	+460
	1.62000	.03685	+560
	1.59000	.03913	+760
	0.00000	0.00000	0.00000

	C-FCTR	VLBAR	VINF
	1.62000	.03173	3760
	1.61000	.03458	+130
	1.60000	.03491	+270
	1.58000	.03515	+380
	1.54000	.03837	+650
	1.50000	.03741	+690
	1.42000	.03741	+690

	C-FCTR
VLBLAR	.03085
1INF	.03630
3810	.03191
41060	.03447
41060	.03605
00000	.00000
	.130000
	.130000
	.130000
	.130000
	.000000

DATE 04 NOV 75

TABLE II. STATED SOURCE DATA.

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DATA

REF	2690.0000	SO.FT.	XMRP	1076.7000	INCHES	RN/L	.000	RN/L	.250
LREF	474.8000	INCHES	YMRP	.0000	INCHES	ELEVTR	.000	AIRDN	.000
BREF	936.7000	INCHES	ZMRP	375.0000	INCHES	RUDDER	.000	SPDBRK	.000
SCALE	.0100					BDFLAP	.000		.000
MACH	ALPHA	TIME	CN	.20	GRADIENT INTERVAL	-5.00 /	5.00	CAE	CAM
20.200	25.000	80.00000	CLM	.65797	XCP/L	CD	L/D	.03565	.09370
20.400	25.000	90.00000	.62300	-.01350	.52500	.34900	.1.50000	.03630	.09530
19.400	25.000	100.00000	.61600	-.01230	.65735	.34700	1.49000	.03713	.09740
19.400	25.000	110.00000	.60500	-.01180	.65718	.34500	1.47000	.03815	.10000
19.900	25.000	120.00000	.59700	-.01130	.65697	.34300	1.45000	.03915	.10300
			.58600	-.01030	.65645	.34200	1.43000		.10350
			.58000	-.00900	.65600	.34100	1.41000		
			.57400	-.00750	.65557	.34000	1.39000		
			.56800	-.00600	.65514	.33900	1.37000		
			.56200	-.00450	.65471	.33800	1.35000		
			.55600	-.00300	.65428	.33700	1.33000		
			.55000	-.00150	.65385	.33600	1.31000		
			.54400	.00000	.65342	.33500	1.29000		
			.53800	.00150	.65299	.33400	1.27000		
			.53200	.00300	.65256	.33300	1.25000		
			.52600	.00450	.65213	.33200	1.23000		
			.52000	.00600	.65170	.33100	1.21000		
			.51400	.00750	.65127	.33000	1.19000		
			.50800	.00900	.65084	.32900	1.17000		
			.50200	.01050	.65041	.32800	1.15000		
			.49600	.01200	.64998	.32700	1.13000		
			.49000	.01350	.64955	.32600	1.11000		
			.48400	.01500	.64912	.32500	1.09000		
			.47800	.01650	.64869	.32400	1.07000		
			.47200	.01800	.64826	.32300	1.05000		
			.46600	.01950	.64783	.32200	1.03000		
			.46000	.02100	.64740	.32100	1.01000		
			.45400	.02250	.64697	.32000	0.99000		
			.44800	.02400	.64654	.31900	0.97000		
			.44200	.02550	.64611	.31800	0.95000		
			.43600	.02700	.64568	.31700	0.93000		
			.43000	.02850	.64525	.31600	0.91000		
			.42400	.03000	.64482	.31500	0.89000		
			.41800	.03150	.64439	.31400	0.87000		
			.41200	.03300	.64396	.31300	0.85000		
			.40600	.03450	.64353	.31200	0.83000		
			.40000	.03600	.64310	.31100	0.81000		
			.39400	.03750	.64267	.31000	0.79000		
			.38800	.03900	.64224	.30900	0.77000		
			.38200	.04050	.64181	.30800	0.75000		
			.37600	.04200	.64138	.30700	0.73000		
			.37000	.04350	.64095	.30600	0.71000		
			.36400	.04500	.64052	.30500	0.69000		
			.35800	.04650	.64009	.30400	0.67000		
			.35200	.04800	.63966	.30300	0.65000		
			.34600	.04950	.63923	.30200	0.63000		
			.34000	.05100	.63880	.30100	0.61000		
			.33400	.05250	.63837	.30000	0.59000		
			.32800	.05400	.63794				
			.32200	.05550	.63751				
			.31600	.05700	.63708				
			.31000	.05850	.63665				
			.30400	.06000	.63622				
			.29800	.06150	.63579				
			.29200	.06300	.63536				
			.28600	.06450	.63493				
			.28000	.06600	.63450				
			.27400	.06750	.63407				
			.26800	.06900	.63364				
			.26200	.07050	.63321				
			.25600	.07200	.63278				
			.25000	.07350	.63235				
			.24400	.07500	.63192				
			.23800	.07650	.63149				
			.23200	.07800	.63106				
			.22600	.07950	.63063				
			.22000	.08100	.63020				
			.21400	.08250	.62977				
			.20800	.08400	.62934				
			.20200	.08550	.62891				
			.19600	.08700	.62848				
			.19000	.08850	.62805				
			.18400	.09000	.62762				
			.17800	.09150	.62719				
			.17200	.09300	.62676				
			.16600	.09450	.62633				
			.16000	.09600	.62590				
			.15400	.09750	.62547				
			.14800	.09900	.62504				
			.14200	.10050	.62461				
			.13600	.10200	.62418				
			.13000	.10350	.62375				
			.12400	.10500	.62332				
			.11800	.10650	.62289				
			.11200	.10800	.62246				
			.10600	.10950	.62203				
			.10000	.11100	.62160				
			.94000	.11250	.62117				
			.88000	.11400	.62074				
			.82000	.11550	.62031				
			.76000	.11700	.61988				
			.70000	.11850	.61945				
			.64000	.12000	.61902				
			.58000	.12150	.61859				
			.52000	.12300	.61816				
			.46000	.12450	.61773				
			.40000	.12600	.61730				
			.34000	.12750	.61687				
			.28000	.12900	.61644				
			.22000	.13050	.61601				
			.16000	.13200	.61558				
			.10000	.13350	.61515				
			.04000	.13500	.61472				
			.00000	.13600	.61430				

RUN NO.	TIME	RNL = .21 / 0	GRADIENT INTERVAL = -5.0
ALPHA	CN	CLM	XCP/L
35.000	.98000	-.03720	.66397
35.000	.98400	-.03770	.66397
35.000	1.00000	-.03740	.66376
35.000	.99800	-.03360	.66239
35.000	.99800	-.03080	.66147
35.000	.99800	-.03080	.66147

AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5)

REFERENCE DATA

ESTATE (04 NOV 75)

PARAMETRIC DATA

R/N/L		.250	C-FC
.000	AIRRON	.000	1.6000
.000	SPDBRK	55.000	
.000			
	VLBAR		
	.03304		
	.03890		
	.04230		
	.04300		
	.04540		
	.04910		
	.04134		
	.000000		
V-1NF			
.03890			
.04230			
.04300			
.04540			
.04910			
.04134			
.000000			

ESTATE (04 NOV 75)

PARAMETRIC DATA

R/N/L		.250	C-FC
.000	AIRRON	.000	1.6000
.000	SPDBRK	.000	1.6000
.000		55.000	1.6000
			.000000
V-1NF	VLBAR		
.03890	.03304		
.04230	.03610		
.04300	.03475		
.04540	.03639		
.04910	.04134		
.000000	.000000		

ESTATE (04 NOV 75)

PARAMETRIC DATA

R/N/L		.250	C-FC
.000	AIRRON	.000	1.6000
.000	SPDBRK	.000	1.6000
.000		55.000	1.6000
			.000000
V-1NF	VLBAR		
.03890	.03304		
.04230	.03610		
.04300	.03475		
.04540	.03639		
.04910	.04134		
.000000	.000000		

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TABULATED SOURCE DATA, AEDC VA 489, OA81

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AEDC VA489(OA-81), (B26C9F7M7N28) (W116E26) (V8R5)

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

MACH	ALPHA	TIME	PO	P	Q(P51)	T0	RE-L	CPB	C-FCTR
19.900	35.000	90.0000011457.00000	.00176	.47000	5550.0000	.30000	.00880	.03400	.25000
20.300	35.000	100.0000011457.00000	.00154	.46000	5555.0000	.25000	.00880	.03980	.000
20.100	35.000	110.0000010830.00000	.00159	.44000	5343.0000	.21000	.00880	.04060	.03461
19.700	35.000	120.0000010767.00000	.00158	.43000	6296.0000	.20000	.00880	.04130	.03382
18.300	35.000	130.0000010328.00000	.00180	.42000	9535.0000	.17000	.00880	.04150	.03152
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000

AEDC VA489(OA-81), (B26C9F7M7N28) (W116E26) (V8R5)

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.400	20.000	55.00000	.39900	-.01100	.66014	.35100	.20500	1.70000	.07330	.01688
15.400	20.000	60.00000	.41500	-.00843	.65747	.36300	.21600	1.68000	.07910	.01927
15.600	20.000	70.00000	.43500	-.00447	.65378	.37800	.23100	1.64000	.08750	.02185
15.700	20.000	80.00000	.44100	-.00572	.65477	.38300	.23600	1.62000	.09070	.02286
15.600	20.000	90.00000	.43600	-.00878	.65741	.37900	.22300	1.62000	.08960	.02310
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.800	25.000	50.00000	.57100	-.01650	.66063	.48500	.31100	1.56000	.07650	.01659
15.800	25.000	55.00000	.58300	-.01370	.65865	.49400	.31900	1.55000	.08010	.01785
15.600	25.000	60.00000	.59300	-.01070	.65684	.50200	.32700	1.53000	.08440	.01926
15.900	25.000	70.00000	.60500	-.00747	.65454	.51100	.33700	1.52000	.08910	.02128
15.900	25.000	80.00000	.60900	-.00709	.65428	.51300	.34000	1.51000	.09070	.02142
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.00090	.02159

PARAMETRIC DATA

(ST0006) (04 NOV 75)

MACH	ALPHA	TIME	PO	P	Q(P51)	T0	RE-L	CPB	V- INF	VLBAR
19.900	35.000	90.0000011457.00000	.00176	.47000	5550.0000	.30000	.00880	.03400	.02848	.1.25000
20.300	35.000	100.0000011457.00000	.00154	.46000	5555.0000	.25000	.00880	.03980	.03317	.1.25000
20.100	35.000	110.0000010830.00000	.00159	.44000	5343.0000	.21000	.00880	.04060	.03461	.1.25000
19.700	35.000	120.0000010767.00000	.00158	.43000	6296.0000	.20000	.00880	.04130	.03382	.1.25000
18.300	35.000	130.0000010328.00000	.00180	.42000	9535.0000	.17000	.00880	.04150	.03152	.1.25000
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000

MACH	ALPHA	TIME	PO	P	Q(P51)	T0	RE-L	CPB	V- INF	VLBAR
19.900	35.000	90.0000011457.00000	.00176	.47000	5550.0000	.30000	.00880	.03400	.02848	.1.25000
20.300	35.000	100.0000011457.00000	.00154	.46000	5555.0000	.25000	.00880	.03980	.03317	.1.25000
20.100	35.000	110.0000010830.00000	.00159	.44000	5343.0000	.21000	.00880	.04060	.03461	.1.25000
19.700	35.000	120.0000010767.00000	.00158	.43000	6296.0000	.20000	.00880	.04130	.03382	.1.25000
18.300	35.000	130.0000010328.00000	.00180	.42000	9535.0000	.17000	.00880	.04150	.03152	.1.25000
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.400	20.000	55.00000	.39900	-.01100	.66014	.35100	.20500	1.70000	.07330	.01688
15.400	20.000	60.00000	.41500	-.00843	.65747	.36300	.21600	1.68000	.07910	.01927
15.600	20.000	70.00000	.43500	-.00447	.65378	.37800	.23100	1.64000	.08750	.02185
15.700	20.000	80.00000	.44100	-.00572	.65477	.38300	.23600	1.62000	.09070	.02286
15.600	20.000	90.00000	.43600	-.00878	.65741	.37900	.22300	1.62000	.08960	.02310
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.800	25.000	50.00000	.57100	-.01650	.66063	.48500	.31100	1.56000	.07650	.01659
15.800	25.000	55.00000	.58300	-.01370	.65865	.49400	.31900	1.55000	.08010	.01785
15.600	25.000	60.00000	.59300	-.01070	.65684	.50200	.32700	1.53000	.08440	.01926
15.900	25.000	70.00000	.60500	-.00747	.65454	.51100	.33700	1.52000	.08910	.02128
15.900	25.000	80.00000	.60900	-.00709	.65428	.51300	.34000	1.51000	.09070	.02142
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.00090	.02159

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.800	25.000	50.00000	.57100	-.01650	.66063	.48500	.31100	1.56000	.07650	.01659
15.800	25.000	55.00000	.58300	-.01370	.65865	.49400	.31900	1.55000	.08010	.01785
15.600	25.000	60.00000	.59300	-.01070	.65684	.50200	.32700	1.53000	.08440	.01926
15.900	25.000	70.00000	.60500	-.00747	.65454	.51100	.33700	1.52000	.08910	.02128
15.900	25.000	80.00000	.60900	-.00709	.65428	.51300	.34000	1.51000	.09070	.02142
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.00090	.02159

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.800	25.000	50.00000	.57100	-.01650	.66063	.48500	.31100	1.56000	.07650	.01659
15.800	25.000	55.00000	.58300	-.01370	.65865	.49400	.31900	1.55000	.08010	.01785
15.600	25.000	60.00000	.59300	-.01070	.65684	.50200	.32700	1.53000	.08440	.01926
15.900	25.000	70.00000	.60500	-.00747	.65454	.51100	.33700	1.52000	.08910	.02128
15.900	25.000	80.00000	.60900	-.00709	.65428	.51300	.34000	1.51000	.09070	.02142
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.00090	.02159

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.800	25.000	50.00000	.57100	-.01650	.66063	.48500	.31100	1.56000	.07650	.01659
15.800	25.000	55.00000	.58300	-.01370	.65865	.49400	.31900	1.55000	.08010	.01785
15.600	25.000	60.00000	.59300	-.01070	.65684	.50200	.32700	1.53000	.08440	.01926
15.900	25.000	70.00000	.60500	-.00747	.65454	.51100	.33700	1.52000	.08910	.02128
15.900	25.000	80.00000	.60900	-.00709	.65428	.51300	.34000	1.51000	.09070	.02142
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.00090	.02159

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.800	25.000	50.00000	.57100	-.01650	.66063	.48500	.31100	1.56000	.07650	.01659
15.800	25.000	55.00000	.58300	-.01370	.65865	.49400	.31900	1.55000	.08010	.01785
15.600	25.000	60.00000	.59300	-.01070	.65684	.50200	.32700	1.53000	.08440	.01926
15.900	25.000	70.00000	.60500	-.00747	.65454	.51100	.33700	1.52000	.08910	.02128
15.900	25.000	80.00000	.60900	-.00709	.65428	.51300	.34000	1.51000	.09070	.02142
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.00090	.02159

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.800	25.000	50.00000	.57100	-.01650	.66063	.48500	.31100	1.56000	.07650	.01659
15.800	25.000	55.00000	.58300	-.01370	.65865	.49400	.31900	1.55000	.08010	.01785
15.600	25.000	60.00000	.59300	-.01070	.65684	.50200	.32700	1.53000	.08440	.01926
15.900	25.000	70.00000	.60500	-.00747	.65454	.51100	.33700	1.52000	.08910	.02128
15.900	25.000	80.00000	.60900	-.00709	.65428	.51300	.34000	1.51000	.09070	.02142
GRADIENT		.000000	.000000	.000000	.000000	.000000	.000000	.000000	.00090	.02159

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	VLBAR
15.800	25.000	50.00000	.57100	-.01650	.66063	.4850				

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OA81

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AEDC VA489(OA-81), (B26C9F7M/N28) (W116E2S) (V8R5)

REFERENCE DATA

SREF	=	2690.0000 SQ.FT.	XMRP	=	1076.7000 INCHES
LREF	=	474.8000 INCHES	YMRP	=	.0000 INCHES
BREF	=	936.7000 INCHES	ZMRP	=	375.0000 INCHES
SCALE	=	.0100			

RUN NO.	23/ 0	RN/L	=	.48	GRADIENT INTERVAL = -5.00/ 5.00
MACH	ALPHA	TIME	CLM	CL	CD
15.500	30.000	55.00000	.79900	.66290	.47100
15.600	30.000	.80700	-.02800	.65600	.1.38000
15.900	30.000	.82400	-.02460	.66023	.47800
	GRADIENT	.00000	-.02290	.66700	.1.37000
			.00000	.00000	.49200
					.1.35000
					.00000

RUN NO.	22/ 0	RN/L	=	.44	GRADIENT INTERVAL = -5.00/ 5.00
MACH	ALPHA	TIME	CLM	CL	CD
15.900	35.000	55.00000	.96700	-.05310	.74700
15.800	35.000	60.00000	.98300	-.04860	.66819
15.800	35.000	70.00000	1.00000	-.04280	.66475
15.800	35.000	80.00000	1.02000	-.03890	.77000
16.000	35.000	90.00000	1.03000	-.03670	.66403
	GRADIENT	.00000	.00000	.00000	.78200
					.66311
					.79100
					.67100
					.1.18000
					.00000
					.00000

AEDC VA489(OA-81), (B26C9F7M7N28) (W116E2S) (V8R5)

REFERENCE DATA

SREF	=	2690.0000 SQ.FT.	XMRP	=	1076.7000 INCHES
LREF	=	474.8000 INCHES	YMRP	=	.0000 INCHES
BREF	=	936.7000 INCHES	ZMRP	=	375.0000 INCHES
SCALE	=	.0100			

RUN NO.	25/ 0	RN/L	=	.39	GRADIENT INTERVAL = -5.00/ 5.00
MACH	ALPHA	TIME	P	Q (PSI)	T ₀
15.400	20.000	55.00000	5908.00000	.00588	.95000
15.400	20.000	60.00000	5744.00000	.00530	.89000
15.600	20.000	70.00000	5144.00000	.00442	.75000
15.700	20.000	80.00000	4681.00000	.00379	.67000
15.600	20.000	90.00000	4219.00000	.00362	.62000
	GRADIENT	.00000	.00000	.00000	.4486.00000
					.00000
					.00000

(RT0007) (04 NOV 75)			
PARAMETRIC DATA			
BETA	=	.000	RN/L
ELEVTR	=	.000	AIRRON
RUDDER	=	.000	SPDBRK
BDFLAP	=	.000	

(ST0007) (04 NOV 75)			
PARAMETRIC DATA			
BETA	=	.000	V/LBAR
ELEVTR	=	.000	.01660
RUDDER	=	.000	.01876
BDFLAP	=	.000	.02185

(RT0007) (04 NOV 75)			
PARAMETRIC DATA			
BETA	=	.000	V/LBAR
ELEVTR	=	.000	.01682
RUDDER	=	.000	.01854
BDFLAP	=	.000	.02095

(ST0007) (04 NOV 75)			
PARAMETRIC DATA			
BETA	=	.000	V/LBAR
ELEVTR	=	.000	.01927
RUDDER	=	.000	.02510
BDFLAP	=	.000	.02286

(RT0007) (04 NOV 75)			
PARAMETRIC DATA			
BETA	=	.000	V/LBAR
ELEVTR	=	.000	.01910
RUDDER	=	.000	.02230
BDFLAP	=	.000	.02185

(ST0007) (04 NOV 75)			
PARAMETRIC DATA			
BETA	=	.000	V/LBAR
ELEVTR	=	.000	.02310
RUDDER	=	.000	.02660
BDFLAP	=	.000	.00000

DATE 04 NOV 75

AEDC Y4A489(0A-81) TABULATED SOURCE DATA, AEDC VA 499, 0A81 {B26C9F7M7N28} (W 116E26)

REFERENCE DATA

REFERENCE DATA

SREF	2690.0000	SQ.FT.	XMRP	-1076.7000	INCHES
LREF	474.8000	INCHES	YMRP	.0000	INCHES
BREF	936.7000	INCHES	ZMRP	375.0000	INCHES
SCALE	.0.20				

MACH	ALPHA	TIME	CN	RN/L =	.09	GRADIENT INTERVAL =	-5.00/	5.00	L/D	CAE	CAM	CAF M
9.500	25.000	80.00000	.60300	-	.00848	.65517	.49100	.37500	.31000	.05905	.13190	.13190
9.400	25.000	90.00000	.60100	-	.01020	.65615	.49000	.37100	.32000	.05692	.12887	.12887
9.500	25.000	100.00000	.59800	-	.01000	.65570	.48700	.37000	.32000	.05617	.12884	.12884
9.100	25.000	110.00000	.59300	-	.01140	.65707	.48100	.37100	.30000	.05719	.13280	.13281
8.500	25.000	120.00000	.58200	-	.01130	.65714	.46900	.37200	.26000	.05972	.13900	.13981
25.000	25.000	125.00000	.58100	-	.01130	.65714	.46900	.37200	.26000	.05972	.13900	.13981

RUN NO.	26 / 0	RN/L =	.09	GRADIENT INTERVAL =	-5.00/	5.00
MACH	ALPHA	TIME	CN	CLM	XCP/L	CD
9.200	35.000	80.00000	1.07000	.66059	.79200	.73700
9.300	35.000	90.00000	1.07000	.66008	.79300	.73100
9.200	35.000	100.00000	1.07000	.66045	.79100	.73100
8.900	35.000	120.00000	1.05000	.65922	.78300	.72100
9.200	35.000	140.00000	1.05000	.65999	.77700	.72100
7.700	35.000	160.00000	1.05000	.66034	.77300	.72300
GRADENT		0.00000	0.00000	0.00000	0.00000	0.00000
CAF						
					L/D	CAM
					1.08000	.14800
					.06290	.14300
					.06334	.14400
					.06020	.14000
					.06195	.14000
					.06369	.14800
					.00000	.00000

(RT0008) (04 NOV 75)
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PARAMETRIC DATA

BETA	.000	RNL	.070
ELEVTR	.000	AIRRON	.000
RUDDER	.000	SPDBRK	55.000
RFELAP	.000		

CAE	CAM	CAF	CAF M
.05059	.11000	.11062	.11062
.05284	.111700	.111762	.111762
.05496	.12500	.12562	.12562
.05530	.12800	.12862	.12862
	.00000	.00000	.00000

	CAF	CAM	CAF
.05905	.13100	.13190	.13190
.056892	.12800	.12887	.12887
.056117	.12800	.12884	.12884
.057119	.13200	.13281	.13281
.05972	.13900	.13981	.13981
	.12600	.12681	.12681

CAFM	1.3443
CAM	1.3300
CAE	1.3597
CAE	1.05505
CAE	0.05697
CAE	0.05959
CAE	0.06224
CAE	0.06610
CAE	0.00000

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OA81

AEDC VA489(0A-81), (826C9F7M7N28) (W116E26) (W8R5)

(S1008) (04 NOV 75)

REFERENCE DATA

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OA81

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AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (VBR5)

REFERENCE DATA

SREF	=	2690.0000 SQ.FT.	XMRP	=	1076.7000 INCHES
LREF	=	474.8000 INCHES	YMRP	=	.0000 INCHES
BREF	=	936.7000 INCHES	ZMRP	=	375.0000 INCHES
SCALE	=	.0100			

RUN NO.	30 / 0	RNL =	.36	GRADIENT INTERVAL =	-5.00 / 5.00
MACH	ALPHA	TIME	CN	CLM	XCP/L
15.900	35.000	70.00000	.95700	-.06350	.67442
15.900	35.000	80.00000	.96200	-.06610	.67528
16.000	35.000	90.00000	.96300	-.06750	.67579
16.100	35.000	100.00000	.96400	-.06900	.67634
	GRADIENT		.000000	.000000	.000000

AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (VBR5)

REFERENCE DATA

SREF	=	2690.0000 SQ.FT.	XMRP	=	1076.7000 INCHES
LREF	=	474.8000 INCHES	YMRP	=	.0000 INCHES
BREF	=	936.7000 INCHES	ZMRP	=	375.0000 INCHES
SCALE	=	.0100			

RUN NO.	30 / 0	RNL =	.36	GRADIENT INTERVAL =	-5.00 / 5.00
MACH	ALPHA	TIME	P0	P	Q(P\$1)
15.900	35.000	70.00000	5381.00000	.00436	.75000 4068.00000
15.900	35.000	80.00000	4872.00000	.00382	.67000 4131.00000
16.000	35.000	90.00000	4410.00000	.00323	.60000 4250.00000
16.100	35.000	100.00000	4028.00000	.00289	.53000 4127.00000
	GRADIENT		.00000	.00000	.00000 .00000

(RT0009)

(04 NOV 75)

(ST0009)

(04 NOV 75)

(VT0009)

(04 NOV 75)

PARAMETRIC DATA

BETA	=	
ELEVTR	=	
RUDDER	=	
BDFLAP	=	
		.000
		.000
		.000
		.000

CA	=	
L/D	=	
CAF	=	
VBAR	=	
		.08120
		.120000
		.08490
		.08750
		.08970
		.09080
		.00000

RNL/L	=	
AIRRON	=	
SPDBRK	=	
		.000
		.000
		.000

PARAMETRIC DATA

BETA	=	
ELEVTR	=	
RUDDER	=	
BDFLAP	=	

CA	=	
L/D	=	
CAF	=	
VBAR	=	

RNL/L	=	
AIRRON	=	
SPDBRK	=	

AEDEC V4489(04-81); (826C9F7M7N28) (W116E26) (V885)

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

RUN NO.		49 / 0		R/N/L =	.11		GRADIENT INTERVAL =		-5.00 / 5.00		CAF M	
ACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE	CAM	CAF M	
.000	30,000	80,00000	.81600	-.01170	.65528	.64500	.51400	1.26000	.05380	.12100	.12228	
.200	30,000	90,00000	.81600	-.01170	.65528	.64600	.51300	1.26000	.05311	.12000	.12128	
.300	30,000	100,00000	.81600	-.01170	.65528	.64500	.51500	1.25000	.05400	.12200	.12328	
.000	30,000	110,00000	.81100	-.01470	.65667	.64100	.51200	1.25000	.05331	.12100	.12228	
.500	30,000	120,00000	.80900	-.01640	.65746	.63900	.51000	1.25000	.05331	.12100	.12228	
GRADIENT			.00000		.00000		.00000		.00000		.00000	
RUN NO.		31 / 0		R/N/L =	.10		GRADIENT INTERVAL =		-5.00 / 5.00		CAF M	
ACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE	CAM	CAF M	
.000	35,000	80,00000	1.04000	-.03490	.66235	.77300	.70700	1.09000	.05795	.13400	.13546	
.700	35,000	90,00000	1.03000	-.03420	.66222	.76600	.70700	1.08000	.06015	.13900	.14046	
.500	35,000	100,00000	1.03000	-.03520	.66258	.76100	.70800	1.08000	.06145	.14200	.14347	
.500	35,000	110,00000	1.02000	-.03560	.66284	.75500	.70800	1.07000	.06276	.14500	.14649	
.500	35,000	120,00000	1.02000	-.03820	.66378	.74900	.70800	1.06000	.06450	.14900	.15051	
.400	35,000	130,00000	1.02000	-.03920	.66414	.74500	.71000	1.05000	.06624	.15300	.15454	
GRADIENT			.00000		.00000		.00000		.00000		.00000	

AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (Y8R5)

REFERENCE DATA

SREF = 2690.0000 \$0.ET XMBP = 1076.7000 INCHES

PARAMETRIC DATA

	RNL	ALRDN	SPDBRK	CAM	CAFM
BETA	.000				.070
ELEVTR	.000				.000
RUDDER	.000				55.000
BDFLAP	.000				
0/ 5.00	L/D	CAE		CAM	CAFM
	1.260000	.05380		.12100	.1222
	1.260000	.05311		.12000	.1215
	1.250000	.05400		.12200	.1235
	1.250000	.05331		.12100	.1222
	1.250000	.05331		.12100	.1222
	.000000	.00000		.00000	.00000
0/ 5.00	L/D	CAE		CAM	CAFM
	1.090000	.05795		.13400	.1354
	1.080000	.06015		.13900	.1404
	1.080000	.06145		.14200	.1434
	1.070000	.06276		.14500	.1464
	1.060000	.06450		.14900	.1505
	1.050000	.06624		.15300	.1545
	.000000	.00000		.00000	.00000

(ST0010) (04 NOV 75)		AMMETERIC DATA	
		R/N/L	.070
.000		AILRON	.000
.000		SPDBRK	55.000
.000			.0000
V-1NF	VLBAR	C-FC1	
.049970	.04193	1.2200	
.05320	.04670	1.2300	
.05620	.04814	1.2300	
.05690	.04764	1.2400	
.06190	.052236	1.2400	
	.000000	.000000	

DATE 04 NOV 75

TABULATED SOURCE DATA. AEDC VA 489. OAB1

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AEDC VA489(OA-81). (B26C97M7N28) (W116E26) (V8R5)

REFERENCE DATA

SREF	2690.0000	SQ.F.T.	XMRP	1076.7000	INCHES
LREF	474.8000	INCHES	YMRP	.0000	INCHES
BREF	936.7000	INCHES	ZMRP	.375.0000	INCHES
SCALE	.0100				

RUN NO. 31 / 0 RNL = 1.10 GRADIENT INTERVAL = -5.00 / 5.00

TIME	P0	Q(PSI)	T0	RE-L	CPB	V-INF	VLBAR	C-FCTR
35.000	80.00000	4071.00000	.00063	.18000	4418.00000	.00940	.04916	1.28000
35.000	90.00000	3969.00000	.00063	.17000	5219.00000	.10000	.04948	1.28000
35.000	100.00000	3731.00000	.00061	.16000	5536.00000	.09000	.00950	1.28000
35.000	110.00000	3529.00000	.00058	.15000	5334.00000	.09000	.00960	1.28000
35.000	120.00000	3410.00000	.00056	.15000	5422.00000	.08000	.00970	1.28000
35.000	130.00000	3257.00000	.00055	.14000	5392.00000	.07000	.00990	1.28000
GRADIENT	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000

AEDC VA489(OA-81). (B26C97M7N28) (W116E26) (V8R5)

REFERENCE DATA

SREF	2690.0000	SQ.F.T.	XMRP	1076.7000	INCHES
LREF	474.8000	INCHES	YMRP	.0000	INCHES
BREF	936.7000	INCHES	ZMRP	.375.0000	INCHES
SCALE	.0100				

TIME	CN	CLM	XCP/L	CL	L/D	CA	CAF	VLBAR
60.00000	.65700	.02220	.66243	.56100	.35100	.60000	.08100	.01174
65.00000	.65800	-.02270	.66269	.56100	.35200	.59000	.08200	.01279
70.00000	.66100	-.02240	.66247	.56300	.35500	.59000	.08350	.01350
75.000	.67300	-.02400	.66312	.57400	.36200	.58000	.08550	.01396
80.00000	.67800	-.02480	.66346	.57800	.36500	.58000	.08630	.01395
90.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
GRADIENT	.000000							

TIME	CN	CLM	XCP/L	CL	L/D	CA	CAF	VLBAR
60.00000	.80500	-.03620	.66655	.65600	.47200	.139000	.08050	.01110
65.000	.81600	-.03640	.66641	.66500	.47900	.139000	.08020	.011320
70.00000	.82500	-.03560	.66558	.67200	.46600	.138000	.08400	.01395
75.000	.85200	-.03360	.66451	.69400	.50100	.138000	.08630	.01398
80.00000	.87400	-.03310	.66394	.71200	.51500	.138000	.08960	.01485
90.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
GRADIENT	.000000							

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TABULATED SOURCE DATA, AEDC VA 489, OABI

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REFERENCE DATA

PARAMETRIC DATA

REFERENCE DATA

AEDC VA489(0A-81). (8296C9F7M7N28) (W11SE26) (V8R5)

PARAMETRIC DATA

	REF	SCALE	5Q.FT.	XMRP	YMRP	ZMRP	RNL	BETA	ELEVTR	RUDDER	BDFLAP	RNL/L	AIRRON	SPDBRK	V-1 INF	CPB	RE-L	Q(PSI)	TIME	P	RUN NO.						
SGREF	=	2690.0000	5Q.FT.	=	1076.7000	INCHES						.0000				1.100											
LREF	=	474.8000	INCHES	=	.0000	INCHES						.0000				.000											
BREF	=	936.7000	INCHES	=	375.0000	INCHES						.0000				55.000											
SCALE	=	.0100																									
												GRADIENT INTERVAL = -5.00 / 5.00															
MACH	ALPHA	TIME	PO	P	Q(PSI)	TO		RE-L	CPB													V-LBAR					
16.300	25.000	60.0000012259.00000		.00970	1.760000	3679.00000		1.34000														.01174					
16.600	25.000	65.0000011482.00000		.00812	1.560000	3689.00000		1.17000														.01279					
16.800	25.000	70.0000011036.00000		.00725	1.430000	3531.00000		1.09000														.01350					
17.000	25.000	80.0000010114.00000		.00623	1.290000	3448.00000		1.05000														.01396					
16.700	25.000	90.000009309.00000		.00644	1.250000	3397.00000		1.02000														.01395					
GRADIENT .00000												.00000 .00000												.00000		.00000	

(RT0011) (04 NOV 75)

DATE 04 NOV 75

TABULATED SOURCE DATA. AEDC VA 489, OA81

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REFERENCE DATA

AEDC V44P9(0A-81), (B26C9F7M7N28) (W116E26) (VBR5)

(RT0012) (04 NOV 75)

PARAMETRIC DATA			
	BETA	RNL	RNL
LREF	.0000	15.000	AIRRON
XMRP	.0000	.0000	SRDBRK
YMRP	.0000	.0000	
ZMRP	.0000	.0000	
REFL	.0000	.0000	
SCAL	.0000	.0000	

PARAMETRIC DATA

MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	CAF	VLBLAR
6.300	30.000	70.00000	.86500	-.1C200	.70190	.69800	.52100	.1.34000	.10200	.01351	
5.900	30.000	75.00000	.85900	-.12300	.70269	.69200	.52000	.1.33000	.10400	.01435	
6.000	30.000	80.00000	.85500	-.12400	.70337	.68800	.51900	.1.33000	.10500	.01500	
5.600	30.000	90.00000	.83900	-.12400	.70439	.67400	.51200	.1.32000	.10600	.01536	
5.500	30.000	100.00000	.82600	-.12500	.70569	.66200	.50600	.1.31000	.10700	.01655	
GRADIENT		.00000	.00000		.00000		.00000		.00000	.00000	
	RUN NO.	35/ 0	RN/L =	.99	GRADIENT INTERVAL	= -5.00/ -5.00					
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CA	CAF	VLBLAR
6.200	35.000	55.00000	1.30000	-.16300	.6961	.99600	.85100	.1.17000	.12500	.01217	
5.800	35.000	60.00000	1.30000	-.16200	F'-.36	.99100	.84700	.1.17000	.12500	.01281	
6.000	35.000	70.00000	1.29000	-.16000	69564	.98100	.84200	.1.16000	.12700	.01434	
6.100	35.000	80.00000	1.27000	-.15600	69520	.96900	.83500	.1.16000	.12700	.01502	
6.600	35.000	90.00000	1.27000	-.15800	69578	.96660	.83300	.1.16000	.12800	.01568	
GRADIENT		.00000	.00000		.00000		.00000		.00000	.00000	

AEDC VA489(0A-81), (B26C9F7M7N28) (W116E26) (V8R5)

REFERENCE DATA				PARAMETRIC DATA			
SREF	2690.0000	SO.FT.	XMRP	1076.7000	INCHES	BETA	1.100
LREF	474.8000	INCHES	YMRP	.0000	INCHES	ELEVTR	.000
BREF	936.7000	INCHES	ZMRP	375.0000	INCHES	RUDER	.000
SCALF	.0100					ROFLAP	.000

REFERENCE DATA

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, 0AB1

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AEDC VA489 (CA-B1), (B26C9F7M7N28) (W116E26) (V8R5)

REFERENCE DATA

SREF	=	2690.0000 SQ.FT.	XMRP	=	1076.7000 INCHES
LREF	=	474.800 INCHES	YMRP	=	.0000 INCHES
BREF	=	936.7000 INCHES	ZMRP	=	.375.0000 INCHES
SCALE	=	.0100			

RUN NO.		RN/L =	.99	GRADIENT INTERVAL =	-5.00 / 5.00		PARAMETRIC DATA
MACH	ALPHA	TIME	P0	Q(PSI)	TO	RE-L	
16.200	25.000	60.0000011626.00000	.00977	1.800000	3638.00000	1.16000	CPB
15.900	25.000	65.0000011256.00000	.00986	1.630000	4249.00000	.94000	V-LBAR
16.200	25.000	70.0000010658.00000	.00840	1.520000	4004.00000	.85000	.01400
16.300	25.000	80.00000 9633.00000	.00728	1.370000	3783.00000	.62000	.01520
16.400	25.000	90.00000 8893.00000	.00668	1.220000	3666.00000	.83000	.01338
GRADIENT		.00000	.00000	.00000	.00000	.00000	.01640
MACH	ALPHA	TIME	P0	Q(PSI)	TO	RE-L	
16.300	30.000	70.0000011378.00000	.00839	1.560000	4118.00000	.98000	CPB
15.900	30.000	75.0000010689.00000	.00907	1.540000	4229.00000	.82000	V-LBAR
16.000	30.000	80.0000010230.00000	.00811	1.500000	4409.00000	.75000	.0113
15.600	30.000	90.00000 9426.00000	.00879	1.450000	4598.00000	.67000	.01125
15.500	30.000	100.00000 8622.00000	.00858	1.400000	4160.00000	.59000	.011435
GRADIENT		.00000	.00000	.00000	.00000	.00000	.01720
MACH	ALPHA	TIME	P0	Q(PSI)	TO	RE-L	
16.200	35.000	55.0000012503.00000	.01070	1.960000	3579.00000	1.24000	CPB
15.800	35.000	60.0000011587.00000	.01050	1.830000	4274.00000	1.01000	V-LBAR
16.000	35.000	70.0000010781.00000	.00885	1.600000	4394.00000	.62000	.01360
16.100	35.000	60.00000 9754.00000	.00766	1.390000	4196.00000	.77000	.01460
16.600	35.000	90.00000 8955.00000	.00613	1.200000	3695.00000	.78000	.01434
GRADIENT		.00000	.00000	.00000	.00000	.00000	.01640
MACH	ALPHA	TIME	P0	Q(PSI)	TO	RE-L	

REFERENCE DATA

SREF	2690.00000	SQ.FT.	XMRP	1076.7000	INCHES	BETA	.000	RN/L	.250
LREF	474.80000	INCHES	YMRP	.0000	INCHES	ELEVTR	15.000	AIRRON	.000
BREF	936.70000	INCHES	ZMRP	375.0000	INCHES	RUDDER	.000	SPDBRK	55.000
SCALE	.01000					BOFLAP	16.300		
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE
20.300	20.000	120.00000	.47500	-.07140	.70531	.41100	.26100	1.57000	CAM
19.500	20.000	130.00000	.47200	-.07010	.70465	.40800	.26000	1.57000	CAF
19.600	20.000	140.00000	.46200	-.06910	.70528	.39900	.25500	1.57000	.04844
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.046662	.10500
			RUN NO.	42 / 0	RN/L =	.28	GRADIENT INTERVAL =	-5.00/ 5.00	.10531
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE
20.100	25.000	90.00000	.67400	-.07520	.69106	.56200	.38900	1.45000	CAM
19.600	25.000	100.00000	.65700	-.07730	.69330	.54700	.38200	1.43000	CAF
19.400	25.000	110.00000	.65100	-.08150	.69607	.54100	.37900	1.43000	.042222
19.600	25.000	120.00000	.63400	-.08200	.69759	.52600	.37300	1.43000	.042225
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.04225	.11400
			RUN NO.	41 / 0	RN/L =	.19	GRADIENT INTERVAL =	-5.00/ 5.00	.11447
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE
20.100	30.000	100.00000	.99200	-.09970	.68698	.79300	.61100	1.30000	CAM
19.800	30.000	110.00000	.99300	-.09600	.68558	.79700	.60700	1.31000	CAF
19.600	30.000	120.00000	.99400	-.09530	.68528	.80100	.60200	1.32000	.04992
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.046339	.12600
			RUN NO.	40 / 0	RN/L =	.22	GRADIENT INTERVAL =	-5.00/ 5.00	.12000
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE
20.000	35.000	100.00000	.99200	-.09970	.68698	.79300	.61100	1.30000	CAM
19.800	35.000	110.00000	.99300	-.09600	.68558	.79700	.60700	1.31000	CAF
19.600	35.000	120.00000	.99400	-.09530	.68528	.80100	.60200	1.32000	.05339
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.13200
			RUN NO.	39 / 0	RN/L =	.17	GRADIENT INTERVAL =	-5.00/ 5.00	.13282
MACH	ALPHA	TIME	CN	CLM	XCP/L	CL	CD	L/D	CAE
20.000	35.000	90.00000	1.20000	-.12700	.68894	.89400	.80800	1.11000	CAM
20.000	35.000	100.00000	1.18000	-.12900	.69023	.88300	.79900	1.11000	CAF
19.600	35.000	110.00000	1.17000	-.13100	.69120	.87800	.79200	1.11000	.05730
19.300	35.000	120.00000	1.16000	-.12900	.69092	.86400	.78200	1.10000	.05614
19.500	35.000	140.00000	1.13000	-.13400	.69364	.84000	.77100	1.09000	.05615
	GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000	.05912	.14400
								.00000	.14900
									.14981
									.00000

PARAMETRIC DATA

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OA81

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AEDC VA489(OA-81), (B26C9F7M7N28) (W116E26) (V8R5)

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

RUN NO. 42/ 0

RN/L =

TIME PO

P

Q(P(S1))

TO

RE-L

CPB

GRADIENT INTERVAL = -5.00/ 5.00

MACH ALPHA 20.000 120.0000010122.00000

LREF 20.000 130.00000 9757.00000

BREF 20.000 140.00000 9575.00000

SCALE .00000 .00000

RUN NO. 41/ 0

RN/L =

TIME PO

P

Q(P(S1))

TO

RE-L

CPB

GRADIENT INTERVAL = -5.00/ 5.00

MACH ALPHA 25.000 90.0000010548.00000

LREF 25.000 100.0000010194.00000

BREF 25.000 110.00000 9722.00000

SCALE .00000 .00000

RUN NO. 40/ 0

RN/L =

TIME PO

P

Q(P(S1))

TO

RE-L

CPB

GRADIENT INTERVAL = -5.00/ 5.00

MACH ALPHA 30.000 100.0000010456.00000

LREF 30.000 110.0000010098.00000

BREF 30.000 120.00000 9740.00000

SCALE .00000 .00000

RUN NO. 39/ 0

RN/L =

TIME PO

P

Q(P(S1))

TO

RE-L

CPB

GRADIENT INTERVAL = -5.00/ 5.00

MACH ALPHA 35.000 90.0000010297.00000

LREF 35.000 100.0000010238.00000

BREF 35.000 110.00000 9648.00000

SCALE .00000 .00000

RUN NO. 38/ 0

RN/L =

TIME PO

P

Q(P(S1))

TO

RE-L

CPB

GRADIENT INTERVAL = -5.00/ 5.00

MACH ALPHA 40.000 90.0000010344.00000

LREF 40.000 100.00000 6305.00000

BREF 40.000 120.00000 9691.00000

SCALE .00000 .00000

RUN NO. 37/ 0

RN/L =

TIME PO

P

Q(P(S1))

TO

RE-L

CPB

GRADIENT INTERVAL = -5.00/ 5.00

MACH ALPHA 45.000 90.0000010397.00000

LREF 45.000 100.00000 6877.00000

BREF 45.000 120.00000 9900.00000

SCALE .00000 .00000

RUN NO. 36/ 0

RN/L =

TIME PO

P

Q(P(S1))

TO

RE-L

CPB

GRADIENT INTERVAL = -5.00/ 5.00

MACH ALPHA 50.000 90.0000010444.00000

LREF 50.000 100.00000 6877.00000

BREF 50.000 120.00000 9900.00000

SCALE .00000 .00000

RUN NO. 35/ 0

RN/L =

TIME PO

P

Q(P(S1))

TO

RE-L

CPB

GRADIENT INTERVAL = -5.00/ 5.00

PARAMETRIC DATA

(ST0013) (04 NOV 75)

REFERENCE DATA

AEDC VA489(OA-81), (B26C9F7M7N28) (W116E26) (V8R5)

PARAMETRIC DATA

(04 NOV 75)

BETA

ELEVTR

RUDDER

BDFLAP

RN/L

AILRON

SPDBRK

VLBAR

C-FCTR

V-1NF

CPB

GRADIENT INTERVAL = -5.00/ 5.00

BETA

ELEVTR

RUDDER

BDFLAP

RN/L

AILRON

SPDBRK

VLBAR

C-FCTR

V-1NF

CPB

GRADIENT INTERVAL = -5.00/ 5.00

BETA

ELEVTR

RUDDER

BDFLAP

RN/L

AILRON

SPDBRK

VLBAR

C-FCTR

V-1NF

CPB

GRADIENT INTERVAL = -5.00/ 5.00

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OA81

AEDC489(OA81) B26C9F7M7N28 W116E26 V8R5 INVERTED

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(RT0014) (04 NOV 75)

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES	
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES	
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES	
SCALE	=	.0100						

RUN NO. 43/ 0 RN/L = .97 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	TIME	CN	CL	CD	L/D	CA	CAF	VLBAR
16.400	25.000	60.00000	.57100	-.00853	.48900	1.62000	.06700	.06720	.01169
16.100	25.000	65.00000	.59000	-.00870	.50400	1.61000	.07050	.07070	.01295
16.300	25.000	70.00000	.60300	-.01000	.51600	1.61000	.07300	.07320	.01424
16.200	25.000	80.00000	.62100	-.01050	.55628	1.60000	.07700	.07720	.01527
16.200	25.000	90.00000	.62900	-.01080	.65632	1.59000	.07980	.08000	.01563
15.900	25.000	100.00000	.63900	-.01130	.65660	1.58900	.08000	.08020	.01660
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

RUN NO. 44/ 0 RN/L = 1.03 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	TIME	CN	CL	CD	L/D	CA	CAF	VLBAR
16.200	35.000	55.00000	1.01000	-.04090	.66490	1.23000	.07200	.07280	.01145
16.000	35.000	60.00000	1.02000	-.04400	.66587	1.23000	.07300	.07380	.01226
16.300	35.000	70.00000	1.04000	-.04750	.66681	1.23000	.07550	.07630	.01383
16.300	35.000	80.00000	1.05000	-.05020	.66759	1.23000	.07760	.07840	.01476
16.200	35.000	90.00000	1.06000	-.05370	.66864	1.22000	.07900	.07980	.01515
16.200	35.000	100.00000	1.07000	-.05480	.66885	1.22000	.08000	.08080	.01584
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

AEDC489(OA81) B26C9F7M7N28 W116E26 V8R5 INVERTED

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES	
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES	
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES	
SCALE	=	.0100						

RUN NO. 43/ 0 RN/L = .97 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	TIME	PQ	P	Q(PSI)	T0	RE-L	CPB	V-INF	VLBAR
16.400	25.000	60.0000012480.00000	.01000	1.87000	3462.00000	1.39000	.0025	.01300	.01169	
16.100	25.000	65.0000011689.00000	.00948	1.72000	4113.00000	1.04000	.0025	.01470	.01295	
16.300	25.000	70.0000011220.00000	.00840	1.56000	4129.00000	.88000	.0030	.01610	.01424	
16.200	25.000	80.0000010226.00000	.00780	1.40000	4247.00000	.75000	.0030	.01740	.01527	
16.200	25.000	90.000009288.00000	.00697	1.30000	4187.00000	.72000	.0025	.01780	.01563	
15.900	25.000	100.000008467.00000	.00637	1.22000	4319.00000	.61000	.0025	.01890	.01660	
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	

(ST0014) (04 NOV 75)

PARAMETRIC DATA

BETA	=	.000	RN/L	=	1.100
ELEVTR	=	.000	AIRRON	=	.000
RUDDER	=	.000	SPDBRK	=	55.000
BDFLAP	=	-11.700			

(ST0014) (04 NOV 75)

BETA	=	.000	RN/L	=	1.100
ELEVTR	=	.000	AIRRON	=	.000
RUDDER	=	.000	SPDBRK	=	55.000
BDFLAP	=	-11.700			

(ST0014) (04 NOV 75)

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OA81

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AEDC489(OA81) B26C9F7M7N28 W116E26 V8R5 INVERTED

REFERENCE DATA

SREF	=	2690.0000	SO.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.6000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

RUN NO.	44 / 0	RNL =	1.03	GRADIENT INTERVAL =	-5.00 /	5.00
MACH	ALPHA	TIME	P0	0 (PSI)	RE-L	CPB
16.200	35.000	55.0000012729.00000	.01070	1.96000 .3672 .00000	1.39000	.00500
16.000	35.000	60.0000012065.00000	.00994	1.75000 .4198 .00000	1.14000	.00510
16.300	35.000	70.0000011097.00000	.00808	1.50000 .4194 .00000	.93000	.00516
16.300	35.000	80.0000010160.00000	.00787	1.35000 .3814 .00000	.84000	.00520
16.200	35.000	90.00000 9282.00000	.00710	1.29000 .4128 .00000	.77000	.00523
15.800	35.000	100.00000 8373.00000	.00746	1.25000 .4134 .00000	.67000	.00530
GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000

AEDC489(OA81) B26C9F7M7N28 W116E26 V8R5 INVERTED

REFERENCE DATA

SREF	=	2690.0000	SO.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.6000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

RUN NO.	45 / 0	RNL =	.23	GRADIENT INTERVAL =	-5.00 /	5.00
MACH	ALPHA	TIME	CN	XCP/L	CD	L/D
20.300	25.000	100.00000	.50800	-.01530	.51000	.3400
19.800	25.000	110.00000	.61000	-.01520	.51100	.3700
19.600	25.000	120.00000	.61200	-.01360	.51200	.35000
19.400	25.000	140.00000	.61300	-.01160	.51300	.35700
19.500	25.000	160.00000	.61600	-.01170	.51600	.36700
GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000

PARAMETRIC DATA

BETA	=	.000	RNL	=	1.100
ELEVTR	=	.000	AIRLON	=	.000
RUDDER	=	.000	SPDBRK	=	55.000
BDFLAP	=	-11.700			

PARAMETRIC DATA

BETA	=	.000	RNL	=	.250
ELEVTR	=	.000	AIRLON	=	.000
RUDDER	=	.000	SPDBRK	=	55.000
BDFLAP	=	-11.700			

PARAMETRIC DATA

SREF	=	2690.0000	SO.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.6000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

RUN NO.	46 / 0	RNL =	.19	GRADIENT INTERVAL =	-5.00 /	5.00
MACH	ALPHA	TIME	CN	CLM	CD	L/D
19.700	35.000	80.00000	1.04000	-.04630	.78300	.69700
19.700	35.000	90.00000	1.04000	-.04630	.66638	.78200
19.600	35.000	100.00000	1.04000	-.04530	.66603	.77900
19.400	35.000	120.00000	1.04000	-.04470	.66582	.77500
19.500	35.000	140.00000	1.03000	-.04360	.66558	.77000
19.300	35.000	160.00000	1.03000	-.04050	.66447	.76800
GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000

SREF	=	2690.0000	SO.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.6000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

RUN NO.	47 / 0	RNL =	.19	GRADIENT INTERVAL =	-5.00 /	5.00
MACH	ALPHA	TIME	CN	CLM	CD	L/D
19.700	35.000	80.00000	1.04000	-.04630	.78300	.69700
19.700	35.000	90.00000	1.04000	-.04630	.66638	.78200
19.600	35.000	100.00000	1.04000	-.04530	.66603	.77900
19.400	35.000	120.00000	1.04000	-.04470	.66582	.77500
19.500	35.000	140.00000	1.03000	-.04360	.66558	.77000
19.300	35.000	160.00000	1.03000	-.04050	.66447	.76800
GRADIENT	.00000	.00000	.00000	.00000	.00000	.00000

BETA	=	.000	RNL	=	.250
ELEVTR	=	.000	AIRLON	=	.000
RUDDER	=	.000	SPDBRK	=	55.000
BDFLAP	=	-11.700			

BETA	=	.000	RNL	=	.09571
ELEVTR	=	.000	AIRLON	=	.09500
RUDDER	=	.000	SPDBRK	=	.09890
BDFLAP	=	-11.700			.04051

BETA	=	.000	RNL	=	.09571
ELEVTR	=	.000	AIRLON	=	.09500
RUDDER	=	.000	SPDBRK	=	.09890
BDFLAP	=	-11.700			.04051

DATE 04 NOV 75

TABULATED SOURCE DATA, AEDC VA 489, OA81

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AEDC489(OA81) B26C9F7M7N28 W116E26 V8RS INVERTED

(ST0016) (04 NOV 75)

REFERENCE DATA

SREF	=	2690.0000	SQ.FT.	XMRP	=	1076.7000	INCHES
LREF	=	474.8000	INCHES	YMRP	=	.0000	INCHES
BREF	=	936.7000	INCHES	ZMRP	=	375.0000	INCHES
SCALE	=	.0100					

RUN NO. 47 / 0 RN/L = .29 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	TIME	PO	P	Q1PSI1	TO	RE-L	CPB	V-INF	VLBAR
15.700	35.000	80.00000	4959.00000	.00402	.70000	4453.00000	.38000	.00506	.02370	.02070
15.800	35.000	90.00000	4504.00000	.00365	.63000	4295.00000	.30000	.00513	.02680	.02356
15.800	35.000	100.00000	4128.00000	.00323	.57000	4424.00000	.26000	.00517	.02880	.02520
16.000	35.000	110.00000	3598.00000	.00271	.49000	4385.00000	.24000	.00524	.03030	.02659
GRADIENT		.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000

PARAMETRIC DATA

BETA	=	.000	RN/L	=	.500
ELEVTR	=	.000	AILRON	=	.000
RUDDER	=	.000	SPDBRK	=	.55.000
BDFLAP	=	-1.1.700			